



**Bellcomm**

955 L'Enfant Plaza North, S.W.  
Washington, D. C. 20024

date: March 31, 1972  
to: Distribution  
from: D. A. Corey, D. N. Sakolosky, E. W. Radany  
subject: A Computer Program to Aid Skylab Earth Resources  
Experiment Planning - Case 610

B72 03019

ABSTRACT

The multitude of data which must be handled by the mission planner in his planning studies of the Skylab Earth Resources Experiment Program makes such studies most formidable. This memorandum documents a computer program designed to handle much of the necessary data for the planner thus making such studies more feasible. The program is designed to be used by the planner in an interactive time sharing environment. The documentation describes the data base, the theory and workings of the program, its use, its known errors, and the necessary details of running it.

(NASA-CR-126940) A COMPUTER PROGRAM TO AID  
(NASA-CR-126940) SKYLAB EARTH RESOURCES EXPERIMENT PLANNING  
(Bellcomm, Inc.) 159 p

N79-72943

00/43 Unclas  
12893



TABLE OF CONTENTS

<u>Section Name</u>	<u>Page</u>
INTRODUCTION	1
DATA BASE	3
PROGRAM OPTIONS, OPERATION, AND USE	12
BASIC ASSUMPTIONS AND ALGORITHMS	20
WORKING DATA ARRAYS AND VARIABLES	25
PROGRAM AVAILABILITY	34
PROGRAM USERS GUIDE	35
SUBROUTINE DESCRIPTIONS	44
KNOWN ANOMALIES AND INCONSISTENCIES	79
THE WORLD IN OMEGA THETA COORDINATES $50^{\circ}$ INCLINATION, 235 NM CIRCULAR ORBIT	FIGURE 1
THE CONTINENTAL UNITED STATES IN OMEGA THETA COORDINATES $50^{\circ}$ INCLINATION, 235 NM CIR- CULAR ORBIT	FIGURE 2
THE DIFFERENCE BETWEEN THE VALUE OF OMEGA AT THE EDGE OF THE FIELD OF VIEW AND THE VALUE OF OMEGA OF THE SPACECRAFT V.S. THETA FOR THE S190 CAMERAS	FIGURE 3
PROGRAM LISTINGS	APPENDIX A
PROGRAM OUTPUT EXAMPLES	APPENDIX B
DISTRIBUTION LIST	



**Bellcomm**

955 L'Enfant Plaza North, S.W.  
Washington, D. C. 20024

date: March 31, 1972  
to: Distribution  
from: D. A. Corey, D. N. Sakolosky, E. W. Radany B72 03019  
subject: A Computer Program to Aid Skylab Earth Resources  
Experiment Planning - Case 610

MEMORANDUM FOR FILE

INTRODUCTION

The operating aspects of the Skylab Earth Resources Experiment Program (EREP) differ significantly from the other scientific investigations to be carried out on Skylab in that a basic facility is being supplied for use by a large number of investigators rather than just a few. Each of the investigators is interested in different data taken under widely-differing conditions. For example, a study of a candidate set of 161 investigations, or proposals, (the same set used in an MSC mission sizing study) revealed some 891 schedulable objectives. Many of these objectives require multiple passes over the ground site to be satisfied. The proposals referenced 556 different, though in many cases, overlapping, ground sites. The data required is to be taken under a wide variety of lighting constraints, cloud cover constraints, and in various months throughout the year. Each objective requires that a certain set of sensors be used in the various modes in which the sensors can operate. In addition, many of the proposals have special constraints. Some require data obtained only on an ascending (heading in a northeasterly direction) pass across the ground site. One requires a night pass occurring within one day of a daylight pass. The mission planner must combine this complex set of requirements with the trajectory data, and with the other constraints placed on the crew work schedule. The uncertainty of local weather in the multitude of areas of interest further adds to his problem. We concluded that the planning task was too formidable to be efficiently done by hand. Certainly, the numerous studies one would like to perform would be severely limited. Consequently, it was decided to build a computer program which would take care of most of the bookkeeping and data handling required, and hopefully free the planner of most of the tedious labor involved.



This memorandum describes that program as it exists now. While there are desirable additional features and capabilities not currently implemented, it is felt that the program is advanced enough to be useful in mission planning studies and, perhaps, in the real time flight control environment. The demise of Bellcomm precludes further development and use by Bellcomm, but it is hoped that NASA will find it useful enough to justify the usual pangs of picking up somebody else's program. Since the current situation was known at the program's inception, every effort was made to keep its structure and coding as straightforward as possible.

The program was designed to be used by the planner, as opposed to a programmer, in an interactive time-sharing environment. The user is only required to remember a few commands and some flexibility in their proper spelling is allowed. He is requested, more or less in English, to supply necessary problem data at the time it is required. Where a particular format for user supplied data is expected, he is reminded of the correct format. Some data is less frequently supplied by the user so provision is made for him to be reminded of proper spelling if he so desires. The net result, it is hoped, is that the program is relatively simple to operate and the user's thought processes may be given over to more important things. None of this, of course, obviates the need for the user to have a basic understanding of what the program is doing when he gives it a command. He must have the understanding to properly operate the program and to correctly interpret the results.

Operation of the program in a batch mode is somewhat awkward in that the data deck must be carefully stacked and program requests for data or responses must be anticipated. There are certain situations, however, where batch operation is worth the trouble - where voluminous displays of data are required for example.

With the exception of one subroutine, the entire program is written in Fortran and uses standard UNIVAC library routines. It is currently structured to be run on a UNIVAC 1108 under the Exec 8 operating system. It currently requires approximately 13000 words for program storage and about 45000 words for data storage. Further segmentation and linked operation of the program is, of course, possible if the storage requirements are too large for a particular environment.



The non-Fortran subroutine, CINOTE, supplies a convenience feature and is not actually required by the program. It allows interruption of a lengthy data display, in the event the user gets tired of looking at it, and returns control to the user. It is written in 1108 assembly language and a listing is included with the other listings.

#### DATA BASE

A group of files which catalog the requirements of the various proposals, describe the ground sites, and specify the parameters of each revolution of the trajectory comprise the main data base. The contents and organization of the files in each of these categories are described below.

#### Proposal Requirements

The data file titled EREPD describes, in a coded format, the requirements of the majority of the proposals for performing earth resources investigations presently under consideration. EREPD is the raw data file constructed directly from a reading of the proposals. Several proposals were omitted for the following reasons: (a) the proposer merely required data from some other investigation, (b) the associated ground site would only be specified shortly before the mission, or would be specified in real time.

The file EREPD is constructed to be read by a namelist read directly into an array, EREP, dimensioned 15, n; where n is the number of proposal subobjective (FO's) to be considered. Accordingly, the first card image in EREPD is "A\$EREPD", the second card image is "AEREP =", and the last card image is "A\$END". (The A indicates a blank in column 1.) Each card image in between contains a blank in column 1 followed by fifteen floating point numbers. A comma follows each number.

The sequence and format of each of the entries in a row (or card image) are described below.

Entry 1 - proposal number, assigned by NASA. Most usually a three digit number; e.g. 052., 356., etc. Sub-sectioned proposals are encoded with a four digit number with the proposal sub-section number appearing as the thousands digit; e.g. proposal 540-3 is indicated by "3540."



- Entry 2 - flight objective number. A flight objective is defined to be an individual, schedulable event necessary for the performance of a particular proposal. For example, a particular proposal specifies that two passes over a prescribed site are required; one a daylight pass in May, and the other a night pass in November. One daylight pass over the site occurring specifically in May would be denoted as flight objective #1 of this proposal, and a night pass occurring specifically in November would be flight objective #2. Multiple sites, variable instrument combinations, different seasonal requirements, etc., associated with a proposal result in the designation of multiple flight objectives.
- Entry 3 - site number. This number denotes the ground site associated with the particular proposal. The geographic description of each of the sites referenced in the proposals is contained in a file titled SITE which is described later.
- Entry 4 - discipline. A one-digit number which encodes the discipline into which the proposal is categorized; viz. 1 = agriculture/forestry, 2 = ecology, 3 = geography, 4 = geology, 5 = hydrology, 6 = interpretation techniques development, 7 = meteorology, 8 = oceanography, 9 = sensor technology.
- Entry 5 - sensors required. A five digit number, each digit of which is either "0" or "1", used to encode the sensor requirements for the particular flight objective. The units digit indicates S-190, the tens digit indicates S-191, etc. The digit "1" indicates that the instrument is required, and the digit "0" indicates that the instrument is not required.
- Entry 6 - seasonal requirements. A four digit number which indicates the extent of the calendar months within which a flight objective may be satisfied. The tens and units digits indicate the latest acceptable month for satisfying the flight objective; the thousand and hundreds digits indicate the



earliest acceptable month. A flight objective for which a pass at anytime between May and September was acceptable would thus have the season entry "0509."; a flight objective for which a pass between November and March was acceptable would have the season entry "1103."; a flight objective which required a September pass would have the season entry "0909.".

Entry 7 - pass type. This four digit number describes the pass, or passes, required to satisfy a particular flight objective. The units digit may be either "0", "1", or "2" with the following interpretations:

"0" - arbitrary, unrelated pass(es) over a site is required.

"1" - repeating passes at "ab" day intervals are required, where "ab" are the hundreds and tens digits of the pass type descriptor.

"2" - complete coverage of the site is required.

The thousands digit indicates either the number of passes required when the units digit is "0" or "1", or the number of complete coverages required when the units digit is "2". A requirement for two arbitrary passes is indicated as "2000.", "2101." indicates that two passes separated by a ten-day interval are required, and "2002." indicates that two complete coverages of the site are required (the number of passes to achieve two complete coverages is unspecified). In some cases multiple repeat coverages are required. Two total coverages 20 days apart for example, would be coded 2202.

Entry 8 - special note. This is a number which refers to a separate list of unique requirements which could not be reasonably formatted otherwise. Examples



are: Note #3 (Proposal 380), "One night pass ideally to occur within one day of a daylight pass; Note #11 (Proposal 446), "Lighting to give minimal sun glint". When a note is not required, "0" was entered.

Entries 9 and 10 - lighting data. Entries 9 and 10, respectively, give the minimum and maximum acceptable sun elevation angles at the site in degrees.

Entries 11 and 12 - cloud cover data. Entries 11 and 12, respectively, give the minimum and maximum allowable cloud cover over the site in percent of complete overcast; no reference is made to cloud type.

Entries 13 and 14 - snow cover data. Entries 13 and 14, respectively, give the minimum and maximum acceptable snow cover in percent of the site area which is snow covered. Required snow depths, if any, are not recorded.

Entry 15 - flight objective weight. This entry gives the weight, within each individual proposal, assigned to the corresponding flight objective. The total of the weights of all the flight objectives of each proposal is 100; i.e. all the proposals are considered to be of equal importance, though the individual FO's may be unequally weighted if the proposer so indicated.

The program currently takes no action as a result of a special note (or entry in item 8). The following is a list of the meaning of the notes.

#1, Prop. 057, "Sites provided during mission in real time".

#2, Prop. 357, "Pass 1 at 1000, passes 2-9 between 0600 and 1200, pass 10 at 1500". (All times are local.)

#3, Prop. 380, "One night pass ideally to occur within one day of a daylight pass".

#4, Prop. 423, "Descending passes preferred".





- #5, Prop. 541, "Ascending pass required".
- #6, Prop. 541, "Descending pass required".
- #7, Prop. 583, F01, "Look for glory or glitter and non-glory or non-glitter; corn".
- #8, Prop. 583, F02, "Ground track crossover points at 1.5, 3.0, or 4.5 hour intervals".
- #9, Prop. 583, F03, "Look for glory and non-glory; wheat".
- #10, Prop. 070, "Require observation of spring tides".
- #11, Prop. 446, "Lighting to give minimal sun glint".
- #12, Prop. 571, "Frequency: Pairs either at 2-3 day intervals or 3 passes during SL-2, 3 passes during SL-3, and 4 passes during SL-4." (The EREPD array entries assumed the second option.)
- #13, Prop. 587, "Imagery of earth limb".
- #14, Prop. 433, "Preferred: descending passes over areas 1, 2, and 3. First alternate: ascending passes over areas 2, 3, and 4. Second alternate: descending passes over areas 4 and 5".
- #15, Prop. 464, "Specific geographic area supplied in real time".
- #16, Prop. 052, "Pass between 1200 and 1600". (Corresponding sun angles depend on latitude and season.)
- #17, Prop. 052, "Pass between 1100 and 1300". (Corresponding sun angles depend on latitude and season.)
- #18, Prop. 441, "One pass, pre-dawn". (Sun elevation angle information precludes distinguishing between pre-dawn and post sun-down.)
- #19, Prop. 441, "One pass, after sun-down".
- #20, Prop. 457-1, "Two passes with S-193".



#21, Prop. 540-6, "Lock on and track site 541 with S-191 during all overflights".

The proposal requirements data are input to the main EREP program from the data file ERDATA. ERDATA is generated from EREPD by the auxiliary routine MSORT. The EREPD file is read via a namelist read, and each FO is sorted into a mission category as follows:

<u>Category</u>	<u>Description</u>
2	FO's which can be accomplished only during the SL-2 mission
23	FO's which can be accomplished only during the SL-2 mission or the SL-3 mission
24	FO's which can be accomplished only during the SL-2 mission or the SL-4 mission
234	FO's which can be accomplished during any of the three missions
3	FO's which can be accomplished only during the SL-3 mission
34	FO's which can be accomplished only during SL-3 or SL-4
4	FO's which can be accomplished only during SL-4

Any FO not falling into one of the categories is ignored. The sorted FO's are then placed into a proposal information array by mission category in the order 2, 23, 24, 234, 3, 34, 4.

The following arrays are output in binary to file ERDATA: (1) a 7x2 integer array where (k,1) and (K,2) indicate the start and end of the data for the Kth mission category in the proposal array and (2) the 15x1000 floating point proposal array where



(J,K), J=1,15 is the proposal information for the Kth entry in the proposal array.

### Ground Site Description

Throughout the proposals, ground sites were generally specified in one of three fashions. First, many proposals explicitly indicated the geographic coordinates defining the site; such data were recorded verbatim. Secondly, certain proposals referenced designated Earth Resources Experiment Package Test Sites; the coordinate data for these sites were obtained from the February 1971 issue of the EREP Users Handbook. Lastly, many proposals merely described the required sites verbally; viz. the Sahara, Mediterranean shore of Israel, names of cities, names of Calderas, etc. Coordinate data for such sites were variously obtained from the following sources:

- (1) Rand McNally Road Atlas, 47th Annual Edition
- (2) The Times Atlas of the World, Mid-Century Edition, Houghton Mifflin Company, Boston 1957.
- (3) The National Atlas of the United States of America, United States Department of the Interior, Geological Survey, Washington, D.C., 1970.
- (4) The Encyclopedia of Oceanography, Rhodes W. Fairbridge, Reinhold Publishing Co., New York, 1966.
- (5) Volcanic Landforms and Surface Features, J. Green and N. M. Short, Springer-Verlag, New York, 1971.

The raw ground site data are maintained in a file element named NEWDAT. This file lists each site number and the latitude and longitude (both in degrees) of the corner points of each site.

The data in NEWDAT is organized to be read by a namelist read directly into two arrays in the auxiliary program SITES. The first card image in NEWDAT is  $\Delta$ \$INEREP and the last card image is  $\Delta$ \$END where the " $\Delta$ " indicates the first column blank. All card images in between must contain a blank in column 1. The site data, which is to be read into the array EREP, is preceded by the card image  $\Delta$ EREP=. Subsequent card images contain the site corner point values - expressed as the latitude (in degrees)



followed by the longitude (in degrees) of the point. The set of points for one site is separated from the set of points for the next by the two numbers: 9999.0, 9999.0,. (A comma must follow every data value in namelist formatted input.) The end of the last set of points is indicated by the two numbers: -9999.0, -9999.0,. The remainder of the data in NEWDAT is a list of the site numbers assembled in the same order as the sets of corner points. This data will be read into the array EREPNO and so is preceded by the card image  $\Delta$ EREPPNO=.

The auxiliary program SITES is located on program file (and element) EREP.SITES. SITES reads the site data from NEWDAT, converts it to a format readable by the main program (subroutine COMMON reads it) and outputs it onto the data file LATLON.

LATLON contains a counter and two arrays written out in binary. The counter is the integer value of the highest numerical value assigned as a site number. One array is an integer array, dimensioned 2x600. Entry (1,K) indicates the start location in the array of corner points at which points for the Kth site are defined. Entry (2,K) indicates the number of points which define the Kth site. The other array contains the site corner points, is floating point, and is dimensioned 2x5000. Entry (1,L) is the latitude of a point, and entry (2,L) is the longitude of a point (degrees). Note that if either the total number of sites or the highest numerical value of a site exceeds 600, or if more than 5000 corner points are required, the dimensions of the arrays should be enlarged - in subroutine COMMON as well as in SITES.

### Trajectory Description

The trajectory data for the current nominal launch conditions, generated by a modified version of the Bellcomm Apollo Simulation Program and including the three missions, are stored in binary on file NEWANO. Routine MISION on program file EREP reads the data from NEWANO, gleans all data necessary for input to the EREP program, and stores it in binary on data files SL2DAT, SL3DAT, and SL4DAT. These data files contain trajectory information for SL-2, SL-3, and SL-4 respectively. The format of NEWANO will not be described because the file is basically unnecessary. It was available to Bellcomm and was therefore referenced in lieu of generating another set of trajectory data



in the formats described below for SL2DAT, SL3DAT and SL4DAT. It is assumed that any user generating his own trajectory information can create the necessary data files in the appropriate format initially.

The start and end times for trajectory data for each mission are specified in the body of the MISION routine. Each data file created by MISION contains the following written out as one binary record:

- (1) An integer specifying the number of revolution descriptions included on the file
- (2) An integer indicating the number of the first rev for which the absolute magnitude of beta exceeds the maximum allowed value (currently  $50^\circ$ ). (Beta is the angle between the solar vector and the orbit plane.)
- (3) An integer indicating the number of the last rev for which the absolute magnitude of beta exceeds the maximum allowed value. (If the absolute value of beta never exceeds the maximum, the values for the integers in (2) and (3) are set to zero).
- (4) A floating point array dimensioned  $5 \times 700$  where, for the Kth entry,
  - (1,K) is the rev number
  - (2,K) is the longitude (degrees) of the ascending equatorial crossing.
  - (3,K) is the value of beta at the ascending equatorial crossing
  - (4,K) is the elapsed time (in days from midnight, Dec 31) of the ascending equatorial crossing
  - (5,K) is the value of theta at orbit noon

No data is included for revs for which the absolute magnitude of beta is greater than the maximum allowed.



Revolutions are counted from 50°S latitude to 50°S latitude. In terms of theta/omega coordinates, a revolution from 50°S latitude to 50°S latitude corresponds to a theta range from -90° to 270°. This definition was selected for convenience so that passes over all sites presently recorded in the LATLON file can occur on one revolution. That is, if revolutions were counted from an ascending equatorial crossing to the next ascending equatorial crossing, for example, overflights of sites which straddle the equator would have to be counted as occurring partially in revolution "N", and partially in revolution "N+1". This bookkeeping complication has thus been avoided.

#### PROGRAM OPTIONS, OPERATION, AND USE

The user directs the operation of the program by specifying various options to be executed. Before discussing the various options and the operation of the program, a description of the few rules governing the user's communications with the program will be helpful.

Yes and No responses: In several cases, the program requires a yes or no type response from the user. For example, the program may display the settings of several program variables and then ask the user if they are set to the values he wants by typing "OK?". The user signifies a negative response by typing "no" and a carriage return. Any other response, including a simple carriage return, will be interpreted as "yes".

NAMelist INPUT: There are several situations where the user may wish to change the values of one or more program variables and the system namelist routine is used to read the new values. Where the program desires input in this form, it will so indicate by saying for example: "SUPPLY NEW VALUES NAMelist NAME": where NAME is the name of the list to be used. The user then types, starting in column 2, \$NAME (or whatever the namelist name is), followed by at least one blank, the name of the variable to be changed, an equal (=), the new value desired, a comma and finally "\$END". As many variable names, =, and new values as are desired may be inserted between the \$NAME and the \$END card but a comma must follow each set. Multiple lines may be used and no more than 72 columns should be used on any one line. If the same variable is set more than once, the last value will prevail. If the user changes his mind and wants no values changed a simple \$NAME \$END will work. The following



is a typical example used at program initiation time or under the CHANGE option.

```
$CONTRL APASDE=120., MCATDE=2, (carriage return)
MISSION=3, $END (carriage return)
```

= INPUT: The program also requires some direct input from the user which may be either data or a Hollerith word. (The context of the question should help the user distinguish between these and the questions requiring a yes or no input.) In these cases, the required information is simply typed in directly. Only the questions "OPTION=" and "PRINT OPTION=" require Hollerith input. In the cases where integer data is required, the program will indicate the proper format. If, for example, the question "REV NO.? (I4 FORMAT)" is asked, the program expects a four digit number. If the user wishes to enter the value 61, the leading zeros must be typed - 0061. Blanks may be substituted for leading zeros.

When the program is ready to accept an option request, the phrase "OPTION=" will be typed by the computer. The names, alternate spellings and descriptions of the available options follow.

#### EVALUATE OR EVAL OR E

The evaluate option will cause the program to evaluate a designated revolution in order to determine which FO's can be satisfied on that revolution. The FO's which can be at least partially satisfied on the rev will be listed in the SCORE array together with the theta limits of the portion of the rev satisfying the FO and the constrained and unconstrained weight accomplished on the revolution. The decision as to whether the FO should be listed is based on whether the unconstrained weight accomplished is non zero, not the constrained weight.

Following entry of the EVALUATE option, the user will be requested to supply the revolution number to be evaluated by the statement "REV NO.? (I4 FORMAT)". The current trajectory data numbers the revolutions with revolution 1 commencing the first time theta = 270 degrees (= -90°) following SL-1 launch - that is, at the first southern-most extent of the SL-1 orbit following launch. Revolutions are counted as nodal revolutions from  $\theta = -90^\circ$  to  $\theta = +270^\circ$ . The I4 format statement indicates that the program expects a four digit number - leading zeros should



be supplied as zeros or blanks. Normally, no further output is to be expected from the evaluate option.

#### MAXWGT or M

The MAXWGT option searches through the most recently evaluated revolution to find the optimum location for the Z-local vertical pass in terms of maximizing the constrained weight achieved by the pass. The user is allowed to specify the length of the pass and the sense in which it is optimized. He may also constrain the search to a specified segment of the revolution if he wishes.

After the user types "maxwgt", the program will display the default values of ARCPAS, MCATEG, THMIN, and THMAX and ask if these are "OK". ARCPAS is the length, in degrees, of the Z local vertical pass. The value of MCATEG will control the sense in which the optimization is performed. MCATEG can be equal to 2,3,4,23,24,34, or 234. The program in its optimization, will ignore those FO's which have mission categories which include a number not found in MCATEG. For example, if MCATEG=23, FO's in the following categories will not be included in the search for the optimum: 4,24,34, and 234. All FO's will be included if MCATEG=234. THMIN and THMAX are the minimum and maximum values of theta which will be considered for starting the Z-local vertical pass. If a "no" response is received, the program types "SUPPLY VALUES, NAMELIST MAX", indicating that new values should be entered using namelist \$MAX.

The program next displays some summary results of the optimization - the rev number, the optimum value of theta at the start of the pass (THSTRT), theta at the end of the pass (THEND), and the constrained and unconstrained weights achieved by the pass (WGTMXC and WGTMXU). These weights do not include FO's which would be accomplished on the pass but which do not fall in a category included in the optimization. (A complete display of the achievements may be obtained by calling for print option REVSUM.) Note that the MAXWGT option results in values actually being written into THSTRT and THEND. These pass theta limits will be retained through subsequent options unless expressly overwritten by the user. A particular set of theta limits may result in the extent of the useful pass being shortened but not eliminated for one or more FO's. In these cases, either at least ARCMIM degrees of the FO pass must remain





or at least one-half the arc length of the possible pass must remain in order for the FO pass to be counted (see subroutine PASBAD).

The answer provided by MAXWGT is not necessarily unique. There may be many or a wide range of places to start the Z local vertical pass which will achieve the same weight. The user can explore for these by using the print suboption REVSUM.

### SCAN or S

The SCAN option automatically executes the EVALUATE and the MAXWGT options for every revolution over a prescribed set of days and determines the n best in terms of the maximum constrained weight accomplished.

On receiving the SCAN option, the program asks the user to supply the day number, (in terms of January 1, at Greenwich = day 0) of the first day of the scan by printing "FIRST DAY OF SCAN =? (I4 FORMAT)". The user supplies the properly formatted answer and then responds similarly to a request for the last day of the scan. These day numbers may be identical if only one day is to be scanned. The program next asks "OMEGA LIMITS DESIRED?" If the user knows he wants to consider only certain parts of the world, e.g., U.S. passes, he can substantially speed up the program by limiting the allowable range of omega of the revolutions considered. An affirmative response will evoke the request "TYPE IN WESTERN OMEGA LIMIT." which the user then supplies as a floating point number in degrees. This is followed by "TYPE IN EASTERN OMEGA LIMIT" which the user similarly answers. The limits may be entered either with the convention 0 through  $\pm 180^\circ$  or 0 through  $360^\circ$ . In either case, the angle is measured positive eastward. (Note that it is permissible for the Eastern limit to be numerically smaller than the Western limit.) If the user is particularly interested in foreign areas, applying omega limits will generally be required since the heavy weighting given U.S. passes may swamp out the weight achieved on foreign passes.

Next the program asks the user to "INPUT NUMBER OF ORBITS TO BE RETURNED (I2 FORMAT)". Recall that the scan option will display the "n" best revs. This question allows the user to specify "n".



Finally the program displays the same data displayed following the MAXWGT option and asks if it is OK?. The user responds just as he would for the MAXWGT option.

The computer then scans the revolutions, determines the best ones and displays a summary of the scan by printing the rev number, THSTRT and THENEND for the best Z local vertical pass location, and the constrained and unconstrained weight achieved by the pass.

The user should keep in mind that when the SCAN option is completed, the information in the SCORE matrix and in THSTRT and THENEND pertains to the last revolution evaluated which may or may not be one of the listed best. In order to further study one of the best revolutions, the evaluate option should be called for that revolution.

#### ACCEPT or A

The ACCEPT option simulates the effect of actually performing a Z local vertical pass by adding the accomplishments of the pass to what has previously been accomplished (accepted). For details, see the descriptions of the OMECOV, PASS, TOTSCR, and PASDAT arrays in the section entitled WORKING DATA ARRAYS AND VARIABLES.) After receiving the ACCEPT option, the program asks: "ACCEPT REV = IREV, PERWT = "PERWT" OK?", where IREV is the number of the revolution most recently evaluated and "PERWT" is the current value of PERWT. A negative response to this question cancels the ACCEPT option. (IREV can be changed via the EVALUATE option and PERWT can be changed via the CHANGE option.) If a positive response is received, the program displays the values of THSTRT and THENEND and asks "OK?". A negative response will enable the user to change the theta limits of the pass prior to acceptance. Otherwise the current values are used to update the required arrays and variables.

#### CHANGE or C

The CHANGE option affords the user the opportunity to change a number of control variables via a namelist input. The user is first given the opportunity to see the current values displayed before he is requested to supply the new values. The variables which can be changed are APASDE, ARCMIN, IDEVIC,



IERSAV, IEROLD, MCATDE, MISION, PERWT, THMADE, and THMIDE. The definitions of these variables will be found in the section labeled VARIABLES IN COMMON SECTION [CONTRL].

In essence, the CHANGE option, with the display, is invoked automatically at program initiation. At any other time, changing IEROLD via the CHANGE option will be ignored. IEROLD is the number of the data unit on which the results of a previous session are stored. Since the constrained weight accomplished by a revolution is a function of what has already been accepted, attempting to add accomplishments from a prior session to the accomplishments of the current session would simply produce garbage and is, therefore, not allowed. All other parameters may be meaningfully changed either at the beginning or during the session and these changes are heeded.

#### PRINT or P

The PRINT option enables the user to have printed any of a number of data displays. He selects which particular display by entering the name of a print option. The names and a brief description of the currently available print options are given below. In a few cases, the print options require additional data from the user. The program will request this data if required.

In a few cases, the amount of output printed depends on the parameter IDEVIC. If IDEVIC = PRNTR, substantially more output will be provided than if IDEVIC=TERMNL.

In some cases, the user may abort the printing (by depressing the ATTN key on an IBM 2741 Terminal) (if the CINOTE subroutine works at the particular installation). The program will skip the detailed print, continue processing quietly, print a short summary, and then return control to the user. If CINOTE is not working, depressing the ATTN key may abort the program (at least on Bellcomm's UNIVAC 1108 it does). There seems to be a problem with CINOTE when using the teletype terminal at Bellcomm. Depressing the BREAK key, which should be equivalent, simply suspends processing. A carriage return then causes it to resume as though nothing happened. The PRINT OPTIONS:

ALDONE: Prints a list of the FO's which have been completely accomplished by the revolutions accepted so far  
(Abortable)



- ERESUM: For every proposal which can be at least partially worked on the currently considered mission (the list depends on the value of the parameter MISSION) the constrained and unconstrained weight accomplished on the most recently evaluated revolution, and the accumulated accomplished weight and the total possible weight are printed. (See Samples of Printout.) If IDEVIC=TERMNL, only the total weights for each proposal are listed. IF IDEVIC = PRNTR, every FO is listed. A summary is printed at the end. (Abortable)
- MISSUM: Prints a summary of the accomplishments accepted so far for the mission currently being considered. The weight accomplished (WGT SAT), the total weight possible (TOT WGT POS), the number of FO's which have been at least partially satisfied (NON ZERO FOS), and the total number of FO's in the mission category (TOT FOS) are printed by mission category. Totals are printed as well.
- NOWORK: Provides a list of every FO on which nothing has yet been accomplished. A summary is printed at the end. (Abortable)
- PRGSUM: Essentially identical to MISSUM except data for the mission categories not included in the currently considered mission are printed as well.
- PASSUM: Provides a summary of the passes which have been accepted so far. For each pass accepted, the following is printed: The revolution number (REV NO), the geographic longitude of the ascending equatorial crossing of the revolution (OMEGA), the GMT of the start of the Z local vertical pass expressed as month (MO), date (DAY) and hour (HR), theta at the start (THSTRT) and end (THEEND) of the pass, and the sun elevation angle at the start and end of the pass.
- PRPSUM: For every FO of a specified proposal, the constrained and unconstrained weight which can be accomplished on the most recently evaluated revolution, the total weight accomplished (accepted) so far and the total



possible weight are printed. Required additional input - the proposal number.

- PSCORE: Prints the entire contents of the SCORE array. (See section entitled WORKING DATA ARRAYS AND VARIABLES.)
- REVSUM: Prints a summary of what can be accomplished on the most recently evaluated revolution. If IDEVIC="PRNTR", an elaboration of the SCORE matrix is printed. For each FO which has a non-zero unconstrained weight for the revolution, the following is printed: the proposal number (PROP), the FO number (FO), theta at the start (TSTRT) and end (TEND) of the pass over the applicable properly lighted ground site, the total possible weight of the FO (TWEIGH), the unconstrained (WGT U) and constrained (WGT C) weight accomplished on the rev, the mission category of the FO (MCATEG), and the weight accomplished (accepted) so far (CURWGT). Then, regardless of IDEVIC, a summary of the contents of the SCORE array, as limited by the selected Z local vertical pass theta limits, is printed. This data is summarized by mission categories and totals and includes the constrained and unconstrained weight accomplished, the accumulated weight, the total weight possible, and the number of FO's with non-zero unconstrained weights.

#### STOP or HALT

This option terminates the current session. If IERSAV>0, the accomplishments, as of the end of the current session are written on logical unit "IERSAV" and the program so states before terminating. If IERSAV=0, the program tells the user that "PROPOSAL ACCOMPLISHMENTS WILL NOT BE SAVED OK?". An affirmative response simply terminates the program. A negative response causes the program to request the number of the logical unit on which to write the data.

WARNING - Either way of writing on unit IERSAV involves a rewind operation before writing. Anything previously stored on that unit will be overwritten. NOTE: The IERSAV file (as well as the IEROLD file) must be catalogued and a number assigned (via



an @USE card) prior to the beginning of program execution. There is no way to save the data if a file has not been assigned.

### BASIC ASSUMPTIONS AND ALGORITHMS

In order to effectively use the program, the mission planner should understand some of the basic assumptions and algorithms used in the program.

#### The Omega-Theta Transformation

While the basic ground site data is stored in terms of the latitude and longitude of the site corner points, it is transformed to omega-theta space for use by the ground site coverage routine FLYOVR. This omega-theta mapping describes points on earth in terms of the parameters of the orbit passing over them. Omega is the geographic longitude of the ascending equatorial crossing of the orbit which passes directly over the ground site. Theta is the angle, measured in the instantaneous orbit plane, from the ascending equatorial crossing to the point where the spacecraft is directly over the ground site. The spacecraft is said to be on the ascending leg of the orbit when its velocity vector has a northward component. Consequently,  $-90 < \theta < 90$  degrees on the ascending leg and  $90 < \theta < 270$  degrees on the descending leg. If the absolute value of the latitude of the ground site corner point is less than the inclination of the orbit plane, the point will map into two omega-theta pairs - one corresponding to the orbit which passes over the point on the ascending leg, the other pair corresponds to a descending pass. If the absolute value of the latitude is equal to the inclination, a single omega-theta pair is obtained. If the absolute value of the latitude is greater than the inclination, no values of omega-theta result since the orbit does not pass over the point.

The purpose of using the transformation is to simplify and speed up the computations made in subroutine FLYOVR. Knowing the extremes of omega for the corner points of a site and the omega of a particular orbit of interest for example, one can immediately determine whether or not the orbit passes over the ground site.

The use of the transformation does have one rather awkward implication. Sites which lie just outside the latitude



boundaries imposed by the orbit inclination, but which may still be viewable within the field of view of the instruments are not properly handled. Special logic, (not currently in FLYOVR), would have to be added to detect when such sites are viewable. Sites which straddle the critical latitudes are effectively handled in FLYOVR. A simple, but probably adequate, artifice for properly handling sites outside the latitude boundaries, but which should be in the field of view when the spacecraft is at its latitude extremes, would be to redefine the ground site corner point latitudes so that the sites do straddle the critical latitude.

The equations used to determine omega-theta from latitude-longitude are given below.

For  $\dot{L} > 0$

$$\theta = \sin^{-1} \frac{\sin L}{\sin i}$$

$$\Omega = \lambda - \tan^{-1} (\cos i \tan \theta) + R\theta$$

For  $\dot{L} < 0$

$$\theta = 180 - \sin^{-1} \frac{\sin L}{\sin i}$$

$$\Omega = \lambda - \tan^{-1} (\cos i \tan \theta) + R\theta + 180$$

where  $L$  = geocentric latitude of the point

$\lambda$  = geocentric longitude of the point

$i$  = orbit inclination

and  $R$  = the orbit precession of the longitude of the ascending equatorial crossing (earth rate + inertial precession) in terms of degrees precession per degree travel of the spacecraft.

These equations assume a circular, fixed inclination, fixed period orbit. The orbit inclination and precession terms could be made input variables but they currently are fixed in the program. The equations would have to be changed for a substantially non circular orbit. (The site data currently used is actually expressed in terms of geodetic latitude though



the above equations assume geocentric latitude. This anomaly should probably be corrected.)

The basic program does not depend on working in terms of omega-theta space but subroutine FLYOVR would have to be rewritten in order to change. The transformation is also quite useful when working with ground tracks graphically. Figures 1 and 2 depict the world and the United States in omega-theta coordinates for the nominal Skylab orbit. In this projection, ground tracks are represented by vertical straight lines. (These figures will be helpful when one wishes to limit the range of omega considered in the SCAN option or study the geographic areas overflown for a particular revolution.)

#### Revolution Definition

A revolution is defined as starting when the spacecraft is at the southernmost point in its orbit ( $\theta = -90$  degrees) and ending the next time it is at the southernmost point of the orbit ( $\theta = 270$  degrees). This latitude to latitude (essentially equivalent to a nodal orbit) definition was selected in favor of the longitude to longitude definition commonly used by MSC because it features a constant period revolution. Some computations were simplified this way. The "strange" beginning and end points were selected because there were no sites near 50 degrees south latitude. Thus, the awkwardness of having a contiguous pass occur on two revolution numbers was avoided. A small portion of South America does extend south of 50°S but currently, at least, no sites are defined in that region.

#### Groundsite Coverage - Overflight Rules

The algorithms used to decide whether the site required by a given FO will be covered on a given revolution require explanation. The logic discussed here is contained in Subroutine FLYOVR. Associated with each FO is a list of the instruments which should be turned on to provide the required data. The field of view of the required instrument having the narrowest field of view is used in the coverage computations.

The range of omega is determined for the required ground site. If the omega of the particular revolution falls within the site range of omega, the pass is immediately deemed acceptable in terms of adequate coverage. If it does not, additional





computations are made to determine the adequacy of coverage. First the extent of the field of view in terms of omega is computed. Consider the two points at either end of the edge of the field of view, (along a line orthogonal to the velocity vector) when the spacecraft is positioned at a value of theta equal to the middle of the theta range of the site. The omega values of these points are called the extent of the field of view in terms of omega. Figure 3 shows the omega range of the field of view of the S190 equipment as a function of theta. If the omega range of the site is larger than the field of view, the site is called a large site. The coverage algorithm requires that the spacecraft fly directly over a large site in order for the pass to be counted. If however, the site is small, at least  $1/4$  of the site must be within the field of view for the pass to count.

If the pass goes directly over the site, the theta limits of the pass are the values of theta when the subvehicle point enters and leaves the site. If an acceptable pass does not go directly over the site and the site extends beyond the edge of the field of view, the theta limits are defined as the values of theta when the subvehicle point of a pseudo-spacecraft flying on an orbit with an omega equal to the omega of the edge of the field of view enters and leaves the ground site. If an acceptable pass does not go directly over the site and the site will be entirely in view, the pass theta limits are equal to the theta limits of the site.

#### Ground Site Coverage - Lighting Rules

The logic described here is contained in Subroutine PASBAD. Consider the extent of the site measured in terms of the range of orbit arc. At least "ARCMIN" degrees of the overflight must be adequately lit or at least  $1/2$  the theta range of the pass must be adequately lit. If either of these criteria are satisfied, the pass is acceptable. If necessary, the theta limits of the pass are adjusted so that they only cover the range of the adequately lit ground site.

#### Ground Site Coverage - Z Local Vertical Pass Limiting

The subroutine PASBAD logic is reapplied to each FO when considering what is accomplished by a Z local vertical pass. This logic attempts to ensure that coverage is not counted if



the selected theta limits for the pass cut off too much of the coverage. This checking is done in several print routines, and under the MAXWGT and ACCEPT options.

#### Constrained vs. Unconstrained Weighting

The unconstrained weight accomplished on a pass is the portion of the FO weight accomplished without regard to what has already been accomplished. If N passes are required to satisfy an FO and W is the total FO weight, a single acceptable pass will be given an unconstrained weight equal to  $W/N$ .

The constrained weight accomplished on a pass takes into account what has been accomplished on previously accepted revolutions. If  $W_{TP}$  is the total weight possible for the FO,  $W_U$  is the unconstrained weight accomplished on a pass, and  $W_A$  is the previously accumulated FO weight, the constrained weight is equal to either  $W_U$  or  $W_{TP} - W_A$ , whichever is least.

The MAXWGT (and SCAN) option optimizes the placement of the Z local vertical pass on the revolution on the basis of the constrained weight accomplished.

#### Mission Categories

The FO's have been split into categories according to the mission(s) on which they can be satisfied (seasonal constraints): 2 only, 3 only, 4 only, 2 or 3, 2, or 4, 3 or 4, 2 or 3 or 4, and none. While numerous print options display data broken down into these categories, the only processing effect of the categorization occurs in the MAXWGT option (and therefore SCAN). Maximization will be done considering only those FO's with category numbers included in the value of "MCATEG". For example, if MCATEG=23, only the weights accumulated from FO's in categories 2, 3, or 23 are considered. If MCATEG=234, the accomplished weights for every FO are considered.

#### Repeat Coverage

The following logic applies to the determination of the acceptability of a pass for an FO requiring repeat coverage. If no pass has previously been accepted which covers the FO, any otherwise acceptable pass will be counted. Once at least one pass has been accepted, the dates of all subsequent otherwise



acceptable passes will be compared with the already accepted passes to see that they have acceptable date spacing. The algorithm is designed such that the longer the required interval between passes, the more tolerance is allowed the acceptable spacing. Basically, a 10% slop is permitted on the spacing. If for example, 5 day spacing is required, the interval between passes must be an integer multiple of 5 days  $\pm$  .5 day. If 20 day spacing is allowed, the interval between passes is permitted to be an integer multiple of 20 days  $\pm$  2 days.

There does exist a flaw in the pass spacing logic which future users may wish to fix. The logic permits a subsequent pass so long as it occurs on a date which is an even multiple of the required day spacing away from a previously accepted pass. For example, if the passes are required 5 days apart, a pass 20 days later than an already accepted pass would be acceptable. If the proposer has requested 4 passes, 5 days apart, this is acceptable, but if he requested 2 passes 5 days apart it probably isn't. Additional logic incorporating the number of passes required might be desirable. This logic would be added to Subroutine FOSCAN.

#### WORKING DATA ARRAYS AND VARIABLES

Basic working data is stored in a number of arrays and variables and is available to the various subroutines via labeled common. The contents, source, and use of the arrays and variables will be described here. The names in brackets indicate the labeled common section in which the variables will be found. The labeled common sections are actually defined in PDP elements which are added to the subroutines as required by the Fortran INCLUDE statement.

#### EREP [EREPC]

The EREP array is essentially a copy of the EREPD file except that the entries for each FO have been reassigned so that all FO's which are in the same mission category, 2, 23, etc., are contiguously represented in EREP. The lines representing FO's which cannot be accomplished because the season constraints do not match any of the missions, are not read into EREP. EREP is dimensioned 15 x 1000. The 15 entries for each FO are identical to the corresponding entries in the EREPD file. Once



the EREP array is filled by a file read in subroutine COMMON, during program initialization, it is never altered by the program. EREP is used by a majority of the major subroutines.

#### SITE [EREPC]

Site contains the corner points of each of the ground sites referenced by the various proposals expressed in terms of the latitude and longitude of the points. Although the latitude information supplied by the proposers as well as Bellcomm is actually expressed as geodetic latitude, it is currently treated as though it were geocentric. This minor anomaly can be most efficiently cured by either converting the latitude information on the file to geocentric or by converting it at the time it is read in subroutine COMMON. (The latter approach would leave the file data directly compatible with most maps.) SITE is dimensioned 2 x 5000 and is used by FLYOVR.

#### TRAJ [EREPC]

TRAJ contains the basic trajectory information required. Only the data required for the mission currently being studied is maintained in TRAJ. When the variable MISSION is changed, COMMON refills TRAJ with the appropriate data. TRAJ is dimensioned 5 x 700. Row 1 contains the revolution number; row 2 contains the geographic longitude of the ascending equatorial crossing; row 3 contains the value of Beta, (the angle between a vector pointing at the sun and the orbit plane at the time of the crossing); row 4 contains the time, measured in days, from 0000 GMT January 1; and row 5 contains the value of theta at orbit noon. The external routine EREP.MISSION reads the basic trajectory file, separates it into the three chunks required by the program and excludes those revolutions on which the magnitude of Beta is excessive (greater than 50 degrees currently). TRAJ is used by MAIN, WGTADD, and FOSCAN.

#### SCORE [SCOREC]

The SCORE array stores a summary of the FO's which can be satisfied on the most recently evaluated revolution. For each FO which can be satisfied, six parameters are recorded.

IEROW - The column number of the FO within the EREP array.



- TSTART - The value of theta at the start of coverage.
- TEND - The value of theta at the end of coverage.
- PWGHU - The unconstrained partial weight accomplished on this revolution - unconstrained means without regard for any previous accomplishments on this FO.
- PWGHC - The constrained partial weight accomplished on this revolution - the constrained weight takes into consideration what has previously been done on this FO and consequently represents the value of new accomplishments.
- MCATEG - The mission category of this FO, for example, MCATEG = 2 if this FO can be satisfied only during the SL-2 mission. MCATEG can equal 2,3,4,23,24,34, or 234.

SCORE is written in subroutine FOSCAN, and is used by numerous print routines and the WGTADD routine. The parameter NSCORE (also carried in SCOREC) contains the number of valid columns in SCORE for the current revolution. SCORE is currently dimensioned 6 x 100. If more than 100 FO's can be satisfied on the revolution, a printed alarm message will occur and only 100 columns will be filled.

#### TOTSCR [SCOREC]

The TOTSCR array is dimensioned 1 x 1000. Each column corresponds to the FO listed in the same column number of the EREP array. The entries in TOTSCR represent the partial FO score accumulated under the ACCEPT option so far. Values in TOTSCR can range from zero to the total possible score for the FO. TOTSCR is written by subroutine WGTADD and is used by numerous print routines and by FOSCAN to determine the constrained score achieved by a revolution under consideration.

#### PASDAT [SCOREC]

PASDAT has as many columns as there are FOs requiring repeating type coverage. For each column, the first row contains the column number of the applicable FO in the EREP array, the second row contains the total number of passes which have been accepted



and which contributed to the coverage requirements of the FO. The remaining rows list the dates on which those accepted passes were made. This data is used by FOSCAN to determine whether a revolution under consideration, which otherwise satisfies the FO, is properly separated in time from previously accepted revolutions. PASDAT is written in subroutine WGTADD.

#### OMECOV [SCOREC]

The OMECOV array currently exists but is not used. Its intended use is similar to that of PASDAT except that it is associated with the FOs which require total area coverage. The geographic longitude of the ascending equatorial crossing of accepted revolutions would be stored in place of the pass date. OMECOV would be written in WGTADD and used by FOSCAN (or, perhaps FLYOVR) to determine whether a new pass accomplished new coverage.

#### PASS [SCOREC]

The PASS array stores a summary of the local vertical passes which have been accepted so far. PASS is dimensioned 7 x 200, so up to 200 passes may be accepted before overflow occurs. The data stored in each row is as follows:

1. IREV - The revolution number of the pass.
2. OMEGA - The geographic longitude of the ascending equatorial crossing of the revolution.
3. DATE - The time of the start of the pass measured in days from 0000 January 1.
4. THSTRT - Theta at the start of the pass (in degrees).
5. THEND - Theta at the end of the pass (in degrees).
6. ELS - Sun elevation angle at the start of the pass (in degrees).
7. ELE - Sun elevation angle at the end of the pass (in degrees).



PASS is written in WGTADD and is used by various print routines.

#### INDEX [EREP]

INDEX contains pointing information to locate the corner points of a site in the SITE array. INDEX is dimensioned 2 x 600, each column containing the pointing data for the corresponding site number. The first row contains the SITE array column number of the first corner point of the site and the second column contains the number of corners describing the site. INDEX is written by the external routine EREP.SITES, is brought in with the other input data files, and is used by subroutine FLYOVR.

#### IERPNT [EREP]

IERPNT contains information which points to the location limits of proposal data in the EREP array for each of the mission categories. IERPNT is dimensioned 7 x 2. Each row contains data pertaining to the mission categories, 2, 23, 24, 234, 3, 34, 4, respectively. The first column lists the EREP column number of the first FO in the mission category, the second column lists the column number of the last FO. IERPNT is written by the external data preparation routine MSORT, is brought in with the EREP file, and is used by COMMON and various print routines.

#### IEREP [CNSTNT]

IEREP is essentially identical to IERPNT except that it contains pointing information for only those FO's which are to be considered for the current mission. IEREP is dimensioned 4 x 2, corresponding to the four mission categories considered for a given mission. IEREP is written in COMMON and used by FOSCAN and various print routines.

#### MCAT [CNSTNT]

MCAT is dimensioned 4 x 1 and contains the mission category values applicable for the currently considered mission. If MISSION = 2 for example, the four entries in MCAT would be 2, 23, 24, and 234. MCAT is written in COMMON and is used by various print routines.



### PRLIST [CNSTNT]

PRLIST is a list of the proposal numbers which contain FO's which can be satisfied on the currently considered mission. It is dimensioned 200 x 1, is written in COMMON, and is used by various print routines.

### VARIABLES IN COMMON SECTION [CONTRL]

These variables can all be easily changed either at the time of program initiation or via the CHANGE option. Most of them are assigned nominal values by Subroutine COMMON which hold unless they are changed by the user. Variables ending in DE indicate default variables. A similarly named variable will be set to the value of the default variable unless the user specifies a different value at the time it is to be used.

<u>NAME</u>	<u>DESCRIPTION</u>	<u>NOMINAL VALUE</u>
<u>APASDE</u>	The default value of ARCPAS which is the arc length of a Z-local vertical pass under the MAXWGT and SCAN options.	60 degrees
<u>ARCMIN</u>	The spacecraft must be over an adequately lighted site for at least ARCMIN degrees for the pass to be counted unless the site is less than 2*ARCMIN long in terms of theta.	1.0 degree
<u>IDEVIC</u>	Identifies the user's output device and controls the volume of print under certain print options. Can be equal to 'PRNTR' or 'TERMNL'.	TERMNL
<u>IERSAV</u>	= 0, Do not save accomplishments at the end of the computer run, (A last chance to save the data will be offered following entry of the STOP option.)  >0, save mission accomplishments on unit "IERSAV" at the end of the computer run.	0





<u>IEROLD</u>	= 0, Initialize the current run from scratch	0
	 >0, Initialize the current run as of the end of a previously saved run. Necessary data is stored on logical unit "IEROLD." IEROLD is only meaningfully non-zero at program initiation. Changing the value of IEROLD via the CHANGE option will be ignored.	
<u>MCATDE</u>	The default value of MCATEG. Only FO's which are in mission categories included in MCATEG will be considered in optimizing the location of the Z-local vertical pass under the MAXWGT or SCAN options. If MCATEG = 23, for example, only FO's in categories 2, 3, or 23 will be considered.	234
<u>MISION</u>	Indicates the mission currently being studied. Can be set to "2" or "3" or "4".	2
<u>PERWT</u>	Scores accomplished on an accepted pass will be multiplied by PERWT before being added to TOTSCR.	1.
<u>THMADE</u>	The default value of THMAX. THMAX is the maximum value of theta which will be considered in evaluating a revolution under option EVALUATE, MAXWGT, or SCAN.	270.
<u>THMIDE</u>	The default value of THMIN. THMIN is the minimum value of theta which will be considered in evaluating a revolution under options EVALUATE, MAXWGT, or SCAN.	-90.

VARIABLES IN COMMON SECTION [EREPC]

<u>NAME</u>	<u>DESCRIPTION</u>
<u>ITOTRJ</u>	The number of revolutions on the trajectory file for the currently considered mission. This parameter is read in with the trajectory data and is computed and written by the auxiliary data routine



MISION. The number of revolutions on which Beta is too large are not included in the ITOTRJ count. Used in FOSCAN.

NBETA1 The revolution number of the first revolution excluded from the mission by high beta. Also written by MISION, and read in with the trajectory data.

NBETA2 The revolution number of the last revolution excluded from the mission by high beta. Also written by MISION, and read in with the trajectory data (TRAJ). Used in FOSCAN.

LINE The column number of the trajectory data pertaining to a revolution being evaluated in the TRAJ array. Computed and used in FOSCAN.

VARIABLES IN COMMON SECTION [SCOREC]

<u>NAME</u>	<u>DESCRIPTION</u>
<u>NSCORE</u>	The number of valid entries in the score matrix, equal to the number of FO's which could be covered on the most recently evaluated revolution regardless of prior accomplishments.
<u>IPASS</u>	The number of passes which have been accepted so far--indicates the number of valid columns in the PASS array. Written by WGTADD.
<u>IDAY</u>	The integer value of the day of the year on which the current revolution occurs. Strictly speaking, IDAY is a local variable in FOSCAN which probably should not be in Common.

VARIABLES IN COMMON SECTION [CNSTNT]

<u>DTOR*</u>	The multiplicative conversion factor from degrees to radians. Set to .01745329 in COMMON.
<u>RTOD</u>	The multiplicative conversion factor from radians to degrees. Set to 1/DTOR in COMMON.
<u>XINCL*</u>	The orbit inclination expressed in degrees. Set to 50. in COMMON.

---

\* These variables are not set or included in common section [CNSTNT] in the attached listings - See the section entitled KNOWN ANOMALIES AND INCONSISTENCIES.



PI\* Set to 3.14159265 IN COMMON.

EARRAT\* The rotation rate of the earth in degrees per hour. Set to 15.04104953 in COMMON.

PRERAT\* The inertial precession rate of the orbit expressed in degrees per hour. Set to - .21174988 in COMMON.

NUMPRO The number of proposals which have FO's which can be accomplished in the currently considered mission = number of valid entries in PRLIST. Computed in COMMON and used in some print routines.

ORBPOR The orbit nodal period expressed in minutes. Set to 93.283019 in COMMON.

HAS190 The half angle field of view of the S190 cameras. Set to 10.6 degrees in COMMON. Used in LONRAN and FLYOVR.

HAS191 The half angle field of view of the S191 equipment. Set to 20.0 degrees in COMMON. Used in LONRAN and FLYOVR.

HAS192 The half angle field of view of the S192 equipment. Set to 5.0 degrees in COMMON. Used in LONRAN and FLYOVR.

HAS193 The half angle field of view of the S193 equipment. Set to 12.4 degrees in COMMON. Used in LONRAN and FLYOVR.

HAS194 The half angle field of view of the S194 equipment. Set to 7.5 degrees in COMMON. Used in LONRAN and FLYOVR.

RE\* The radius of the earth expressed in feet. Set to 20925738.2 in COMMON. Used by LONRAN.

RORB\* The radius of the orbit expressed in feet. Set to 22353625.3 in COMMON. Used by LONRAN.

VARIABLES IN COMMON SECTION [PDATA]

IREV The revolution number to be evaluated or most recently evaluated. This is supplied by the user



when requesting the EVALUATE option or by the program when the SCAN option has been requested. Used by FOSCAN, WGTADD, and various print routines.

THSTRT The value of theta at the start of a Z-local vertical pass expressed in degrees. THSTRT is automatically written by WGTMAX in the MAXWGT option but may be changed by the user for print options and at the time of accepting a pass.

THEND Similar to THSTRT except it is the value of theta at the end of the pass.

#### OTHER VARIABLES COMMONLY SEEN BY THE USER

IFDAY The first day to be considered in the SCAN option where day 0 corresponds to 0000 to 2400 GMT on January 1. Supplied by the user.

ILDAY The last day to be considered in the SCAN option. Definition is otherwise the same as IFDAY.

WGTMXC The maximum new FO weight which can be accomplished on a revolution during a pass of ARCPAS degrees taking into account the weight already accomplished. WGTMXC is computed and displayed by WGTMAX following completion of the MAXWGT option.

WGTMXU The unconstrained weight which could be accomplished assuming the same location and length of pass which produced WGTMXC. Also computed and displayed by WGTMAX. NOTE: WGTMXC is maximized with the MAXWGT option, not necessarily WGTMXU.

#### PROGRAM AVAILABILITY

Three copies of the program currently exist. One is on saved files at Bell Laboratories' Murray Hill New Jersey location. This copy is associated with the newly installed Univac 1106 serving the Cost Systems Studies Center. A second copy, written on magnetic tape has been delivered to the Mission Planning and Analysis Division at the Manned Spacecraft Center, Houston, Texas. Mr. Kenneth Young is the point of contact. The third copy, also on tape, has been delivered to Mr. Don M. Turner,



Space Station Task Force, NASA Headquarters, Washington, D.C. Each copy includes relocatable and symbolic versions of the necessary routines, including the auxiliary data set-up routines, and all necessary input data files. Instructions for reading the necessary information from the tapes are included in the section entitled PROGRAM USER'S GUIDE. A copy of the crude flow charts from which several of the subroutines were written has been delivered to Mr. Max Kilbourn of the Mission Planning and Analysis Division at MSC.

#### PROGRAM USER'S GUIDE

This User's Guide will assume the user has one of the two copies of the tape available. The version stored at the BTL Murray Hill, UNIVAC 1106 has some of the data and programs on differently named files than is assumed here. The tape version is more compact in its use of files.

#### Instructions for Reading the Program Tape onto Fastrand Files

The fastrand files must initially be assigned after the RUN card has been submitted. There are three ways to assign the files:

- (1) @ASG,CP FILENAME. where FILENAME is the name of the file. This assign statement causes FILENAME to be catalogued if the run terminates normally (C), as a public (P) file which subsequently can be referenced in any run by using the same project field on the run card as is currently being used.
- (2) @ASG,T FILENAME. This specifies that the file is to be temporary. It will exist only from the time it is assigned until run termination.
- (3) @ASG,AX FILENAME. This specifies that the file is currently catalogued (A) and that this run will have exclusive use (X) of the file until this run is terminated. This is the type of assign statement to use to assign a file catalogued during a previous run.

The required file names to be assigned are EREP, EREPD, SL2DAT, SL3DAT, SL4DAT, NEWANO, and ERDATA. Each must be assigned in an individual assign statement.



The format of control cards to be submitted to read the program tape into files is given below:

```
@RUN      RUN-ID,ACCOUNTING,PROJECT,RUNNING TIME, PAGES
@ASG,CP(or AX or T)      EREP
@ASG,CP(or AX or T)      EREPD
@ASG,CP(or AX or T)      SL2DAT
@ASG,CP(or AX or T)      SL3DAT
@ASG,CP(or AX or T)      SL4DAT
@ASG,CP(or AX or T)      NEWANO
@ASG,CP(or AX or T)      ERDATA
@ASG,TM      A, T, XXXX (where XXXX is the tape number,
                        T indicates a temporary assign-
                        ment, and M specifies that the
                        tape is to be read in medium
                        density, 556 bits per inch)

@COPIN      A.,EREP. (This copies file 1 of the tape
                  into program file EREP. It contains the
                  Earth Resources Experiment Analysis Program
                  in the form of 33 symbolic, 27 relocatable
                  and 1 absolute elements. The element names
                  can be listed by @PRT,T EREP).

@COPIN      A.,EREP. (File 2 of the tape. This adds to
                  the EREP file the card-image row data ele-
                  ment NEWDAT containing the array of site num-
                  bers and the corresponding array of corner
                  points.)

@COPIN      A.,EREPD. (File 3 of the tape. This copies
                  the card-image raw proposal information
                  matrix into EREPD.EREPD)

@COPY,G      A.,SL2DAT. (File 4 of the tape. This copies
                  into SL2DAT, the trajectory information for
                  SL-2. The data is already formatted to be
                  read by the main program)

@COPY,G      A.,SL3DAT. (File 5 of the tape. This copies
                  into SL3DAT the trajectory information for
                  SL-3. The data is already formatted to be
                  read by the main program)
```



@COPY,G      A.,SL4DAT. (File 6 of the tape. This copies into SL4DAT the trajectory information for SL-4. The data is already formatted to be read by the main program)

@COPY,G      A.NEWANO. (File 7 of the tape. This copies into data file NEWANO, the Bellcomm generated trajectory data from which SL2DAT, SL3DAT, and SL4DAT are created)

@COPY,G      A.,ERDATA. (File 8 of the tape. This copies into data file ERDATA the proposal matrix sorted by FO's into mission categories and formatted to be read by the EREP program)

One additional file, LATLON, is required which was not included on the tape. This file, however, can be created from other data in the tape. In addition to writing the file, the following control cards will yield a printout describing the groundsites in both latitude/longitude coordinate and omega/theta coordinates.

```
@ASG,CP (or AX or T)      LATLON
**@ASG,T                  THOMEG
@USE                      2, LATLON
**@USE                    3, THOMEG
@ASG,AX                  EREP
@FOR                      EREP.SITES, SITES
@XQT
@ADD                      EREP.NEWDAT
```

After the necessary files have been created from the tape, the runs described below may be executed.

## 2. Executable Runs

A. To execute the main program:

```
@RUN                  RUN-ID,ACCOUNTING,PROJECT,RUNNINGTIME,PAGES
@ASG,AX                EREP
*@ASG,AX               SL2DAT
*@ASG,AX               SL3DAT
*@ASG,AX               SL4DAT
```

---

\*\* The statement WRITE (3,OUT2) can be safely removed from the SITES program since the file THOMEG is no longer used by the main program. If it is removed, the indicated cards can be deleted.



```
*@ASG,AX  ERDATA
*@ASG,AX  LATLON
*@ASG,T   TFILE   (This is a temporary file for
                   scratch writing)
*@USE     2,SL2DAT
*@USE     3,SL3DAT
*@USE     4,SL4DAT
*@USE    10,ERDATA
*@USE    11,LATLON
*@USE    15,TFILE
  @ASG,AX   MSC1   (These two files may be catalogued
  @ASG,AX   MSC2   and assigned by the user in order
  @USE    20,MSC1   to save the current mission status
  @USE    21,MSC2   at the end of an execution or to
                   initialize a run from the mission
                   status recorded during a previous
                   execution.
@XQT      EREP.ERMAP
```

B. To list the total weight assigned each proposal:

```
@RUN      RUN-ID,ACCOUNTING,PROJECT,RUNNINGTIME,PAGES
@ASG,AX   EREP
@ASG,AX   EREPD
@FOR      EREP.TOWGT,TOWGT
@XQT
@ADD      EREPD.EREPD
```

C. To create and list trajectory files for SL-2, SL-3, SL-4 from a Bellcomm - generated ephemeris file:

```
@RUN      RUN-ID,ACCOUNTING,PROJECT,RUNNINGTIME,PAGES
@ASG,AX   EREP
@ASG,AX   SL2DAT
@ASG,AX   SL3DAT
@ASG,AX   SL4DAT
@USE     2,SL2DAT
@USE     3,SL3DAT
@USE     4,SL4DAT
@ASG,AX   NEWANO
@USE     9,NEWANO
@FOR      EREP.MISION,MISION
@XQT
```

---

\* For the sake of convenience, one command  
@ ADD EREP.FILES  
may be typed in place of the indicated commands. EREP.FILES is  
a data element containing all of the indicated commands and,  
when added, causes the commands to be executed.





D. To sort and list FO's by mission category:

```
@RUN      RUN-ID,ACCOUNTING,PROJECT,RUNNINGTIME,PAGES
@ASG,AX    EREP
@ASG,AX    ERDATA
@ASG,AX    EREPD
@USE       10,ERDATA
@FOR       EREP.MSORT,MSORT
@XQT
@ADD       EREPD.EREPD
```

3. Instructions for changing data files

A. Adding or changing proposal data

- a. Modify element EREPD located on file EREPD
- b. Execute a conversion run:

```
@RUN      RUN-ID,ACCOUNTING,PROJECT,RUNNINGTIME,PAGES
@ASG,AX    EREP.
@ASG,AX    ERDATA
@ASG,AX    EREPD
@USE       10,ERDATA
@FOR       EREP.MSORT,MSORT
@XQT
@ADD       EREPD.EREPD
```

B. Adding or changing site data

- a. Modify element NEWDAT located on file EREP
- b. Execute a conversion run:

```
@RUN      RUN-ID,ACCOUNTING,PROJECT,RUNNINGTIME,
          PAGES
@ASG,AX    LATLON
**@ASG,T    THOMEG
@USE       2,LATLON
**@USE      3,THOMEG
@ASG,AX    EREP
@FOR       EREP.SITES,SITES
@XQT
@ADD       EREP.NEWDAT
```

---

\*\* See previous footnote concerning file THOMEG.



C. Adding or changing trajectory data

There is no card-image file containing trajectory data. Therefore, modifications can be made only by regenerating or reprocessing the data. In either case, it must appear in final form on the trajectory file (i.e. SL2DAT, SL3DAT, or SL4DAT) as one binary record containing:

- a. The integer number of revolutions included on the file
- b. The integer number of the first rev with the absolute value of beta greater than the maximum value allowed.  
(If the absolute value of beta is never greater than the maximum value allowed, set this number and (c) to zero).
- c. The integer number of the last rev with the absolute value of beta greater than the maximum allowed. See (b).
- d. A floating point array dimensioned 5x700 where, for the Kth entry
  - (1,K) is the rev number
  - (2,K) is the geographic longitude (degrees) of the ascending equatorial crossing
  - (3,K) is the value of beta (the angle between the sun vector and the orbit plane)
  - (4,K) is the time of the ascending equatorial crossing expressed in days from 0000 GMT on January 1
  - (5,K) is the value of beta at orbit noon

D. Changing mission dates and running with the Bellcomm - supplied trajectory data.



In element MISION of file EREP, modify as required the values of SL2STR, SL2END, SL3STR, SL3END, SL4STR, SL4END to the desired start (suffix STR) and end (suffix END) times (in days from zero time on January 1). If the maximum allowed magnitude of beta is to be different than 50°, the statement:  
 "IF (ABS(BETSUN).LT.50) GO to 60" should be changed to reflect the new limit.

b. Execute a conversion run:

```
@RUN      RUN-ID,ACCOUNTING,PROJECT,RUNNINGTIME,
          PAGES
@ASG,AX   SL2DAT
@ASG,AX   SL3DAT
@ASG,AX   SL4DAT
@USE      2,SL3DAT
@USE      3,SL3DAT
@USE      4,SL4DAT
@ASG,AX   NEWANO
@USE      9,NEWANO
@FOR      EREP.MISION,MISION
@XOT
```

#### 4. Brief Dictionary of Terms and Variables Required by a User

<u>VARIABLE</u>	<u>DESCRIPTION</u>	<u>NOMINAL VALUE</u>
APASDE	Central Arc of ZLV passes	60.
ARCMIN	Number of degrees of site which must be within spacecraft field of view in order to be considered an acceptable contact	1.0
IDEVIC	User output device. Controls volume of printout = TERMNL or PRNTR	TERMNL
IERSAV	= 0, Do not save mission status at end of computer run  > 0, Save mission status on unit "IERSAV" at end of computer run	0



IEROLD	= 0, Initialize current run from scratch	0
	> 0, initialize current run at previously generated mission status read from unit "IEROLD"	
IFDAY	First day to be scanned for best revs in terms of FO weights accomplished. Day 0 begins at 0 hours, January 1. Required input for the scan option	
ILDAY	Last day to be scanned for best revs. See IFDAY	
FO	Flight objective. Each proposal is broken down into distinct objectives. Each objective is given a weight so that the weight of all FO's of a proposal = 100	
MCATDE	Mission category with respect to which a pass is optimized	234
MISION	Mission being studied	2
PERWT	Weighting factor to be applied to FO weights achieved during a pass	1.0
THEND	Value of theta at the end of the optimum arc of a rev. Calculated by the MAXWGT option. Value can be changed by user	
THMADE	Largest value of theta to consider in optimizing a pass or evaluating a revolution	270.
THMIDE	Smallest value of theta to consider in optimizing a pass or evaluating a revolution	-90.
THSTRT	Value of theta at the beginning of the optimum arc of a rev. See THEND	
WGTMXC	Maximum weight which can be accomplished by a rev during a pass of APASDE degrees,	



taking into account work already accomplished on the FO's under consideration

WGTMXU      Maximum weight which can be accomplished by a rev during a pass of APASDE degrees, not taking into account work already accomplished on the FO's under consideration

5. Brief Description of the Available Options (and Spelling)

Evaluate or Eval or E

Evaluate a designated revolution in terms of which, and how many, of the EREP FO's can be accomplished. Put the results in the SCORE array.

MAXWGT OR M

Determine the range of arc which maximizes the F.O. weights accomplished by a rev.

SCAN OR S

SCAN every rev from IFDAY to ILDAY (inclusive) within specified OMEGA limits, if any, and select the N best revs in terms of the maximum F.O. weights accomplished.

ACCEPT OR A

Cause the program to simulate performing the EREP pass. Update current status information to account for accomplishments.

PRINT OR P

According to which suboption is submitted, print information summary of one of the following:

- Revolution
- Proposal
- Current Mission
- Status



See following page for spelling and description of the print suboptions.

#### CHANGE OR C

Change the value of one or more control variables such as the mission or the default value of the length of the arc of ZLV passes.

#### STOP OR HALT

Save current mission status if specified, and exit from program.

### 6. PRINT SUBOPTIONS

<u>OPTION</u>	<u>DESCRIPTION</u>
ALDONE	List the FO's which have been completely accomplished.
ERESUM	Provide a summary of the conditional, unconditional, and accumulated weights accomplished for every proposal being considered for current mission.
MISSUM	Summarize the current mission status by mission category in terms of the number of FO's worked on and the weights accumulated.
NOWORK	List each FO for which the accumulated weight is zero.
PASSUM	Summarize the pass data accepted so far
PRGSUM	Summarize the accumulated FO weights accomplished and the total number of FO's worked on for each category of each mission.
PRPSUM	Print the conditional, unconditional, and accumulated weights for a given proposal.
PSCORE	Print the score array.
REVSUM	Print a summary of the score array.

#### SUBROUTINE DESCRIPTIONS

The following pages describe each of the program subroutines. The program entry point or main routine is included under the title MAIN. Appendix A contains the program listings.



### Subroutine ALLDON

**Purpose:** List all FO's relevant to the current mission which have been completely accomplished by accepted passes. Conclude with a summary, by mission category, of completed FO's.

**Calling Sequence:** Call ALLDON

**Common Sections Used:** CNTROL, EREPC, SCOREC, CNSTNT

**Calls:** CINOTE

**Called By:** MAIN

**Description:** ALLDON searches the list of FO's which can be accomplished on the current mission for those which have been completed. When a completed FO is encountered, the following information is printed: proposal number, FO number, mission category, total possible FO weight, and total accumulated weight. When the entire list has been searched, a summary of the above information is printed by mission category. The following data is given: mission category, total possible weight of FO's completed, total accumulated weight of FO's completed, and total number of FO's completed.

Should a console interrupt occur during the first stage of printing, that printing will be terminated. However, the remaining FO's will be scanned for completed entries, and the summary will be given.



### Subroutine CINOTE

Description: CINOTE is a UNIVAC 1108 Assembler Language subroutine which allows a user, operating in an interactive mode, to halt undesired printout without terminating program execution. The user does this by depressing a control interrupt key which causes control to switch from the print mode and to return to the next instruction in the subroutine causing the printing. A call to CINOTE must be made in the beginning of each subroutine during which the user might desire to effect an interrupt. The calling sequence is:

Call CINOTE (IPFLAG) where IPFLAG must be set to zero. The value of IPFLAG is changed by an interrupt, and provision is made for resetting it to zero.





### Subroutine COMMON

**Purpose:** Initialize control parameters and arrays, read in data base, set up COMMON sections. Change arrays when MISION is changed.

**Calling Sequence:** Call COMMON(ICHFLG) where ICHFLG = 0 indicates initial set-up of program, and ICHFLG = 1 indicates user's desire to change the mission under consideration.

**Common Sections Used:** CNTROL, EREPC, SCOREC, CNSTNT

**Calls:** None

**Called By:** MAIN

**Description:** Initially, COMMON assigns default values to a group of control variables. It then allows the user to change the value of one or more of these variables using a namelist input. When the variable values have been finalized, COMMON reads in the proposal matrix, the site array, and the trajectory information. It loads the relevant mission category numbers into the MCAT array and sets up pointers to the appropriate blocks of data in the proposal (EREP) matrix. Finally it fills array PRLIST with the proposal numbers for the current mission.

When COMMON is called with the value of ICHFLG = 1, the set of trajectory data for the new mission is read into the TRAJ array, overlaying the data already there. Additionally, the mission category numbers in the MCAT array and the pointers to the EREP matrix are revised. Finally, PRLIST is filled with the proposal numbers for the new mission.



### Subroutine ELTHET

**Purpose:** Given BETA and the value of theta at orbital noon, solve for the values of theta for a given sun elevation angle.

**Calling Sequence:** Call ELTHET(BETA, TNOON, EL, THETUP, THETDN, NOSUCH) where  
BETA is the angle between the solar vector and the orbit plane (degrees).  
(input)

TNOON is the value of theta at orbit noon (degrees). (input)

EL is the sun elevation (degrees).  
(input)

THETUP is the value of theta at which the sun elevation is EL degrees and rising. (output)

THETDN is the value of theta at which the sun elevation is EL degrees and falling. (output)

NOSUCH = 0 unless there is no value of theta corresponding to EL degrees sun elevation, in which case it is set to 1. (output)

**Calls:** SIN, COS, ACOS

**Called By:** FOSCAN

**Description:** ELTHET determines the arc length from orbital noon to a given sun elevation. It then subtracts this value from theta noon to yield THETUP and adds it to theta noon to determine THETDN and returns to the calling routine. If no theta values can be found for a given sun elevation, flag NOSUCH is set to 1 and control returns to the calling routine.



### Subroutine FLYOVR

**Purpose:** Determine whether a given revolution passes over, or in view of, a site. If it does, compute theta limits of pass (and, if total coverage is required, the limits of omega effectively covered\*).

**Calling Sequence:** Call FLYOVR (OMEGA, ISITE, FOV, ITOTAL, IOVER, TSTRT, TEND, OMMIN, OMMAX) where  
OMEGA is the longitude of the ascending equatorial crossing. (degrees) (input)

ISITE is the site number. (input)

FOV is the half angle field of view of the instrument with the narrowest field of view required by the FO. (degrees) (input)

ITOTAL = 0, total coverage of site not required

= 1, total coverage of site required (input)

IOVER = 0, rev does not pass over site

= 1, rev passes over site on ascending leg

= 2, rev passes over site on descending leg

= 3, rev passes over site both ascending and descending (output)

TSTRT(2) value of theta at start of pass over site. TSTRT(1) contains value of theta for ascending pass. TSTRT(2) contains value

---

\* Not implemented.



of theta for descending pass.  
(output)

TEND(2) value of theta at end of pass  
over site. (output)

OMMIN(2) minimum value of omega at which  
site can be viewed. OMMIN(1)  
contains value of omega for as-  
cending pass. OMMIN(2) contains  
value of omega for descending  
pass. (output)

OMMAX(2) maximum value of omega at which  
site can be viewed. (output)

Common Sections Used: EREPC, CNSTNT

Calls: ASIN, COS, LONRAN, SIN, SLOPE

Called By: FOSCAN



### Subroutine FLYOVR

Description: The corner points of the site being considered are extracted from the SITE matrix and are converted to omega-theta coordinates, ascending and descending. Next the maximum and minimum omega values, both ascending and descending, are determined.

The flow of events described next occurs twice, once for the ascending portion of the orbit and once for the descending portion: The average value of theta of the site is calculated and used in a call to LONRAN to determine the edges of the field of view in terms of omega limits. If the orbit passes directly over the site, that is, if the omega of the revolution falls within the omega range of the site, the pass is considered acceptable, and the theta limits of the pass are computed as the values of theta when the subvehicle point enters and leaves the site. (If the site is shaped such that the subvehicle point enters, leaves, enters again and then leaves again, all in one leg of the orbit, only the first entry and last exit values are used.) If the spacecraft does not pass directly over the site and the site is large--defined as having an omega range greater than the omega range of the field of view--the pass is not counted. If the site is small and at least  $1/4$  of the omega range falls within the field of view, the pass is counted. The theta range of a small site which straddles the edge of the field of view is computed as the values of theta when the subvehicle point of a pseudo spacecraft on an orbit with omega equal to the edge of the field of view enters and exits the site. If a small site is not directly overflown but lies entirely within the field of view, the largest and smallest values of theta for the site are used for the pass theta limits. The value of IOVER



is set according to the coverage acquired as defined in the calling sequence above. After both the ascending and the descending portions of the orbit have been considered, control returns to FOSCAN.



### Subroutine FOSCAN

**Purpose:** Scan the EREP matrix to determine which FO's can be satisfied on a given revolution.

**Calling Sequence:** Call FOSCAN

**Common Sections Used:** CNTROL, EREPC, SCOREC, CNSTNT, PDATA

**Calls:** CINOTE, ELTHET, FLYOVR, PASBAD, SUNEL, TIMDAT

**Called By:** MAIN, MASCAN

**Description:** The trajectory data of the rev being considered is extracted from the TRAJ array, and the date of the rev is converted from days since zero hours, January 1, 1973, to month, day and hours by a call to TIMDAT. Control then enters a loop which scans all of the mission FO's for possible accomplishments by the rev.

First the date of the rev is checked against the months desired by an FO. If there is an agreement, the FO is considered further; if not, control loops to the next FO. Next, the flag ITOTAL is set, indicating either that the FO requires total coverage (ITOTAL = 1) or it does not (ITOTAL = 0).

The sensors desired by the FO are next considered, and the half-angle field of view of the spacecraft is set to the narrowest field of any of its desired sensors. This information, combined with the total coverage indicator and pertinent trajectory data, is used in the call to FLYOVR to determine if the FO site will be acceptably in view.

If the site is not overflown, no further consideration is given the FO. However, if the site is overflown, the subroutine checks whether the lighting conditions at the site meet



the FO specifications. If the lighting conditions reduce the usable range of theta of the overflight, the appropriate theta limits are adjusted and subroutine PASBAD is called to determine whether the good part of the pass is long enough to be acceptable. If it is, the program finally looks at the coverage requirements. Since no coding is implemented for total coverage, it is treated as a single pass requirement. For FOs specifying repeating coverage of the site, the date of the rev is checked against the dates, if any, of previous overflights of the site in order to determine if the pass occurs on an acceptable date spacing.

Next the constrained weight and the unconstrained weight accomplished by the pass over the site are determined. The unconstrained weight is calculated by dividing the total weight allotted the FO by the total number of overflights desired. The constrained weight is determined in the same manner, but, if the resultant weight plus the weight already accomplished is greater than the total possible weight, the constrained weight is set to the difference between the total possible weight and the total accumulated weight. The minimum constrained value = 0.

If a site meets all of the requirements for an acceptable pass, the following information for the rev is entered in the SCORE array: the line number of the FO in the EREF matrix, theta at the start of the pass over the site, theta at the end of the pass over the site, the unconstrained and constrained weights accomplished by the pass, and the mission category of the FO.





### Subroutine LATLON

**Purpose:** To convert coordinates expressed by latitude and longitude into theta/omega coordinates, with the effects of the earth's rotation and orbit regression rate included in the computation.

**Calling Sequence:** CALL LATLON (XLATC, XLON, ØMEGA1, ØMEGA2.)

XLATC      The latitude coordinate of a point, measured in a plane normal to the equator. (Degrees) (Input)

XLON      The longitude coordinate of a point, measured in the plane of the equator with respect to a reference meridian. (Degrees) (Input)

ØMEGA1    The longitude of the ascending equatorial crossing of an orbit whose ascending side crosses the point (XLATC, XLON). (Degrees) (Output)

ØMEGA2    The longitude of the ascending equatorial crossing of an orbit whose descending side crosses the point (XLATC, XLON). (Degrees) (Output)

**Common Sections Used:** CNSTNT

**Calls:** No other subroutine.

**Called By:** Subroutine LONRAN.

**Description:** The subroutine operates by first calculating the ascending and descending values of theta corresponding to the calling argument, XLATC; (The values of theta are functions of latitude only). The two values of omega, ØMEGA1



and OMEGA2, corresponding to the ascending and descending orbits which cross the point (XLATC, XLON) are then calculated. This involves subtracting two terms from the longitude (XLON) of the given point. The first is a geometric term which is simply the displacement in longitude between the ascending equatorial crossing and a point on the ground track at latitude XLATC; the displacement in longitude is bi-valued, one for the ascending and one for the descending side of the orbit. The second term is the displacement in longitude which ensues (due to earth's rotation and orbit regression) during the time the spacecraft travels through the control angle theta, again bi-valued.



### Subroutine LONRAN

**Purpose:** To determine the difference in the omega coordinates between a given point and the points at either edge of the field of view at that point. The edges of the field of view lie on a line orthogonal to the ground track.

**Calling Sequence:** CALL LONRAN (THETA, FOV, DOMN, DOMP.)

THETA	The angle between the ascending equatorial crossing and a point on the ground track, measured in the instantaneous orbit plane. (Degrees) (Input)
FOV	The half angle field-of-view of an EREP instrument. (Degrees) (Input)
DOMN	The negative difference in omega, with respect to the omega of the ground track. It is equivalent to the difference in omega between the given orbit and an orbit which passes over a point at the extreme edge (measured normal to the flight direction) of the field-of-view. The negative difference corresponds to the point at the left edge of the field-of-view for ascending orbits, and corresponds to the point at the right edge of the field of view for descending orbits. (Degrees) (Output)
DOMP	The positive difference in omega, with respect to the omega of the ground track. The positive difference corresponds to a point at the right edge of the field of view for ascending or-



bits, and to points at the left edge of the field of view for descending orbits. (Degrees)  
(Output)

Common Sections Used: CNSTNT

Calls: Subroutine LATLON.

Called By: Subroutine FLYOVR.

Description: The central angle from the spacecraft to the edge of the field of view is calculated based on the half angle field of view. Next the latitude and longitude of the point at the edges of the point of view are determined. They are computed on the basis of lying on a great circle orthogonal to the spacecraft orbit and intersecting it at the location of the spacecraft. Subroutine LATLON is then called to determine the values of omega of the field of view edge points. If the value of theta for the spacecraft indicates that it is on the ascending leg of the orbit, the ascending omega values for the edge of the field of view are used and vice versa. Finally the difference between the omega values and the reconstructed omega of the orbit are determined and returned to the calling routine.



## MAIN Routine

**Purpose:** Provide the starting point of the program and control the program flow by means of options input by the user.

**Common Sections Used:** CNTROL, EREPC, SCOREC, CNSTNT, PDATA, PRNT

**Calls:** ALLDON, CINOTE, COMMON, FOSCAN, MASCAN, NONDON, PPASS, PPROP, PSCRM, PTSCRM, PTSCR, THEMEX, WGTADD

**Description:** The MAIN routine is the program starting point. Initially it calls subroutine COMMON with the variable ICHFLG in the calling argument set to zero. This allows the current mission to be defined and the corresponding arrays and variables to be initialized.

After the program has been set up, MAIN accepts the first option from the user. As each option is carried out, control returns to this point to read the next option. There are seven options available to the user. After an option is read, control transfers to the section of MAIN appropriate for that option.

Following is a description of the program activity as a result of each option (underlined letters may be typed instead of the formal option name):

EVAL option. This causes a designated revolution to be evaluated in terms of which and how many of the EREP FO's can be accomplished. The user types the number of the rev he wishes evaluated. MAIN determines if the rev is valid for the current mission. If it is valid, MAIN calls FOSCAN to evaluate it. If it is invalid, an appropriate diagnostic is printed.



MAXWGT option. This determines the range of arc which maximizes the FO weights accomplished by a rev. The length of arc to be maximized, the mission category to be maximized, and the start and end values of theta to be considered are all set to the default values APASDE, MCATDE, THMIDE, THMADE, respectively. The user is then allowed to change any or all of the assigned values through input in namelist MAX. MAIN then calls THEMEX to determine the best range of arc.

SCAN option. This option causes a scan of every rev within specified time limits and within specified omega limits, if any, to select the n best revs in terms of maximum FO weights accomplished. The user is asked to type the first and last days during which he desires the revs scanned. Subroutine MASCAN is then called.

ACCEPT option. This causes the program to simulate performing the EREP pass by updating the current mission status information to account for the accomplishments. The user is shown the current values of the rev number and the multiplier PERWT. If they are not what he desires, he may so indicate, in which case control returns to read another option.

PRINT option. According to which suboption is submitted, the PRINT option causes information to be printed about one or more of the following: the current rev, a particular proposal, the current mission status. The available suboptions are explained on another page. MAIN reads the name of the suboption typed in by the user, and calls the appropriate subroutine to effect the desired information gathering and printing.

CHANGE option. This option allows the user to change one or more of the control variables. These include the central arc of a ZLV pass



(APASDE), the number of the mission (MISSION), the mission category on which to maximize a pass (MCATDE), the minimum and maximum values of theta to consider for pass optimization (THMIDE, THMADE), the weighting factor to be applied to FO weights achieved during a pass (PERWT), the unit number of the file on which to save the mission status (IERSAV), the unit number of the file from which to read a previous mission status (IEROLD), the output device (IDEVIC), and the number of degrees of a site which must be within the field of view to be considered an acceptable contact (ARCMIN). The program reads the changes through namelist CONTRL. If the mission number is changed by the user, MAIN calls COMMON with argument ICHFLG = 1 in order to acquire the correct trajectory data.

STOP (or HALT) option. This option allows the program to terminate after first having saved the current mission status if desired. If the user has assigned a value to IERSAV, the mission status is automatically written onto the unit and program execution is halted. However, if the user has not specified a value for IERSAV, he is given an opportunity to do so before execution stops.



### Subroutine MASCAN

**Purpose:** Examine every rev occurring from IFDAY through ILDAY and within specified omega limits, if any, determine which accomplish the most in terms of FO weight accomplished.

**Calling Sequence:** Call MASCAN (IFDAY, ILDAY) where ILDAY is the first day of the scan. January 1 at Greenwich is Day 0. (Input)  
ILDAY is the last day of the scan. (Input)

**Common Sections Used:** CNTROL, EREPC, PDATA

**Calls:** FOSCAN, THEMEX

**Called By:** MAIN

**Description:** First the user is allowed to specify omega windows for the scan about to be executed and is then permitted to change the values of four control variables (APASDE, MCATDE, THMIDE, THMADE) if so desired. The scan for best revs begins with the TRAJ matrix being searched for all revs occurring within the date and omega constraints. Each of the acceptable revs is evaluated and maximized by a call to FOSCAN and to THEMEX. The results are stored in an array, and, after all possible revs in the trajectory matrix have been considered, the array is scanned for the N entries exhibiting the highest constrained weights. These entries are printed for the user.

**Limitation** - an array (FLY) in MASCAN is currently sized to allow up to 700 revolutions to be considered. If more are contained between IFDAY and ILDAY, an error will most probably occur. This possible, but unlikely, event is not alarmed.





### Subroutine NONDON

Purpose: List the FO's which can be done on the current mission for which the accumulated weight equals zero.

Calling Sequence: Call NONDON

Common Sections Used: CNTROL, EREPC, SCOREC, CNSTNT

Calls: CINOTE

Called By: MAIN

Description: NONDON searches the accumulated score array (TOTSCR) for zero entries corresponding to FO's which can be accomplished during the current mission. For each entry that is found, the following information is printed: proposal number, FO number, mission category, and total possible FO weight. After the entire array has been searched, a summary of the above information is printed by mission category including category number, total possible score of the FO's in that category for which no weight has been accomplished, and total number of FO's in that category for which no weight has been accomplished.

Should a console interrupt occur during the printing of individual FO's, that printing will be terminated, but the summary will be given.



### Subroutine PASBAD

**Purpose:** Determine if the usable portion of a pass over, or in view of a site is acceptable in terms of the length of the usable pass compared to the size of the site. PASBAD is called by FOSCAN to determine whether bad lighting unduly restricts the length of the pass. The other calling routines use PASBAD to determine whether a selected Z-local vertical arc unduly restricts the usable pass.

**Calling Sequence:** Call PASBAD (TSTART, TEND, START, END, IGOOD) where

TSTART	is the value of theta at which the orbit starts a pass over, or in view of, the site (degrees) (input)
TEND	is the value of theta at which the orbit concludes a pass over, or in view of, the site (degrees) (input)
START	is the value of theta at the start of a specified arc of the rev (degrees) (input)
END	is the value of theta at the end of a specified arc of the rev (degrees) (input)
IGOOD	= 0, specified arc does not cover enough of the site for the pass to be acceptable.  = 1, specified arc does cover enough of the site for the pass to be acceptable.

**Common Sections Used:** CNTROL



Calls: None

Called By: FOSCAN, THEMAX, WGTADD, PSCRM

Description: If either  $(END-START) > 1/2 (TEND-TSTART)$  or  $(TEND-TSTART) \geq ARCMIN$ ,  $IGOOD = 1$ . Otherwise  $IGOOD = 0$ .



### Subroutine PPASS

Purpose: Print a summary of the pass data accepted so far.

Calling Sequence: Call PPASS

Common Sections Used: SCOREC, CNSTNT

Calls: TIMDAT

Called By: MAIN

Description: For each accepted pass record contained in the PASS array, PPASS prints the rev number, the longitude of the ascending equatorial crossing, the month, day and hour at the start of the pass, theta at the start and end of the pass, and the sun elevation angle at the start and end of the pass.



### Subroutine PPROP

Purpose: Print constrained, unconstrained, and accumulated weights for a given proposal, IPROP, as of the current rev.

Calling Sequence: Call PPROP (IHDG, IFOPRN, IPROP)

IHDG	= 1, print rev number and column headings
	= 0, do <u>not</u> print rev number and column headings (input)
IFOPRN	= 1, print constrained, unconstrained and accumulated scores for each FO of the proposal. (input)
	= 0, print scores for complete proposal only.
IPROP	is the number of the proposal to be analyzed for constrained, unconstrained, and accumulated weights. (input)

Common Sections Used: CNTROL, EREPC, SCOREC, CNSTNT, PDATA

Calls: None

Called By: MAIN

Description: PPROP scans the EREP array for all FO's of the specified proposal. As each is found, the current SCORE array is searched for a corresponding entry. If one is located, the constrained and unconstrained weights for that FO are added to the array containing the summary information for the complete proposal. If the print flag, IFOPRN, is set to 1, the individual FO scores are also printed.



When the entire EREP array has been scanned for the FO's of the specified proposal, the summary scores for the proposal are printed.



### Subroutine PSCRM

**Purpose:** Print a summary by mission category of the FO's in the current SCORE array which can be worked on by a pass from THSTRT to THEND.

**Calling Sequence:** Call PSCRM

**Common Sections Used:** CNTROL, EREPC, SCOREC, CNSTNT, PDATA

**Calls:** PASBAD

**Called By:** MAIN

**Description:** PSCRM first determines if the output device, designated by the value of the variable IDEVIC, is 'PRNTR' which indicates a high speed printer. If it is, the SCORE matrix is scanned, and, for each entry, the following information is printed: proposal number, FO number, the values of theta at the beginning and end of the ground track pass over, or in view of, the site, the total weight of the FO, the unconstrained weight accomplished on this rev, the constrained weight accomplished on this rev, the mission category of the FO, and the accumulated weight already accomplished on the FO.

Next, regardless of the value of IDEVIC, the SCORE matrix is searched for FO's worked on during a pass from THSTRT to THEND. For each entry in SCORE, subroutine PASBAD is called to determine if the pass is acceptable. If it is acceptable, its accomplished weights are added to a summary table. After the entire SCORE array has been scanned, the summary table is printed, giving the following information for each mission category: total constrained weights accomplished, total unconstrained weights accomplished, total accumulated weight accomplished so far, total weight possible of FO's worked on, and total number of FO's worked on.



### Subroutine PTSCRM

Purpose: For the current mission, print the total number of FO's worked on and the accumulated weights accomplished.

Calling Sequence: Call PTSCRM

Common Sections Used: CNTROL, EREPC, SCOREC, CNSTNT

Calls: None

Called By: MAIN

Description: If the output device indicated by the value of IDEVIC is 'PRNTR', PTSCRM first prints, for each FO applicable to the current mission, the proposal number, the FO number, the total weight accomplished, and the total possible weight.

Regardless of the value of IDEVIC, the following summary is printed for each category of the current mission: the category number, the total accumulated weight for the category, the total weight possible for the category, and the total number of FO's worked on in the category.





### Subroutine PTSCRIP

Purpose: Print a summary of the accumulated FO weights accomplished during all three missions.

Calling Sequence: Call PTSCRIP

Common Sections Used: CNTROL, EREPC, SCOREC, CNSTNT

Calls: None

Called By: MAIN

Description: PTSCRIP scans through the entire EREP matrix accumulating weights for each mission category in a summary table. At the conclusion of the scan, it prints for each category the category number, the total weight accomplished, the total possible weight of all of the FO's in the category, and the total number of FO's worked on.



### Subroutine SLOPE

**Purpose:** Given the coordinates of two points in omega-theta space, form the equation of the straight line between them and determine the value of theta at the point on the line where omega is equal to a given value.

**Calling Sequence:** Call SLOPE (X1, Y1, X2, Y2, OM, THET)

X1, Y1	The omega-theta coordinates of
X2, Y2	the two points defining the line. (input)
OM	The longitude of the ascending equatorial crossing of the ground- track. (input)
THET	The value of theta at the inter- section of the groundtrack and the straight line. (output)

**Common Sections Used:** None

**Calls:** None

**Called By:** FLYOVR

**Description:** The slope and the ordinate intercept are determined for the line connecting the points (X1, Y1) and (X2, Y2). The value of THET is then calculated to be the result of adding the ordinate intercept to the product of OM times the slope.



### Subroutine SUNEL

Purpose: Solve for the sun elevation angle at a given value of theta.

Calling Sequence: Call SUNEL (BETA, TNOON, THETA, EL, NOSUCH)

where	BETA	is the angle between the solar vector and the orbit plane (degrees). (input)
	TNOON	is the value of theta at orbit noon. (input)
	THETA	is the value of theta at which the sun elevation is requested. (input)
	EL	is the calculated sun elevation angle at THETA. (output)
	NOSUCH	= 0, EL has been calculated = 1, EL is set to zero; no elevation can be determined for the given value of THETA. (output)

Common Sections Used: None

Calls: COS, ACOS, TAN, SIN



### Subroutine THEMAX

Purpose: Determine the best location for a z local vertical pass within a revolution in terms of maximizing the FO weight accomplished.

Calling Sequence: Call THEMAX (ARCPAX, MCATEG, THMAX, THMIN, WGTMXC, WGTMXU)

where	ARCPAS	is the length of pass. (degrees) (input)
	MCATEG	is the mission category to be maximized. (For example, if MCATEG = 234, all of the FO's in SCORE are used to maximize the pass weight. If MCATEG = 23, the mission FO's in categories 2, 3, and 23 are used to maximize the pass weight, etc.) (input)
	THMAX	Upper theta boundary of the rev to be scanned for the best pass. (degrees) (input)
	THMIN	Lower theta boundary of rev to be scanned for the best pass. If THMAX = THMIN = 0, the theta limits are set to THMADE and THMIDE respectively. (degrees) (input)
	WGTMXC	is the total constrained weight accomplished during the maximum weight pass. (output)
	WGTMXU	is the total unconstrained weight accomplished during the maximum weight pass. (output)

Common Sections Used: CNTROL, SCOREC, PDATA, PRNT.



Calls: PASBAD

Called By: MAIN, MASCAN

Description: The SCORE array is searched for entries corresponding to the mission category to be maximized. When an entry is found, the number of that entry in SCORE is placed in an array named THETA. After the entire SCORE array has been searched, the pointers in THETA are sorted to point to records in SCORE in ascending order of theta at the start of the pass over a site.

THEMAX then evaluates the constrained and unconstrained weight accomplished assuming the Z local vertical pass starts at the lowest value of theta in THETA and ends at that value plus ARCPAS degrees. This process is repeated for each value of theta in THETA.

Calls to PASBAD identify the entries in SCORE acceptable for the arc. The weights contained in the acceptable entries are added to the array TMAX which contains the constrained weight, unconstrained weight and theta start value for each arc evaluated.

After all possible arcs have been evaluated, the TMAX array is scanned for the entry with the maximum constrained weight. The value of THSTRT in common section PDATA is set to the theta start value of the entry. THENEND is set to THSTRT plus ARCPAS degrees. The constrained and unconstrained weights of the entry are returned to the calling routine as WGTMXC and WGTMCU respectively.



### Subroutine TIMDAT

Purpose: Convert time expressed as time elapsed from midnight on December 31 to month, days and hours.

Calling Sequence: Call TIMDAT (TIME, MO, IDAY, HR)

where	TIME	is days elapsed from midnight on December 31. (input)
	MO	is integer month. (output)
	IDAY	is integer day (output)
	HR	is floating point hours (output)

Common Sections Used: None

Calls: None

Called By: FOSCAN, PPASS

Description: MO is given a value from 1 to 12 depending upon the month in which TIME falls. IDAY is given the value of the integer number of days in the month corresponding to TIME. HR is given the value of the fractional part of TIME multiplied by 24.



### Subroutine WGTADD

Purpose: Simulate the effect of having performed an EREP pass.

Calling Sequence: Call WGTADD

Common Sections Used: EREPC, SCOREC, PDATA

Calls: PASBAD, SUNEL

Called By: MAIN

Description: WGTADD scans the SCORE array to find all FO's which can be worked on during a pass defined from THSTRT to THEND. At each iteration it calls PASBAD to determine if the pass is acceptable. If it is, it adds to the TOTSCR entry for that FO, the constrained weight held in the SCORE array.

If the acceptable FO has a requirement for a pass or passes spaced randomly, the program then loops to consider the next FO since nothing else needs to be recorded. The same thing happens if the FO requires total coverage of the site since no logic is currently implemented to handle total coverage. Finally, if repeating coverage of the site is required by the FO, appropriate entries must be made in the PASDAT array. This includes incrementing the counter contained in PASDAT for the FO representing the number of passes scheduled over the site. Also, the date of the current pass is added to the list of dates on which the passes previously accepted for the FO occurred.

After the entire SCORE array has been scanned, the PASS matrix is updated to reflect the current pass being accepted. IPASS, the counter indicating the number of passes which have been accepted, is incremented by one. The corresponding record in PASS is filled with the follow-



ing data: the number of the rev on which the pass occurs, the longitude of the ascending equatorial crossing, the date at which the rev begins, theta at the start of the pass, theta at the end of the pass, and the sun elevation angles at the start and end of the pass.





## KNOWN ANOMALIES AND INCONSISTENCIES

The writers currently know of only four program anomalies though it should be stressed that the available time has precluded thorough program testing.

The most important anomaly involves the times stored in the PASS array which are supposed to be the time of the start of the ZLV pass. The times currently stored (and printed under the PASSUM print option) are actually the times of the ascending equatorial crossing. To correct this error the statement

```
PASS(3,IPASS) = TRAJ(4,LINE)
in subroutine WGTADD should be changed to
PASS(3,IPASS) = TRAJ(4,LINE)+THSTRT*ORBPER/360.)/1440.
The orbit period, ORBPER, should either be defined by the
statement
ORBPER = 93.283019
or by adding the statement
INCLUDE BLK4, LIST
if ORBPER is defined in subroutine COMMON as discussed below.
```

The second anomaly also involves subroutine WGTADD. The constrained weight accomplished on a pass is supposed to be multiplied by PERWT before being added to the TOTSCR array but it is not. The following statement should be added before any executable code

```
INCLUDE BLK1, LIST
```

The statement:

```
TOTSCR(NPT) = TOTSCR(NPT) + SCORE(S,I)
should be changed to
TOTSCR(NPT) = TOTSCR(NPT) + SCORE(S,I)*PERWT.
```

Under the REVSUM option, the heading format does not permit a four significant digit revolution number. The statement:

```
3000 FORMAT(//T2,'SUMMARY OF REV',I3,',etc.
should be changed to
3000 FORMAT(//T2,'SUMMARY OF REV',I4,',etc.
```



The sun elevation angle data printed under the print suboption PASSUM is improperly formatted if a negative angle is involved. The second line of the 2000 FORMAT statement should be changed from

```
. F6.2,T44,F6.2,T52,F5.2,T60,F5.2)
to:
. F6.2,T44,F6.2,T52,F6.2,T60,F6.2)
```

There is a potential inconsistency in the program in that certain orbit sensitive parameters are multiply defined in the various subroutines. Since the program could be used for non-nominal Skylab orbits and for altogether different orbits as well (e.g. ERTS), these parameters should be singly defined and carried in the labeled common section CNSTNT. In addition, certain fixed constants should be added to CNSTNT and the definitions deleted from the subroutines. The following steps will accomplish this goal.

The PDP element EREP.COM should be edited to add the necessary constants to the common section CNSTNT. The following line should be added to the BLK4 PROC entry:

```
. PI,EARRAT,PRERAT,DTOR,RE,RORB,XINCL
```

(The description in the section labeled "variables in common Section CNSTNT " implies this has already been done but it hasn't.)

Several statements should be added to subroutine COMMON. It is suggested that they be added following the statement PERWT=1.0. If added at this point, prior to the display of the control variables, orbit sensitive parameters could easily be added to the display if desired. These variables could also be added to NAMELIST/CONTRL/ so they could be changed at program initiation (and to NAMELIST/CONTRL/ in the MAIN routine so they could be changed under the CHANGE option). The new statements are:

```
PI=3.14159265
EARRAT=15.04104953  @EARTH ROTATION RATE - DEG/HR
PRERAT=-.21174988  @INERTIAL PRECESSION RATE - DEG/HR
ORBPER=93.283019   @ORBIT NODAL PERIOD - MINUTES
DTOR=.01745329     @DEGREES TO RADIANS
```



RTOD=1./DTOR	@RADIANS TO DEGREES
RE=20925738.2	@EARTH RADIUS - FEET
RORB=22353625.3	@SEMI-MAJOR AXIS - FEET
XINCL=50.	@ORBIT INCLINATION - DEGREES

The following changes should be made to the indicated subroutines.

#### Subroutine ELTHET

Replace the statement:

RTOD=180./3.1415927  
with:  
INCLUDE BLK4, LIST

#### Subroutine FLYOVR

Add the statement  
INCLUDE BLK4, LIST  
Replace the statements beginning EARTH=, DTOR=,  
and RINCL=; with  
EARTH=(EARRAT-PRERAT)/(360./(ORBPER/60.))  
RINCL=XINCL\*DTOR

#### Subroutine LATLON

Remove the two line DATA statements and replace with:

INCLUDE BLK4, LIST  
SIN50 = SIN(XINCL\*DTOR)  
COS50 = cos(XINCL\*DTOR)  
DOMEGA = -(EARRAT-PRERAT)\*ORBPER/21600.  
(DOMEGA is the change in the longitude of the ascending  
equatorial crossing expressed in degrees (or radians)  
change per degree (radian) of spacecraft travel.)

#### Subroutine LONRAN

Remove the two-line data statement and replace with:

INCLUDE BLK4, LIST  
SINI = SIN(XINCL\*DTOR)  
COSI = COS(XINCL\*DTOR)



RATIO = RORB/RE  
E = (EARRAT-PRERAT)\*ORBP/21600.  
(E is the negative of the change in longitude of the ascending  
equatorial crossing.)

Subroutine SUNEL

Replace:

RTOD = 180.3.1415927  
with:

INCLUDE BLK4,LIST

Subroutine WGTADD

Add the statement

INCLUDE BLK4 LIST

The orbit parameters are also defined in the auxiliary routine on file EREP.NEWDAT and should be changed if those in the main program are changed. Failing to change the parameters in NEWDAT will not affect the main program since the site corner points are stored in terms of their latitude/longitude coordinates. However, the printout from NEWDAT listing the site corner points in omega/theta coordinates would be meaningless. The statements involved are

OMEGAO = 360./1.55471699 - the mean motion of the spacecraft  
in degrees per hour

REGRES = .21174988 - the negative of the inertial precession  
of the orbit line of nodes - degrees  
per hour.

XINCL = 50.0 - the orbit inclination in degrees.

When building the SL2DAT, SL3DAT, and SL4DAT trajectory files, the auxiliary routine on file EREP.MISION automatically discards those revolutions for which the absolute value of beta is greater than 50 degrees. If a different limit is desired, MISION must be changed. The statement which discards the revolution reads:



IF (ABS(BETSUN).LT.50) GO TO 60

If the limit is changed, the only main program change would be the second and third lines of the 600 FORMAT statement in subroutine COMMON. They now read:

```
. T2,'FIRST REV WITH ABS BETA GREATER THAN 50: ',I4,/  
. T2,'LAST REV WITH ABS BETA GREATER THAN 50: ',I4)
```

The comment cards describing the purpose of the SUNEL and ELTHET subroutines may be confusing. The code was changed slightly (to adapt them to this program) since the comment cards were written and they were not updated. The purpose of SUNEL should read:

PURPOSE: GIVEN BETA, AND THE VALUE OF THETA AT ORBIT NOON,  
SOLVE FOR THE SUN ELEVATION ANGLE AT THE GIVEN  
VALUE OF THETA

The description of ELTHET should read

PURPOSE: GIVEN BETA, AND THE VALUE OF THETA AT ORBIT NOON,  
SOLVE FOR THE VALUES OF THETA FOR A GIVEN SUN  
ELEVATION ANGLE

The subroutine VROTAT is included on the program tape but is no longer required. It was used with a previous version of LONRAN.

*D. A. Corey*  
D. A. COREY

*D. N. Sakolosky*  
D. N. SAKOLOSKY

*E. W. Radany*  
E. W. RADANY

DAC  
1025-DNS-dvw  
EWR

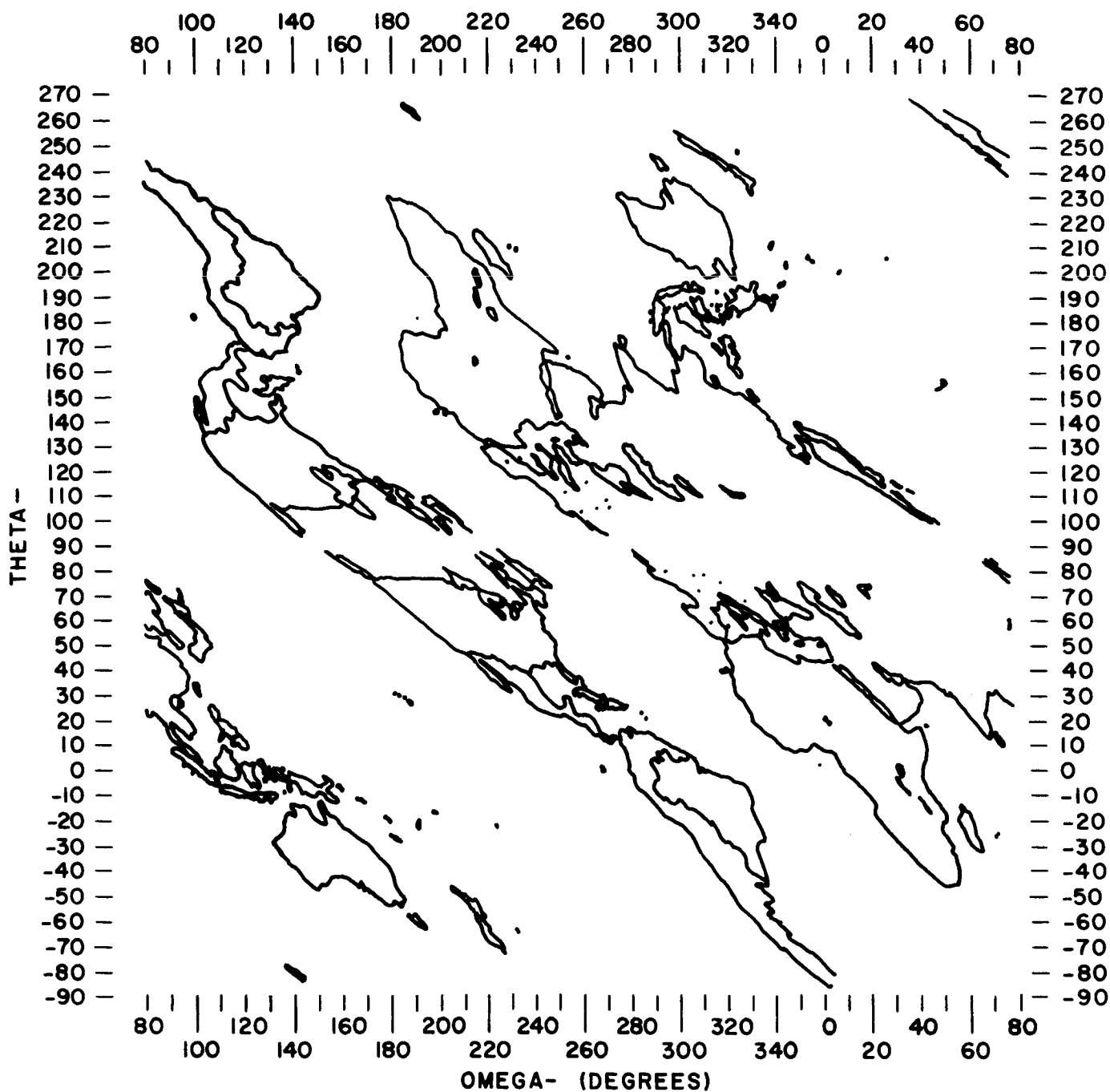


FIGURE 1 THE WORLD IN OMEGA THETA COORDINATES  
50° INCLINATION, 235 NM CIRCULAR ORBIT

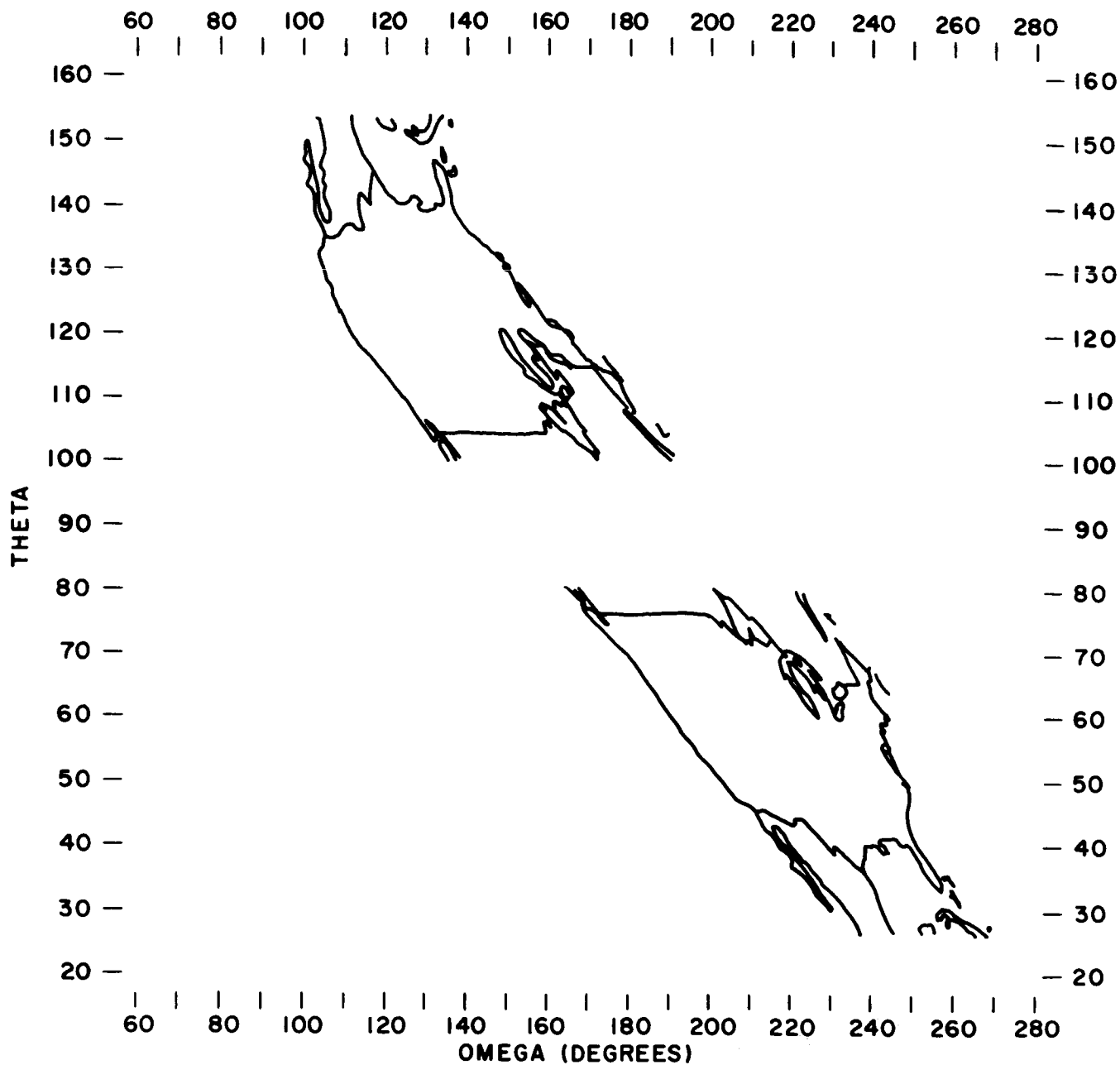


FIGURE 2 THE CONTINENTAL UNITED STATES IN OMEGA THETA COORDINATES 50° INCLINATION, 235 NM CIRCULAR ORBIT

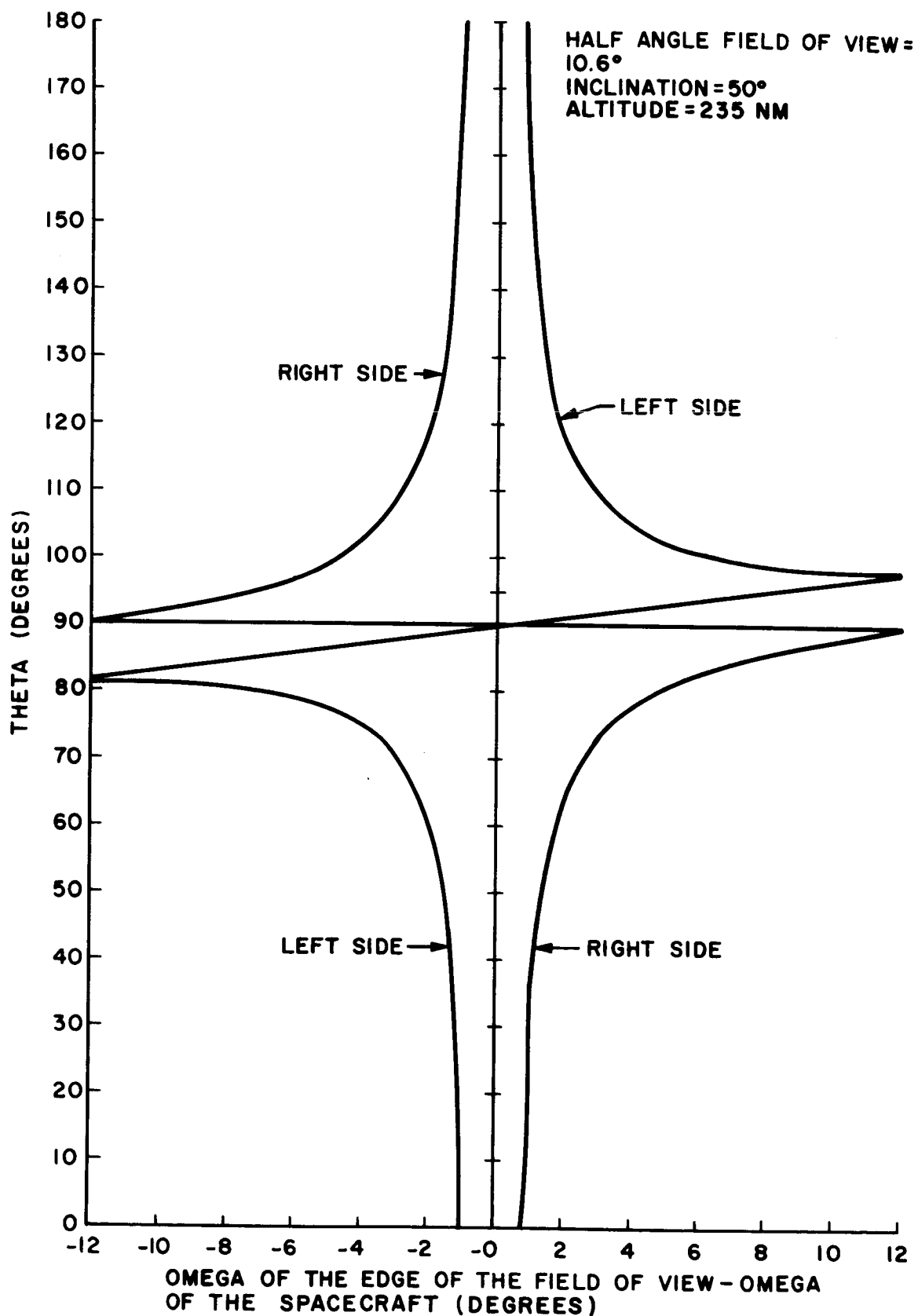


FIGURE 3 THE DIFFERENCE BETWEEN THE VALUE OF OMEGA AT THE EDGE OF THE FIELD OF VIEW AND THE VALUE OF OMEGA OF THE SPACECRAFT V.S. THETA FOR THE SI90 CAMERAS



APPENDIX A

SUBROUTINE LISTINGS

CUCARAU - COLL COREY FILE EREP•EREP

PAGE

DATE 041472

4

08 MAR 72 01:44:32 CYCLE: 0

E•ALLDON

```

1.      SUBROUTINE ALLOCN
2.C
3.C      PURPOSE:  LIST THE F.O.'S WHICH HAVE BEEN COMPLETELY
4.C                ACCOMPLISHED BY ACCEPTED PASSES.  AN ATTENTION
5.C                BUTTON WILL STOP PRINTING INDIVIDUAL
6.C                F.O.'S; SUMMARY STILL OBTAINED
7.C
8.C
9.C      INCLUDE BLK1,LIST
10.C     INCLUDE BLK2,LIST
11.C     INCLUDE BLK3,LIST
12.C     INCLUDE BLK4,LIST
13.C
14.C     DIMENSION PRINT(5,3), IPR(5)
15.C     EQUIVALENCE (PRINT(1,3),IPR(1))
16.C
17.C     DO 10 J = 1,5
18.C       DO 10 K = 1,3
19.C         PRINT(J,K) = 0
20.C       IPR(J) = 0
21.C
22.C     CALL CNOTE(IPR(J))
23.C     WRITE(6,2000)
24.C     DO 100 I = 1,4
25.C       I1 = IREP(I,1)
26.C       I2 = IREP(I,2)
27.C       DO 100 J = 1,12
28.C         IF(TOTSCR(J)•LT. EREP(15,J)) GO TO 100
29.C         PRINT(I,1) = PRINT(I,1) + EREP(15,J)
30.C         PRINT(I,2) = PRINT(I,2) + TOTSCR(J)
31.C         IPR(I) = IPR(I) + 1
32.C       IF(IPR(J)•NE. 0) GO TO 100
33.C       NPRO = EREP(1,J)
34.C       WRITE(6,2010) NPRO,NFO,NCAT(1),EREP(15,J),TOTSCR(J)
35.C     CONTINUE
36.C     WRITE(6,2050)
37.C     DO 300 J = 1,4
38.C       DO 200 K = 1,2
39.C         PRINT(5,K) = PRINT(5,K) + PRINT(J,K)
40.C       IPR(5) = IPR(5) + IPR(J)
41.C       WRITE(6,3000) (MCAT(J),(PRINT(J,K),K = 1,2),IPR(J),J = 1,4),
42.C         (PRINT(5,K),K = 1,2), IPR(5)
43.C     RETURN
44.C 2000  FORMAT(/T2,'COMPLETED F.O.S'//
45.C        • T2,'PROP',T8,'FO',T13,'MCAT',T14,'POS SCR',
46.C        • T29,'ACCOM SCR')
47.C 2010  FORMAT(T2,T4,T9,T2,T13,T13,T20,F6.2,T31,F6.2)
48.C 2050  FORMAT(/T2,'SUMMARY',T2,'MISSION CAT',
49.C        • T15,'TOT POS SCR',T24,'TOT ACCOM SCR',
50.C        • T45,'NOR FOS SCHLG')
51.C 3000  FORMAT(4(T6,T13,T17,F/,T32,F7.1,T50,I3/),
52.C        • T6,'TOTAL',T17,F7.1,T32,F7.1,T49,I4)
53.C     END

```

24 FEB 72 10:52:23 CYCLE: 0

## E.CINOTE

```

1..      CINOTE
2..  TITLE:
3..
4..  AUTHOR: C. MEE
5..
6..  DATE: 4/13/70
7..
8..  LANGUAGE: 110R ASSEMBLER
9..
10.. DESCRIPTION: CINOTE AND ITS COMPANION ROUTINE, CIJUMP, PERMIT FORTRAN
11.. USER PROGRAMS TO MAKE USE OF THE CONSOLE INTERRUPT
12.. (PRESSING THE ATTN KEY ON A REMOTE TERMINAL).
13..
14.. CINOTE ALLOWS A FORTRAN USER TO ESTABLISH A VARIABLE IN
15.. HIS PROGRAM AS A FLAG TO BE SET TO 1 WHENEVER A CONSOLE
16.. INTERRUPT OCCURS; EXECUTION THEN CONTINUES AS IF THE
17.. INTERRUPT HAD NOT OCCURRED. THE USER IS RESPONSIBLE FOR
18.. TESTING THE FLAG AT APPROPRIATE POINTS WITHIN HIS PROGRAM
19.. AND FOR CLEARING IT AFTER AN INTERRUPT HAS BEEN DETECTED.
20.. CINOTE IS CALLED VIA A FORTRAN STATEMENT OF THE FORM
21..
22.. CALL CINOTE(FLAG)
23..
24.. WHERE FLAG IS THE VARIABLE TO BE USED AS A FLAG TO RECORD
25.. INTERRUPTS.
26..
27.. CIJUMP PERMITS THE FORTRAN USER TO ESTABLISH A NUMBERED
28.. STATEMENT WITHIN HIS PROGRAM AS THE LOCATION TO WHICH
29.. CONTROL IS TO BE TRANSFERRED WHEN A CONSOLE INTERRUPT
30.. OCCURS. (THE WHITE-UP FOR SUBROUTINE CIJUMP GIVES
31.. FURTHER DETAILS.)
32..
33.. BOTH CIJUMP AND CINOTE MAY BE CALLED ANY NUMBER OF TIMES.
34.. EACH CALL OVERRIDES THE EFFECT OF THE PREVIOUS CALL TO
35.. EITHER ROUTINE.
36..
37.. SINCE CINOTE ALLOWS THE USER PROGRAM TO RETAIN CONTROL
38.. ON A CONSOLE INTERRUPT, PROGRAMS WHICH USE THIS ROUTINE
39.. CANNOT BE TERMINATED SIMPLY BY PRESSING THE ATTN BUTTON.
40.. A TASK WHICH GETS OUT OF CONTROL, FOR EXAMPLE BY GETTING
41.. INTO AN UNWANTED LOOP, CAN BE TERMINATED BY ASKING THE
42.. COMPUTER OPERATOR TO GIVE THE JOB AN X KEY-IN.
43..
44.. CINOTE IS COMPATIBLE WITH AND MAKES USE OF THE
45.. CONTINGENCY ROUTINE IN THE HELLCOMM FORTRAN I/O PACKAGE.
46.. BCHIO.
47..
48..
49..
50.. S(1),CINOTE.
51.. L
52.. S
53.. L,U
54.. S,H?
55.. J
56..

```

```

      GET ADDRESS OF USER'S FLAG
      AND SAVE FOR LATER USE.
      SET UP INTAD TO RECEIVE CONTROL
      IF ATTN KEY IS HIT.
      RETURN TO CALLER.

```

```

57. J, TAB L, U AD, 1 .
58. S AD, FLAGAD .
59. L, M2 AD, NFR, S .
60. S AD, NEXTI .
61. L AD, NCAS .
62. L ALL, CXII, .
63. E CFND, .
64. J •NEXTI .
65. .
66. S (0), FLA, AD + 0 .
67. NEXTI + 0 .
68. .
69. E, U .

```

SET USER'S FLAG = 1.

COMPUTE LOCATION  
AND SAVE FOR LATER USE.

RESTORE AD.

RESTORE ALL.

ENABLE SUBSEQUENT CONTINGENCIES.

GO TO USER'S NEXT INSTRUCTION.

ADDRESS OF USER'S FLAG.

ADDRESS OF USER'S NEXT INSTRUCTION.

07 MAR 72 22:32:40 CYCLE: 0

E.COM

1.BLK1 PROC

2.C

COMMON/CNTRL/APASDE,MCATDE,THMIDE,THMADE,IDEVIC,MISION,  
IERSAV,IEROLD,PERWT,ARCHIN

5.C

6. END

7.BLK2

8.C

COMMON/EREPC/EREP(15,1000),INDEX(2,600),SITE(2,5000),  
TRAJ(5,700),ITOTRJ, IERPNT(7,2), NBETA1, NBETA2,  
LINE

10.C

11. END

12.BLK3

13.C

COMMON/SCOREC/PASS(7,200),TOTSCR(1000),PASDAT(12,200),  
OMECOV(13,200),NSCORE,SCORE(6,100), IPASS, IDAY

18.C

19. END

20.BLK4

21.C

COMMON/CNSTNT/RTOD,NUMPRO,MCAT(4),PRLIST(200),ORBPFR,IERP(4,2),  
HAS190,HAS191,HAS192,HAS193,HAS194

24.C

25. END

26.BLK5

27.C

COMMON/PDATA/IREV,THSTRT,THEND

28.C

29. END

30. END

DATE 041472

08 MAR 72 01:45:40 CYCLE: 0

E.C.COMON

```

1.  SUBROUTINE COMMON(ICHFLG)
2.  INCLUDE BLK1.LIST
3.  INCLUDE BLK2.LIST
4.  INCLUDE BLK3.LIST
5.  INCLUDE BLK4.LIST
6.  NAMELIST/CONTROL/APASDE,MCATDE,THMIDE,THMADE,IDEVIC,
7.  MISION,IERSAV,IEROLD,PERWT,ARCMIN
8.C
9.  IF(ICHFLG.NE. 0) GO TO 100
10. APASDE = 60.0
11. MCATDE = 234
12. THMIDE = -90.0
13. THMADE = 270.0
14. IDEVIC = 'TERMINL'
15. MISION = 2
16. IERSAV = 0
17. IEROLD = 0
18. ARCMIN = 1.0
19. PERAT = 1.0
20. WRITE(6,5000) APASDE,MCATDE,MISION,THMIDE,THMADE,PERAT,
21. IERSAV,IEROLD,IDEVIC,ARCMIN
22. READ(5,5010) IANS
23. IF(IANS.NE. 'NO') GO TO 10
24. WRITE(6,5020)
25. READ(5,CONTROL)
26. 10 READ(10) IERPNT,EREP
27. READ(11) NENTR, INDEX, SITE
28. IF(IEROLD.EQ. 0) GO TO 100
29. READ(IEROLD) TOTSCR,PASDAT,OMECOV,PASS, IPASS
30. READ(MISION) ITOTRJ, NBETA1, NBETA2, TRAJ
31. IF(MISION.NE. 2) GO TO 130
32. DO 125 J = 1,4
33. DO 125 K = 1,2
34. IEREP(J,K) = IERPNT(J,K)
35. MCAT(1) = 2
36. MCAT(2) = 23
37. MCAT(3) = 24
38. MCAT(4) = 234
39. GO TO 150
40.C
41. 130 IF(MISION.NE. 3) GO TO 140
42. IEREP(1,1) = IERPNT(2,1)
43. IEREP(1,2) = IERPNT(2,2)
44. IEREP(2,1) = IERPNT(4,1)
45. IEREP(2,2) = IERPNT(4,2)
46. IEREP(3,1) = IERPNT(5,1)
47. IEREP(3,2) = IERPNT(5,2)
48. IEREP(4,1) = IERPNT(6,1)
49. IEREP(4,2) = IERPNT(6,2)
50. MCAT(1) = 23
51. MCAT(2) = 234
52. MCAT(3) = 3
53. MCAT(4) = 34
54. GO TO 150
55.C
56. 140 IEREP(1,1) = IERPNT(3,1)

```

```

57. IEREP(1,2) = IERPNT(1,2)
58. IEREP(2,1) = IERPNT(4,1)
59. IEREP(2,2) = IERPNT(4,2)
60. IEREP(3,1) = IERPNT(5,1)
61. IEREP(3,2) = IERPNT(6,2)
62. IEREP(4,1) = IERPNT(7,1)
63. IEREP(4,2) = IERPNT(7,2)
64. MCAT(1) = 24
65. MCAT(2) = 234
66. MCAT(3) = 34
67. MCAT(4) = 4
68. C
69. 150 ICHFLG = 0
70. HAS190 = 10.6
71. HAS191 = 20.
72. HAS192 = 5.
73. HAS193 = 12.4
74. HAS194 = 7.5
75. WRITE(6,6000) TRAJ(1,1), TRAJ(1,ITOTRJ), NBETA1, NBETA2
76. I = IEREP(1,1)
77. NUMPRO = 1
78. PRLIST(1) = EREP(1,1)
79. DO 400 I = 1,4
80. ISTART = IEREP(1,1)
81. IEND = IEREP(1,2)
82. DO 400 J = ISTART, IEND
83. IF(EREP(I,J) .EQ. PRLIST(NUMPRO)) GO TO 400
84. IF(NUMPRO .GE. 200) GO TO 500
85. NUMPRO = NUMPRO + 1
86. PRLIST(NUMPRO) = EREP(I,J)
87. 400 CONTINUE
88. RETURN
89. 500 WRITE(6,6050)
90. RETURN
91. C
92. C FORMATS
93. C
94. 5000 FORMAT(1/2,'CURRENT VALUES OF CONTROL VARIABLES ARE:','/
95. . T2,'APASDE = ',F9.4,T22,'MCATDE = ',I3,
96. . T42,'MISION = ',I3/
97. . T2,'TH*106 = ',F9.4,T22,'THMADE = ',F9.4,
98. . T42,'PERNT = ',F7.3/
99. . T2,'IERCAV = ',I3,T24,'IEROLD = ',I3,T42,'IDEVIC = ',A6/
100. . T2,'ARCIN = ', F6.3,5X,'OK?')
101. 5010 FORMAT(A6)
102. 5020 FORMAT(1/2,'ENTER CHANGED PARAMETERS, NAMELIST CTRL')
103. 6000 FORMAT(1/2,'FIRST REV = ',F5.0,5X,'LAST REV = ',F5.0/
104. . T2,'FIRST REV WITH A65 BETA GREATER THAN 50: ',I4/
105. . T2,'LAST REV WITH A65 BETA GREATER THAN 50: ',I4)
106. 6050 FORMAT(1/2,'.....ERROR NUMBER OF PROPOSALS ',
107. . 'EXCEEDS DIMENSION OF PRLIST ARRAY.')
```

END

DATE 041472

26 JAN 72 14:00:19 CYCLE: 0

CUCHEAN - CULL COREY FILE EREP.EREP

E=LLTHET

1. SUBROUTINE ELTHET(BETA,INOON,EL,THETUP,THETDN,NOSUCH)  
 2.C  
 3.C PURPOSE: GIVEN BETA, AND THE VALUES OF THETA WHEN THE SUN  
 4.C ELEVATION ANGLE IS 30 DEG. AND INCREASING AND 30 DEG.  
 5.C AND DECREASING (OR ANY SIMILAR PAIR), SOLVE FOR THE  
 6.C VALUES OF THETA FOR A GIVEN SUN ELEVATION ANGLE.  
 7.C

8.C  
 9. RTOD=180./3.1415927  
 10. NOSUCH=0  
 11. X=SIN(EL/RTOD)/COS(BETA/RTOD)  
 12. IF(ABS(X).GT.1)GO TO 30  
 13. X=ACOS(X)\*RTOD  
 14. THETUP=INOON-X  
 15. THETDN=INOON+X  
 16. IF(THETUP.LT. -90.) THETUP = THETUP + 360.  
 17. IF(THETDN.GT. 270.) THETDN = THETDN - 360.  
 18. RETURN  
 19. NOSUCH=1  
 20. THETUP=0.  
 21. THETDN=0.  
 22. RETURN  
 23. END

30



COCHRAN - CULL COREY FILE ENFP•EREP

E•ERMAP

1. L13  
2. I1

EREP  
MAIN

DATE 041472

PAGE

11

20 MAR 72 13:27:11 CYCLE: 0

22 FEB 72 12:32:37 CYCLE: 0

E.FILES

1.0ASG,AX SL2DAT  
2.0USE 2.SL2DAT  
3.0ASG,AX  
4.0USE  
5.0ASG,AX SL4DAT  
6.0USE 4.SL4DAT  
7.0ASG,AX ERDATA  
8.0USE 10.ERDATA  
9.0ASG,AX LATLON  
10.0USE 11.LATLON  
11.0ASG,T IFILE.,F  
12.0USE 15.TFILE

SL3DAT  
3.SL3DAT

16 MAR 72 14:58:18 CYCLE: 0

## E.FLYOVR

```

1. SUBROUTINE FLYOVR(OMEGA,ISITE,FOV,ITOTAL,IUOVER,TSTRT,IEND,
2. OMMIN,OMMAX)
3.C
4.C PURPOSE: DETERMINE WHETHER A REV FLIES OVER A SITE. IF SO,
5.C COMPUTE THETA LIMITS OF PASS AND, IF TOTAL COVERAGE IS
6.C REQUIRED, THE LIMITS OF OMEGA EFFECTIVELY COVERED.
7.C
8. INCLUDE BLK2,LIST
9. DIMENSION TSTRT(2), IEND(2), OMMIN(2), OMMAX(2), OMETH(30,4),
10. TH(30)
11.C
12.C CONVERT SITE CORNER POINTS TO OMEGA AND THETA,
13.C ASCENDING AND DESCENDING
14.C
15. NSITE = ISITE
16. EARTH = (15.04104953 + .21174988)/(1360./1.55471699)
17. DTOR = 3.14159265 / 180.
18. RINCL = 50.0 * DTOR
19. SINRIN = SIN(RINCL)
20. PI = 3.14159265
21. NPTS = INDEX(2,ISITE)
22. I1 = INDEX(1,ISITE)
23. I2 = I1 + NPTS - 1
24. DO 20 I = 1,NPTS
25. N = I1 + I - 1
26. IF(ABS(SITE(1,N)) .LT. 50.0) GO TO 15
27. IF(IOUT .NE. 1) GO TO 25
28. IF(SITE(1,N) .LT. 0) SITE(1,N) = -49.99
29. IF(SITE(1,N) .GT. 0) SITE(1,N) = 49.00
30. OETH(1,1) = ASIN(SIN(SITE(1,N) * DTOR)/SINRIN)
31. OETH(1,2) = PI - OMETH(1,1)
32. DO 17 J = 3,4
33. OETH(1,J) = SITE(2,N) * DTOR - ATAN2(COS(RINCL) *
34. SIN(OMETH(1,J-2)),COS(OMETH(1,J-2))) + EARTH * OMETH(1,J-2)
35. DO 18 J = 1,4
36. OMETH(1,J) = OMETH(1,J)/DTOR
37. DO 19 J = 3,4
38. IF(OMETH(1,J) .LT. 0) OMETH(1,J) = OMETH(1,J) + 360.
39. IF(OMETH(1,J) .GT. 360) OMETH(1,J) = OMETH(1,J) - 360.
40. CONTINUE
41. IOUT = 0
42. GO TO 50
43.C
44.C IF ALL POINTS OF SITE ARE OUT OF RANGE, IGNORE SITE.
45.C IF SOME POINTS ARE IN RANGE, SET ABS. VAL. OF OUTSIDE
46.C LATITUDES = 49.99
47.C
48. DO 30 I = 1,I2
49. IF(ABS(SITE(1,I)) .LT. 50) GO TO 40
50. RETURN
51. IOUT = 1
52. GO TO 5
53. IF(OMETH(1,1) .EQ. OMETH(NPTS,1) .AND.)
54. OETH(1,3) .EQ. OMETH(NPTS,3) .AND. NPTS .NE. 1) GO TO 60
55. NPTS = NPTS + 1
56. DO 55 J = 1,4

```

```

57. 55 OMETH(NPTS,J) = OMETH(1,J)
58.C DETERMINE MAX AND MIN OMEGAS FOR ASCENDING AND DESCENDING PASS
59.C
60.C
61. 60 OMMIN(1) = OMETH(1,3)
62. OMMA(1) = OMETH(1,3)
63. DO 65 J = 1,NPTS
64. OMMIN(1) = AMIN(OMMIN(1),OMETH(J,3))
65. OMMA(1) = AMAX(OMMA(1),OMETH(J,3))
66. IF((OMMA(1) - OMMIN(1)) .LT. 180 .OR. 1180 .EQ. 1)
67.   GO TO 75
68. DO 70 J = 1,NPTS
69. IF(OMETH(J,3) .LT. 180) OMETH(J,3) = OMETH(J,3) + 360.
70. 1180 = 1
71. GO TO 60
72. 75 1180 = 0
73. 60 OMMIN(2) = OMETH(1,4)
74. OMMA(2) = OMETH(1,4)
75. DO 85 J = 1,NPTS
76. OMMIN(2) = AMIN(OMMIN(2),OMETH(J,4))
77. OMMA(2) = AMAX(OMMA(2),OMETH(J,4))
78. 85 IF((OMMA(2) - OMMIN(2)) .LT. 180 .OR. 1180 .EQ. 1)
79.   GO TO 110
80. DO 90 J = 1,NPTS
81. IF(OMETH(J,4) .LT. 180) OMETH(J,4) = OMETH(J,4) + 360.
82. 1180 = 1
83. GO TO 80
84. 110 1180 = 0
85. DO 500 K = 1,2
86. J = 0
87.C
88.C
89.C
90. THMIN = OMETH(1,K)
91. THMAX = THMIN
92. DO 120 J = 1,NPTS
93. THMIN = AMIN(THMIN,OMETH(J,K))
94. THMAX = AMAX(THMAX,OMETH(J,K))
95. THAVG = (THMAX + THMIN)/2.
96.C
97.C
98.C
99. CALL LONRAN(THAVG,FOV,DOML,DOMR) 9 CHECK ARGUMENTS
100. FOVOM = DOMR - DOML
101.C
102.C
103.C
104. IF(NPTS .NE. 2) GO TO 125
105. IF(OMMIN(K) .GE. (OMEGA + DOML) .AND.
106.   OMMA(K) .LE. (OMEGA + DOMR)) GO TO 122
107. IF((OMMIN(K) + 360.) .GE. (OMEGA + DOML) .AND.
108.   (OMMA(K) + 360.) .LE. (OMEGA + DOMR)) GO TO 122
109. GO TO 500
110. 122 TSTRT(K) = THMIN
111. TEND(K) = THMAX
112. IOVER = IOVER + K
113. GO TO 500

```

```

114.C
115.C
116.C
117. 125 IF(OMEGA .GE. OMMIN(K) .AND. OMEGA .LE. OMMAX(K)) GO TO 130
118. IF((OMEGA + 360.) .GE. OMMIN(K) .AND. (OMEGA + 360.) .LE.
119. . OMMAX(K)) GO TO 140
120. GO TO 310
121.C
122.C
123.C
124. 04 = OMEGA
125. GO TO 145
126. 04 = OMEGA + 360
127. 145 J = 1
128. N1 = NPTS - 1
129. DO 160 I = 1,N1
130. IF(OMETH(I,K+2) .GT. OMETH(I+1,K+2)) GO TO 150
131. IF(OM .LE. OMETH(I,K+2) .OR. OM .GT. OMETH(I+1,K+2)) GO TO 160
132. GO TO 155
133. 150 IF(OM .LE. OMETH(I+1,K+2) .OR. OM .GT. OMETH(I+1,K+2)) GO TO 160
134. 155 CALL SLOPF(OMETH(I,K+2),OMETH(I,K),OMETH(I+1,K+2),
135. . OMETH(I+1,K),OM,TH(J))
136. J = J + 1
137. 160 CONTINUE
138. J = J - 1
139. TSTRT(K) = TH(1)
140. TEND(K) = TH(1)
141. DO 170 I = 1,J
142. TSTRT(K) = AMIN(TH(I),TSTRT(K))
143. TEND(K) = AMAX(TH(I),TEND(K))
144.C
145.C
146.C
147.
148. IOVER = IOVER + K
149. IF((TOTAL .NE. 1) GO TO 500
150. IF(OMMIN(K) .LT. (OM + DOML)) OMMIN(K) = OM + DOML
151. IF(OMMAX(K) .GT. (OM + DOMR)) OMMAX(K) = OM + DOMR
152. GO TO 500
153.C
154.C
155.C
156.C
157.C
158.
159. 300 IF((OMMAX(K) - OMMIN(K)) .GE. FOVOM) GO TO 500
160. WDIR = (OMMAX(K) - (OMMAX(K) - OMMIN(K))/4.)
161. QML = OMMIN(K) + ((OMMAX(K) - OMMIN(K))/4.)
162. OM = OMEGA
163. IF(OM .LT. 0) OM = OM + 360
164. IF((OM + DOML) .GT. WDIR) GO TO 500 04 OMEGA TOO LARGE
165. IF((OM + DOMR) .LT. QML) GO TO 500 04 OMEGA TOO SMALL
166.C
167.C
168.
169. 310 IOVER = IOVER + K
170. IF(OM .GT. QML) GO TO 320
171. 04 = OM + DOMR
172. SITE TO RIGHT

```

```

171. IF (DUM .GE. OMAX(K)) GO TO 371
172. GO TO 330
173. 320 DUM = OM + DOWL
174. IF (DUM .LE. OMIN(K)) GO TO 371
175. 330 NI = NPIS - 1
176. J = 0
177. DO 350 I = 1,NI
178. IF (OMETH(I+1,K+2) .LT. OMETH(I,K+2)) GO TO 340
179. IF (DUM .LT. OMETH(I,K+2)) .OR.
180. DUM .GT. OMETH(I+1,K+2)) GO TO 350
181. GO TO 345
182. 340 IF (DUM .LT. OMETH(I+1,K+2)) .OR. DUM .GT. OMETH(I+1,K+2)) GO TO 350
183. 345 J = J+1
184. CALL SLOPF(OMETH(I,K+2),OMETH(I,K),OMETH(I+1,K+2),
185. OMETH(I+1,K),DUM,TH(J))
186. CONTINUE
187. IF (J .GE. 2) GO TO 360
188. WRITE(6,1200) NSITE
189. RETURN
190. TSTRT(K) = TH(1)
191. TEND(K) = TH(1)
192. DO 370 I = 1,J
193. TSTRT(K) = AMAX1(TSTRT(K),TH(I))
194. TEND(K) = AMIN1(TEND(K),TH(I))
195. GO TO 372
196. 371 TSTRT(K)=THMIN
197. TEND(K)=THMAX
198. IF (ITOTAL .NE. 1) GO TO 500
199. IF (OM .LE. OMAX(K)) GO TO 340
200. IF (OM + DOWL) .LT. OMIN(K)) GO TO 500
201. OMIN(K) = OM + DOWL
202. GO TO 500
203. 350 IF (OM + DOWL) .GT. OMAX(K)) GO TO 500
204. OMAX(K) = OM + DOWL
205. CONTINUE
206. RETURN
207. 1200 FORMAT(I2,'PASS ENTERED OR LEFT SITE ',I4,' BUT NOT BOTH',
208. '*****ERROR*****')
209. END

```

## E.FOSCAN

06 MAR 72 09:42:32 CYCLE: U

## SUBROUTINE FOSCAN

```

1.
2.C
3.C PURPOSE: SCAN THE ERREP MATRIX TO DETERMINE WHICH F.O.'S CAN
4.C BE SATISFIED ON THE GIVEN REVOLUTION.
5.C
6.
7. DIMENSION IS1(5), TSTRY(2), TEND(2), OMMIN(2), OMMAX(2),
8. THES(2), THEE(2), START(2), END(2)
9. INCLUDE BLK1,LIST
10. INCLUDE BLK2,LIST
11. INCLUDE BLK3,LIST
12. INCLUDE BLK4,LIST
13. INCLUDE BLK5,LIST
14.C
15. IPFLG = 0
16. CALL CIPOTE(IPFLG)
17.C
18.C
19. IF(NBETA1.EQ.0.AND. NBETA2.EQ.0) GO TO 5
20. IF(IREV.LT. NBETA1.AND. IREV.GE. TRAJ(1,1)) GO TO 10
21. IF(IREV.GT. NBETA2.AND. IREV.LE. TRAJ(1,1TOTRJ)) GO TO 20
22. 2 RITE(6,1000) IREV
23. RETURN
24.
25. 5 IF(IREV.GE. TRAJ(1,1).AND. IREV.LE. TRAJ(1,1TOTRJ)) GO TO 10
26. GO TO 2
27. 10 LINE = IREV - TRAJ(1,1) + 1
28. GO TO 30
29. 20 LINE = IREV - TRAJ(1,1) - NBETA2 + NBETA1
30. CALL TIDAT(TRAJ(4,LINE),MO,1DAY,MN)
31.C
32. DO 40 I = 1,6
33. DO 40 J = 1,100
34. SCORE(I,J) = 0
35. NSCORE = 0
36. DO 950 JJ = 1,4
37. ISTART = IREP(JJ,1)
38. ISTOP = IREP(JJ,2)
39. DO 900 J = ISTART, ISTOP
40. ICOVER = 0
41.C
42.C CHECK IF MONTH DURING WHICH THIS REV OCCURS IS
43.C INCLUDED IN THE MONTHS DESIRED FOR THIS F.O.
44. ICAT = IREP(6,J)
45. M1 = ICAT/100
46. M2 = MOD(ICAT,100)
47. IF(M2.LT. M1) GO TO 50
48. IF(M2.LT. M1.OR. M2.GT. M2) GO TO 900
49. GO TO 60
50. IF(M2.LT. M1.AND. M2.GT. M2) GO TO 900
51. OMEGA = TRAJ(2,LINE)
52.C
53.C DETERMINE IF TOTAL COVERAGE REQUIRED
54.C
55. ICOV = IREP(7,J)
56. IF(MOD(ICOV,10).NE.1) GO TO 70

```

```

57.  ITOTAL = 1
58.  GO TO 75
59.  70 ITOTAL = 0
60.C
61.C  DIFFERENTIAL HIGH SENSORS REQUIRED. CHOOSE NARROWEST
62.C  INSTRUMENT FIELD OF VIEW TO CALCULATE IF SITE OVERFLOWN.
63.C
64.  75 DO FOR K = 1,5
65.  80 IS19(K) = 0
66.  ISENS = EREP(5,J)
67.  DO FOR K = 1,5
68.  IF(FOO(TSFNS,10) .EQ. 1) IS19(K) = 1
69.  ISENS = ISENS/10
70.  IF(IS19(3) .NE. 1) GO TO 100
71.  FOV = HAS192
72.  GO TO 150
73.  100 IF(IS19(5) .NE. 1) GO TO 120
74.  FOV = HAS194
75.  GO TO 150
76.  120 IF(IS19(1) .NE. 1) GO TO 130
77.  FOV = HAS190
78.  GO TO 150
79.  130 IF(IS19(4) .NE. 1) GO TO 140
80.  FOV = HAS193
81.  GO TO 150
82.  140 FOV = HAS191
83.C
84.C
85.C
86.  150 ISITE = EREP(3,J)
87.  CALL FLYOVR(OMEGA,ISITE,FOV,ITOTAL,ICOVER,ISTRT,TEND,
88.  * OELMIN,OMMAX)
89.  IF(ICOVER .EQ. 0) GO TO 900
90.  WRITE(15,7114) TSTRT,TFND
91.  7114 FORMAT(I2,'TSTRT = ',2(F8.3,2X),'TEND = ',2(F8.3,2X))
92.C
93.C  CALCULATE SUN ELEVATION DURING OVERFLIGHT
94.C
95.  THNOON = TRAJ(5,LINE)
96.  THNID = THNOON + 180.
97.  IF(THNID .GT. 270.) THNID = THNID - 360.
98.  CALL SUNEL(TRAJ(3,LINE),THNOON,THNID,RSNMAX,NOSUCH)
99.  CALL SUNEL(TRAJ(3,LINE),THNOON,THNID,RSNMIN,NOSUCH)
100.  IF(RSNMIN .GE. EREP(10,J)) GO TO 900 W SUN TOO HIGH THIS REV
101.  IF(RSNMAX .LE. EREP(9,J)) GO TO 900 W SUN TOO LOW THIS REV
102.  LSUN = 1
103.  IF(RSNMAX .GT. EREP(10,J)) GO TO 200
104.  IF(RSNMIN .LT. EREP(9,J)) GO TO 170
105.  THES(1) = -90.
106.  TREF(1) = 270.
107.  GO TO 250
108.  170 RSNMIN = EREP(9,J)
109.  CALL FLTHET(TRAJ(3,LINE),THNOON,RSNMIN,THES(1),THEE(1),NOSUCH)
110.  GO TO 250
111.  200 RSNMAX = EREP(10,J)
112.  IF(RSNMIN .LT. EREP(9,J)) GO TO 220
113.  CALL ELTHET(TRAJ(3,LINE),THNOON,RSNMAX,THEE(1),THES(1),NOSUCH)

```



```

114.      GO TO 250
115. 220 RSNMIN = FREP(9,J)
116.      LSUN = 2
117.      CALL ELTHET(TRAJ(3,LINE),THNOON,RSNMIN,THES(1),THEE(2),NOSUCH)
118.      CALL ELTHET(TRAJ(3,LINE),THNOON,RSNMAX,THEE(1),THEE(2),NOSUCH)
119.C
120.C      SLT K AND KK:
121.C          K = 1 INDICATES ASCENDING LEG
122.C          K = 2 INDICATES DESCENDING LEG
123.C          KK KEEPS TRACK OF THE NUMBER OF VALID LIGHTED
124.C              PASSES ON THIS REV FOR THIS F.O.
125. 250 IF(ICOVER.EQ. 1 .OR. ICOVER.EQ. 3) GO TO 260
126.      K = 2
127.      GO TO 270
128. 260 K = 1
129. 270 KK = 0
130.C
131.C      SET L COUNTER. L KEEPS TRACK OF WHICH LIGHTED PORTION
132.C      OF THE REV IS BEING CONSIDERED IN CASE LSUN = 2.
133.C          L = 1
134. 300 IF(THEE(L) .LT. THEE(L)) GO TO 350
135. 310 IF(TEND(K) .LT. THEE(L) .OR. TSTRT(K) .GT. THEE(L)) GO TO 490
136.      KK = KK + 1
137.      IF(TSTRT(K) .LT. THEE(L)) GO TO 320
138.      START(KK) = TSTRT(K)
139.      GO TO 330
140. 320 START(KK) = THEE(L)
141.      (CASE A)
142.C
143.C
144.C
145. 330 IF(TEND(K) .GT. THEE(L)) GO TO 340
146.      END(KK) = TEND(K)
147.      GO TO 450
148. 340 END(KK) = THEE(L)
149.      GO TO 450
150.C
151.C      PROPERLY LIGHTED PORTION SPANS 270/-90 (CASE B)
152.C
153. 350 IF(TSTRT(K) .LT. THEE(L)) GO TO 400
154.      IF(TEND(K) .LT. THEE(L)) GO TO 490
155.      KK = KK + 1
156.      END(KK) = TEND(K)
157.      IF(TSTRT(K) .LT. THEE(L)) GO TO 360
158.      START(KK) = TSTRT(K)
159.      GO TO 450
160. 360 START(KK) = THEE(L)
161.      GO TO 450
162.C
163.C      (2,3,4)
164.C
165. 400 KK = KK + 1
166.      START(KK) = TSTRT(K)
167.      IF(TEND(K) .GT. THEE(L)) GO TO 410
168.      END(KK) = TEND(K)
169.      GO TO 450
170. 410 END(KK) = THEE(L)

```

CUCHAN - CULL COREY FILE EYEP•EREP

DATE 041472

PAGE

23

```

171. IF(TEND(K) .GT. THES(L)) GO TO 420
172. GO TO 450
173. CALL PASBAD(TSTRT(K),TEND(K),START(KK),END(KK),IGOOD)
174. IF(IGOOD .EQ. 0) GO TO 430
175. KK = KK + 1
176.C
177.C
178.C (B4) DOUBLE PASS
179. 430 START(KK) = THES(L)
180. END(KK) = TEND(K)
181.C
182.C (C2)
183.C
184. 450 CALL PASBAD(TSTRT(K),TEND(K),START(KK),END(KK),IGOOD)
185. IF(IGOOD .NE. 0) GO TO 490
186. KK = KK - 1
187.C
188.C NO LIGHTED PASS
189.C
190. 490 IF(LSUB .NE. 2 .OR. L .NE. 1) GO TO 500
191. L = 2
192. GO TO 310
193. 500 IF(ICOVER .NE. 3 .OR. K .NE. 1) GO TO 510
194. K = 2
195. GO TO 300
196.C
197.C LOGIC DEALING WITH SPECIAL NOTES BELONGS HERE AT 510
198.C
199. 510 CONTINUE
200.C
201.C IF(KK .GT. 1) GO TO 550
202. IF(KK .EQ. 0) GO TO 900
203.
204.C DETERMINE THE LARGEST ARC (END(K) - START(K))
205.C AND PUT THE POINTS IN END(1), START(1)
206.C
207.C 550 IF(MOD(ICOV,10) .EQ. 0) GO TO 560
208. IF(MOD(ICOV,10) .EQ. 1) GO TO 600
209. GO TO 560
210.
211.C (NOTHING DONE YET ON TOTAL COVERAGE)
212.C
213.C
214.C ONE OR MORE SINGLE PASSES REQUIRED
215.C
216.C
217. 555 IOREP = 1
218. 560 NUM = ICOV/1000
219. IF(NSCORE .LT. 100) GO TO 565
220. WRITE(6,45003)
221. RETURN
222. 565 NSCORE = NSCORE + 1
223. SCORE(4,NSCORE) = EREP(15,J)/NUM
224. IF(10OREP .EQ. 1) GO TO 581
225. IF(SCORE(4,NSCORE) + TOTSCR(J) .GT. EREP(15,J)) GO TO 570
226. SCORE(4,NSCORE) = SCORE(4,NSCORE)
227. GO TO 400

```

CUCHAN - CULL COREY FILE EREP\*EREP

```

223. 570 SCORE(5,NSCORE) = AMAX1(0,.(EREP(15,J) - TOTSCR(J)))
229. GO TO 900
230. 580 SCORE(5,NSCORE) = 0
231. INOREP = 0
232. GO TO 900
233.C
234.C
235.C
236. 600 NDAYS = MOD(1COV/10,100)
237. IPCOL = 0
238. 610 IPCOL = IPCOL + 1
239. IF(PASDAT(1,IPCOL) .EQ. J) GO TO 630
240. IF(RCOL(PASDAT(1,IPCOL)) .EQ. 0) GO TO 560
241. IF(IPCOL .EQ. 200) GO TO 620
242. GO TO 610
243. 620 WRITE(6,1500) J
244. RETURN
245. 630 IDAY = TRAJ(4,LINE)
246. ILAST = PASDAT(2,IPCOL)
247. DO 640 N = 1,ILAST
248. IPDAY = PASDAT(N+2,IPCOL)
249. IF(IPDAY .EQ. IDAY) GO TO 555
250. DO 650 N = 1,ILAST
251. DUM = ABS(TRAJ(4,LINE) - PASDAT(N+2,IPCOL))/NDAYS
252. NDUM = DUM
253. DUM = DUM - NDUM
254. 650 IF((DUM .GE. 0 .AND. DUM .LE. .1) .OR.
255. (DUM .GE. .9 .AND. DUM .LE. 1.)) GO TO 560
256. GO TO 555
257. 800 SCORE(1,NSCORE) = J
258. SCORE(2,NSCORE) = START(1)
259. SCORE(3,NSCORE) = END(1)
260. SCORE(6,NSCORE) = MCAT(JJ)
261. CONTINUE
262. 950 CONTINUE
263. RETURN
264. 1000 FORMAT(/T2,'SPECIFIED REV EITHER DOES NOT OCCUR IN CURRENT ',
265. 'MISSION OR OCCURS WHEN ABS BETA GREATER THAN 50 DEGREES')
266. 1500 FORMAT(/T2,'THE F.O. IN ENTRY ',14,' OF THE EREP ARRAY ',
267. ', IS NOT LISTED IN PASDAT.')
268. 4500 FORMAT(/T2,'ENTRIES IN SCORE EXCEED DIMENSION OF ARRAY',
269. '*****ERROR*')
270. END

```

DATE 041472

01 FEB 72 11:00:39 CYCLE: 0

COCHMAN - CULL COREY FILE EREP=EREP

E.LATLON

```

1. SUBROUTINE LATLON(XLATC,XLON,OMEGA1,OMEGA2)
2. DATA SIN50/.75004443/RTOD/57.2957795/PI/3.14159265/
3. COS50/.642787610/OMEGA/-0.065871629/
4. RATIO=SIN(XLATC/RTOD)/SIN50
5. IF(RATIO.GT.1.)RATIO=1.
6. IF(RATIO.LT.-1.)RATIO=-1.
7. THETA1=RTOD*ASIN(RATIO)
8. OMEGA1=XLON-RTOD*ATAN(COS50*TAN(THETA1/RTOD))
9. OMEGA2=THETA1
10. THETA2=180.-THETA1
11. OMEGA2=XLON-RTOD*ATAN(COS50*TAN(THETA2/RTOD))
12. OMEGA2=THETA2+180.
13. 2 CONTINUE
14. RETURN
15. END

```

CUCMRAN - CULL COREY FILE EREP•EREP

E•LLIST

- 1•WRUN DPNLST,DPN,EREP
  - 2•HHDG L LISTING OF MAIN ROUTINE
  - 3•MSYM PRINTS,,TPRS
  - 4•KASG,A EREP
  - 5•KFOR,SL EREP,MAIN,MAIN
- UN SAKOLOSKY

DATE 041472

PAGE

23

28 JAN 72 15:02:42 CYCLE: 0

01 FEB 72 12:29:09 CYCLE: 0

E.LUNKAN

```

1. SU-ROUTINE LUNKAN(THETA,FOV,DUMN,DOMP)
2. DATA SINI/.76604443/COSI/.64278761/RATIO/1.0682359/
3. RTOD/57.2957795/E/.065871629/
4. FOVR=FOV/RTOD
5. KHO=ASIN(RATIO*SIN(FOVR))-FOVR
6. COSKHO=COS(RHO)
7. SINRHO=SIN(RHO)
8. A=COSRHO*COSI
9. H=SINRHO*SINI
10. C=COSRHO*SINI
11. D=SINRHO*COSI
12. IF(THETA.GT.270..AND.THETA.LE.360.)THETA=THETA-360.
13. THETA=THETA/RTOD
14. STHETA=SIN(THETA)
15. F=COSRHO*COS(THETA)
16. G=F*THETA
17. PHI=ASIN(STHETA*SINI)*RTOD
18. XLAM=(ATAN2(STHETA*COSI,COS(THETA))-G)*RTOD
19. CALL LATLON(PHI,XLAM,OMGOA,OMGUD)
20. PHIK=ASIN(C*STHETA-U)*RTOD
21. XLAMR=(ATAN2((A*STHETA+B),F)-G)*RTOD
22. CALL LATLON(PHIR,XLAMR,OMGRA,OMGRD)
23. PHIL=ASIN(C*STHETA+U)*RTOD
24. XLAML=(ATAN2((A*STHETA-B),F)-G)*RTOD
25. CALL LATLON(PHIL,XLAML,OMGLA,OMGLD)
26. DLA=OMGLA-OMGOA
27. DRA=OMGRA-OMGOA
28. DLD=OMGLD-OMGUD
29. DRD=OMGRD-OMGUD
30. IF(THETA.GT.-90..AND.THETA.LE.90.) GO TO 1
31. IF(THETA.GT.90..AND.THETA.LE.270.) GO TO 2
32. 1 DUMN=DLA
33. DOMP=DRA
34. RETURN
35. 2 IF(DLD.LT.0.)DLD=DLD+360.
36. DUMN=DRD
37. DOMP=DLD
38. RETURN
39. END

```

06 MAR 72 01:48:39 CYCLE: 0

E-MAIN

```

1.C
2.C
3.C
4.
5.
6.
7.
8.
9.
10.
11.
12.
13.
14.
15.C
16.C
17.C
18.
19.
20.
21.
22.C
23.C
24.C
25.
26.
27.
28.
29.
30.
31.
32.
33.
34.
35.
36.
37.
38.C
39.C
40.C
41.
42.
43.
44.
45.
46.
47.
48.
49.
50.
51.
52.
53.C
54.C
55.C
56.

      INTEGER OPTION, SUBOPT
      INCLUDE BLK1,LIST
      INCLUDE BLK2,LIST
      INCLUDE BLK3,LIST
      INCLUDE BLK4,LIST
      INCLUDE BLK5,LIST
      COMMON/PRINT/PRINT
      DATA IBLANK/'.'/
      NAMELIST/MAX/ARCPAS,MCATEG,THMIN,THMAX
      NAMELIST/CONTRL/APASDE,MCATDE,THMIDE,THMADE,IDEVIC,MISION,
      IERSAV,IEROLD,PERNT,ARCMIN
      .
      INITIALIZE DATA, SET UP COMMON, READ FILES

      IPFLG = 0
      CALL CINOTE(IPFLG)
      ICHFLG = 0
      CALL COMMON(ICHFLG)

      READ NEXT OPTION

      1  WRITE(6,1000)
      LETTER = IBLANK
      READ(5,2000) OPTION
      IF(OPTION.EQ.'STOP'.OR.OPTION.EQ.'HALT') GO TO 900
      FLD(0,6,LETTER) = FLD(0,6,OPTION)
      IF(LETTER.EQ.'E') GO TO 100
      IF(LETTER.EQ.'P') GO TO 200
      IF(LETTER.EQ.'M') GO TO 300
      IF(LETTER.EQ.'A') GO TO 400
      IF(LETTER.EQ.'C') GO TO 500
      IF(LETTER.EQ.'S') GO TO 600
      WRITE(6,2050) OPTION
      GO TO 1

      EVALUATE REVOLUTION

      100  WRITE(6,2075)
      READ(5,2090) IREV
      IF(IREV.GE. TRAJ(1,1) .AND. IREV.LE. TRAJ(1,ITOTRJ)) GO TO 110
      WRITE(6,3000) IREV
      GO TO 1

      110  DO 120 J = 1, ITOTRJ
      IF(TRAJ(1,J).GT. IREV) GO TO 125
      120  IF(IREV.EQ. TRAJ(1,J)) GOTO 130
      125  WRITE(6,3010) IREV
      GO TO 1
      130  CALL FUSCAM
      GO TO 1

      PRINT OPTION

      200  WRITE(6,3020)

```



```

57. READ(5,2000) SUBOPT
58. IF(SUBOPT.NE.'REVSUM') GO TO 205
59. WRITE(6,3050) THSTRT,THEND
60. READ(5,2000) IANS
61. IF(IANS.NE.'NO') GO TO 202
62. WRITE(6,3055)
63. READ(5,3070) THSTRT
64. WRITE(6,3075)
65. READ(5,3070) THEND
66. 202 CALL PSCRMC
67. GO TO 1
68.C
69. 205 IF(SUBOPT.NE.'PRPSUM') GO TO 210
70. WRITE(6,3040)
71. READ(5,2090) IPROP
72. IHDG = 1
73. IFOPKN = 1
74. CALL PPROP(IHDG,IFOPKN,IPROP)
75. GO TO 1
76. 210 IF(SUBOPT.NE.'ERESUM') GO TO 220
77. IHDG = 1
78. IFOPKN = 0
79. 00 215 J=1,NHUPRO
80. IPROP = PRELIST(J)
81. CALL PPROP(IHDG,IFOPKN,IPROP)
82. IPF = IPFLG
83. IF(IPF.EQ.1) GO TO 216
84. 215 IHDG = 0
85. GO TO 1
86. 216 IPFLG = 0
87. READ(5,2000) IANS
88. GO TO 1
89. 220 IF(SUBOPT.NE.'PASSUM') GO TO 230
90. CALL PPASS
91. GO TO 1
92. 230 IF(SUBOPT.NE.'NOHORK') GO TO 240
93. IWORK = 0
94. CALL NORDON
95. GO TO 1
96. 240 IF(SUBOPT.NE.'ALDONE') GO TO 250
97. IWORK = 1
98. CALL ALDONN
99. GO TO 1
100. 250 IF(SUBOPT.NE.'MISSUM') GO TO 260
101. CALL PTSCRN
102. GO TO 1
103. 260 IF(SUBOPT.NE.'PRGSUM') GO TO 265
104. CALL PTSCRIP
105. GO TO 1
106. 265 IF(SUBOPT.NE.'PSCORE') GO TO 270
107. WRITE(6,4000) (SCORE(J,K),J = 1,6),K = 1,NSCORE)
108. GO TO 1
109. 270 WRITE(6,3050) SUBOPT
110. GO TO 1
111.C
112.C
113.C

```

```

114. 300 ARCPAS = APASDE
115. MCATEG = PCATDE
116. THMIN = THNIDE
117. THMAX = THMADE
118. IPRINT = 0
119. WRITE(6,3005) ARCPAS,MCATEG,THMIN,THMAX
120. READ(5,2000) IANS
121. IF(IANS.EQ.'NO') GO TO 310
122. WRITE(6,3100)
123. READ(5,MAX)
124. 310 CALL THEMATX(ARCPAS,MCATEG,THMAX,THMIN,WGTMXC,WGTMXU)
125. GO TO 1
126.C
127.C ACCEPT OPTION
128.C
129. 400 WRITE(6,3120) IREV, PERWT
130. READ(5,2000) IANS
131. IF(IANS.EQ.'NO') GO TO 1
132. WRITE(6,3060) THSTRT,THEND
133. READ(5,2000) IANS
134. IF(IANS.EQ.'NO') GO TO 410
135. WRITE(6,3065)
136. READ(5,3070) THSTRT
137. WRITE(6,3075)
138. READ(5,3070) THEND
139. 410 CALL %STADD
140. GO TO 1
141.C
142.C CHANGE OPTION
143.C
144. 500 WRITE(6,3140)
145. READ(5,2000) IANS
146. IF(IANS.EQ.'NO') GO TO 510
147. WRITE(6,3130) APASDE,MCATDE,MISION,THNIDE,THMADE,PERWT,
    * IENSAV,IEROLD,ILEVIC,ARCMIN
148. 510 MISSAV = MISION
149. WRITE(6,3150)
150. READ(5,CTRL)
151. IF(MISION.EQ. MISSAV) GO TO 1
152. ICHFLG = 1
153. CALL COMMON (ICHFLG)
154. GO TO 1
155.
156.C
157.C SCAN OPTION
158.C
159. 600 WRITE(6,3160)
160. READ(5,2090) IFDAY
161. WRITE(6,3170)
162. READ(5,2090) ILDAY
163. CALL MASCAN(IFDAY,ILDAY)
164. GO TO 1
165.C
166.C HALT OPTION
167.C
168. 900 IF(ITERSAV.GT. 0) GO TO 910
169. WRITE(6,3180)
170. READ(5,2000) IANS

```

CUCURAN - CULL COKEY FILE EREP\*EREP

```

171. IF(IANS.NE.'NO') CALL EXIT
172. WRITE(6,3190)
173. READ(5,2090) IERSAV
174. 911 REWIND IERSAV
175. WRITE(IERSAV) TOTSCR,PASDAT,UMECOV,PASS, IPASS
176. WRITE(6,3200) IERSAV
177. CALL EXIT
178.C
179.C FORMATS
180.C
181. 1000 FORMAT(/T2,'OPTION = ')
182. 2000 FORMAT(A6)
183. 2050 FORMAT(T2,'OPTION ',A6,' IS NOT VALID.')
```

(14 FORMAT)'

```

184. 2075 FORMAT(/T2,'REV NO.?'
185. 2090 FORMAT(14)
186. 3000 FORMAT(T2,'REV ',14,' IS NOT IN THIS MISSION.')
```

(14 FORMAT)'

```

187. 3010 FORMAT(T2,'REV ',14,' NOT EVALUATED ',
188. 'BECAUSE ABSOLUTE BETA GREATER THAN 50')
```

(14 FORMAT)'

```

189. 3020 FORMAT(T2,'PRINT OPTION = ')
190. 3040 FORMAT(/T2,'TYPE PROPOSAL NUMBER
191. 3050 FORMAT(T2,'OPTION ',A6,' NOT AVAILABLE')
```

(14 FORMAT)'

```

192. 3060 FORMAT(/T2,'THSTRT = ',F9.4,5X,'THEND = ',F9.4,5X,'O.K.?'')
193. 3065 FORMAT(/T2,'THSTRT = ')
194. 3070 FORMAT(F12.6)
195. 3075 FORMAT(/T2,'THEND = ')
196. 3085 FORMAT(/T2,'ARCPAS = ',F9.4,5X,'MCATEG = ',13/
197. T2,'THMIN = ',F9.4,5X,'THMAX = ',F9.4,5X,'O.K.?'')
198. 3100 FORMAT(/T2,'SUPPLY VALUES, NAMELIST MAX')
```

(14 FORMAT)'

```

199. 3120 FORMAT(/T2,'ACCEPT REV = ',14,' PERWT = ',F7.3,' OK?')
```

(14 FORMAT)'

```

200. 3130 FORMAT(/T2,'CURRENT VALUES OF CONTROL VARIABLES ARE:')
```

(14 FORMAT)'

```

201. T2,'APASDE = ',F9.4,122,'MCATDE = ',13,
202. T42,'MISSION = ',13/
203. T2,'THMIDE = ',F9.4,122,'THMADE = ',F9.4,
204. T42,'PERWT = ',F7.3/
205. T2,'IERSAV = ',13,T22,'IEROLD = ',13,T42,'IDEVIC = ',A6/
206. T2,'ARCMIN = ',F6.3)
207. 3140 FORMAT(/T2,'PRINT CONTROL DATA ?')
```

(14 FORMAT)'

```

208. 3150 FORMAT(/T2,'ENTER CHANGED PARAMETERS, NAMELIST CONTRL')
209. 3160 FORMAT(/T2,'FIRST DAY OF SCAN = ?
210. 3170 FORMAT(/T2,'LAST DAY OF SCAN = ?
211. 3180 FORMAT(/T2,'PROPOSAL ACCOMPLISHMENTS WILL NOT BE SAVED, O.K.?'')
212. 3190 FORMAT(/T2,'SAVE ON UNIT NUMBER ? (TYPE ANSWER IN 14 FORMAT)')
```

(14 FORMAT)'

```

213. 3200 FORMAT(/T2,'PROPOSAL ACCOMPLISHMENTS HAVE BEEN WRITTEN ',
214. 'ON UNIT ',14)
215. 4000 FORMAT(/T2,'EREP PTR',T13,'THETA S',T23,'THETA E',T33,
216. 'GT UNC',T43,'WGT CON',T53,'CATEG',
217. '100(T2,6(F8,2,2X)/))
218. END
```

06 MAR 72 15:17:38 CYCLE: 0

```

E.MASCAN
1.      SUBROUTINE MASCAN(IFDAY,ILDAY)
2.C
3.C      PURPOSE:  EXAMINE EVERY ORBIT FROM IFDAY THROUGH ILDAY
4.C                AND DETERMINE WHICH N ACCOMPLISH THE MOST IN TERMS
5.C                OF F.O. WEIGHTS.
6.C
7.C
8.
9.
10.
11.C      INCLUDE BLK1,LIST
12.C      INCLUDE BLK2,LIST
13.C      INCLUDE BLK5,LIST
14.
15.
16.
17.C      NAMELIST/INPUT/APASDE,MCATDE,THMIDE,THMADE
18.C      DIMENSION FLY(5,700)
19.C      COMMON/SCAN/ISCAN
20.C      COMMON/PRNT/IPRINT
21.
22.      IPRINT = 1
23.      MAX = 0
24.      IFLY = 0
25.      NOMEGA = 0
26.      BETA = 0
27.      WRITE(6,40000)
28.      READ(5,5150) IANS
29.      IF(IANS.NE. 'NO') GO TO 20
30.      NOMEGA = 1
31.      GO TO 90
32.      WRITE(6,4100)
33.      READ(5,4200) OME
34.      WRITE(6,4250)
35.      IF(OMW.LT. 0) OME = OME + 360
36.      IF(OME.LT. 0) OME = OME + 360
37.      IF(OME.LT. OME) OME = OME + 360
38.      WRITE(6,5000)
39.      READ(5,5050) N
40.      WRITE(6,5100) APASDE,MCATDE,THMIDE,THMADE
41.      READ(5,5150) IANS
42.      IF(IANS.NE. 'NO') GO TO 100
43.      WRITE(6,5200)
44.      READ(5,INPUT)
45.      START = IFDAY
46.      AFND = 1 + ILDAY
47.      DO 500 I = 1,ITOTRJ
48.      LINE = I
49.      IF(TRAJ(4,LINE).LT. START) GO TO 500
50.      IF(TRAJ(4,LINE).GT. XEND) GO TO 510
51.      IF( I.EE. NDETAI .AND. I.LE. NDETA2) GO TO 120
52.      GO TO 130
53.      IF(INEI.EQ. 1) GO TO 500
54.      I ETA = 1
55.      WRITE(6,6000)
56.      GO TO 510
57.      IF(NOMEGA.EQ. 1) GO TO 400
58.      IF(TRAJ(2,LINE).LT. 0) TRAJ(2,LINE) = TRAJ(2,LINE) + 360

```

COCHEMAN - CULL COREY FILE EREP\*ERE\*

```

57. IF (TRAJ(2,LINE) .GE. OME .AND. TRAJ(2,LINE) .LE. OME)
58.   GO TO 400
59. IF (OME .GE. 360 .AND. TRAJ(2,LINE) + 360 .GE. OME .AND.
60.   TRAJ(2,LINE) + 360 .LE. OME) GO TO 400
61.   GO TO 500
62. 400 IPEV = TRAJ(1,LINE)
63.   ISCAN = 1
64.   CALL FOSCAN
65.   CALL THENAX(APASDE,MCATDE,THMADE,THMIDE,WGTMXC,WGTMXU)
66.   IFLY = IFLY + 1
67.   FLY(1,IFLY) = TRAJ(1,LINE)
68.   FLY(2,IFLY) = THSTRT
69.   FLY(3,IFLY) = THEND
70.   FLY(4,IFLY) = WGTMXC
71.   FLY(5,IFLY) = WGTMXU
72.   500 CONTINUE
73.C
74.C
75. 510 IFIRST = IFDAY
76.   ILAST = ILDAY
77.   WRITE(6,5250) N, IFIRST, ILAST
78.   DO 700 I = 1,N
79.     WMAX = 0
80.     DO 600 J = 1,IFLY
81.       IF (WMAX .LT. FLY(4,J)) IMAX = J
82.     WMAX = ANAX(WMAX,FLY(4,J))
83.     WREV = FLY(1,IMAX)
84.     WRITE(6,5300) WREV, (FLY(K,IMAX),K= 2,5)
85.     FLY(4,IMAX) = 0
86.     700 CONTINUE
87.     IPRINT = 0
88.     RETURN
89.C
90.C FORMATS
91.C
92. 4000 FORMAT(/T2,'OMEGA LIMITS DESIRED?')
93. 4100 FORMAT(T2,'TYPE IN WESTERN OMEGA LIMIT.')
94. 4200 FORMAT(F8.3)
95. 4250 FORMAT(T2,'TYPE IN EASTERN OMEGA LIMIT.')
96. 5000 FORMAT(/T2,'INPUT NUMBER OF ORBITS TO BE RETURNED ',
97.   *12 FORMAT('))
98. 5050 FORMAT(T2)
99. 5100 FORMAT(T2,'CURRENT PARAMETER VALUES ARE:  APASDE = ',F8.3/
100.   * T2,'MCATDE = ',F8.3,'THMIDE = ',F8.3,'THMIDE = ',F8.3,'
101.   * THMADE = ',F8.3,' WMAX = ',F8.3,' WREV = ',F8.3,'
102. 5150 FORMAT(A6)
103. 5200 FORMAT(T2,'INPUT CHANGED PARAMETERS, NAMELIST INPUT')
104. 5250 FORMAT(/T2,'THE ',I2,' MAXIMUM EIGHT REVS FROM DAYS ',
105.   * I4,' THROUGH ',I4/
106.   * T3,'REV',T10,'THSTRT',T24,'THEND',T36,
107.   * 'GT CORE',T4,'GT UNCOM')
108. 5300 FORMAT(T2,I4,T10,F8.3,T22,F8.3,T34,F9.3,T47,F9.3)
109. 6000 FORMAT(/T2,'DAYS OF INTEREST INCLUDE PERIOD WHEN ABSOLUTE ',
110.   * 'ETA GREATER THAN 50')
111.   E

```

14 MAR 72 13:07:17 CYCLE: 0

## E.MISSION

```

1. DIMENSION TRAJ(5,700)
2. SL2STR = 120,708
3. SL2END = 148,273
4. SL3STR = 208,232
5. SL3END = 264,217
6. SL4STR = 304,604
7. SL4END = 361,563
8. IFILL = 2
9. IMIS = 1
10. START = SL2STR
11. FINISH = SL2END
12. WRITE(6,1050) IFILE
13.C
14. 50 READ(9, END = 200) NREV,DUA,XLON,BETSUN,D1JA73,THE200,THE200
15. IF(D1JA73.LT. START) GO TO 50
16. IF(D1JA73.GT. FINISH) GO TO 200
17. IFABS(BETSUN).LT. 50) GO TO 60
18. IF(NBETA1.NE. 0) GO TO 50
19. WRITE(6,3000)
20. NBETA1 = NREV
21. GO TO 50
22. 60 IF(NBETA2.NE. 0 .OR. NBETA1.NE. 0) GO TO 70
23. NBETA2 = NREV - 1
24. THNOON = (THE200+THE200)/2
25. ICOUNT = ICOUNT + 1
26. IF(ICOUNT.LE. 111) GO TO 100
27. ICOUNT = 1
28. WRITE(6,1050) IFILE
29. 100 WRITE(6,2000) NREV,XLON,BETSUN,D1JA73,THNOON
30. I = 1 + 1
31. TRAJ(I,1) = NREV
32. TRAJ(2,1) = XLON
33. TRAJ(3,1) = BETSUN
34. TRAJ(4,1) = D1JA73
35. TRAJ(5,1) = THNOON
36. GO TO 50
37. 200 ICOUNT = 0
38. ITOTRJ = 1
39. WRITE(IFILE) ITOTRJ,NBETA1,NBETA2,TRAJ
40. DO 220 K = 1,ITOTRJ
41. DO 220 J = 1,5
42. 220 TRAJ(J,K) = 0
43. I = 0
44. NBETA1 = 0
45. NBETA2 = 0
46. GO TO (250,300,350), IMIS
47. 250 IFILL = 3
48. IMIS = 2
49. START = SL3STR
50. FINISH = SL3END
51. WRITE(6,1050) IFILE
52. GO TO 51
53. 300 IFILL = 4
54. IMIS = 3
55. START = SL4STR
56. FINISH = SL4END

```

```
57.      WRITE(6,1050) IFILE
58.      GO TO 50
59.      350  CONTINUE
60.C
61.      1050  FORMAT(1H1/T2,'SL',11,T8,'WREV',T18,'XLUN',T29,'BETSUN',T41,
62.      '01JA73',T54,'THNOON')
63.      2000  FORMAT(18,14,T15,F9.4,T28,F8.4,T40,F9.4,T53,F8.4)
64.      3000  FORMAT(T2,'ABSOLUTE BETA GREATER THAN 50')
65.      END
```

20 MAR 72 09:46:40 CYCLE: 0

E,NSORT

```

1.C
2.C THIS ROUTINE SORTS THE EREP ARRAY BY MISSION
3.C CATEGORY, REARRANGES THE ARRAY AND OUTPUTS
4.C THE ARRAY IN BINARY ALONG WITH THE POINTERS TO
5.C THE LIMITS OF EACH CATEGORY.
6.C
7. INTEGER SL2(2), SL3(2), SL4(2), SEASON
8. DIMENSION MISSION(7,400), IMIS(7), NONE(50),
9. EREP(15,1000), EREP2(15,1000), IERPNT(7,2), IHIER(7)
10. DATA IHIER/2,23,24,234,3,34,4/
11. DATA SL2/5,5/, SL3/7,9/, SL4/10,12/
12. NAMELIST/INEREP/NEREP
13.C
14.C
15. READ(5,INEREP)
16. J=J+1
17. IF(EREP(I,J) .LT. 0) GO TO 500
18. IYR = 0
19. ISL2 = 0
20. ISL3 = 0
21. ISL4 = 0
22. SEASON=NEREP(6,J)
23. ISTART=SEASON/100
24. IEND=MOD(SEASON,100)
25. IF(IEND .LT. ISTART) IYR=1
26.C
27.C
28. IF(ISTART .GE. SL2(1) .AND. ISTART .LE. SL2(2)
29. .OR. IEND .GE. SL2(1) .AND. IEND .LE. SL2(2)
30. .OR. ISTART .LT. SL2(1) .AND. IEND .GE. SL2(2)) GO TO 20
31. IF(IYR .EQ. 0) GO TO 30
32. IF(ISTART .LE. SL2(2) .AND. IEND .LE. SL2(2)
33. .OR. ISTART .GE. SL2(2) .AND. IEND .GE. SL2(1)) GO TO 20
34. GO TO 30
35. ISL2=1
36. IF(ISTART .GE. SL3(1) .AND. ISTART .LE. SL3(2)
37. .OR. IEND .GE. SL3(1) .AND. IEND .LE. SL3(2)
38. .OR. ISTART .LT. SL3(1) .AND. IEND .GE. SL3(2)) GO TO 40
39. IF(IYR .EQ. 0) GO TO 50
40. IF(ISTART .LE. SL3(2) .AND. IEND .LE. SL3(2)
41. .OR. ISTART .GE. SL3(2) .AND. IEND .GE. SL3(1)) GOTO 40
42. GO TO 50
43. ISL3=1
44. IF(ISTART .GE. SL4(1) .AND. ISTART .LE. SL4(2)
45. .OR. IEND .GE. SL4(1) .AND. IEND .LE. SL4(2)
46. .OR. ISTART .LT. SL4(1) .AND. IEND .GE. SL4(2)) GO TO 60
47. IF(IYR .EQ. 0) GO TO 70
48. IF(ISTART .LE. SL4(2) .AND. IEND .LE. SL4(2)
49. .OR. ISTART .GE. SL4(2) .AND. IEND .GE. SL4(1)) GO TO 60
50. GO TO 70
51. ISL4=1
52.C
53.C DETERMINE FLIGHT CATEGORY FOR FO
54.C
55. IF(ISL2 .EQ. 1 .AND. ISL3 .EQ. 1 .AND. ISL4 .EQ. 1) GO TO 80
56. IF(ISL2 .EQ. 1 .AND. ISL3 .EQ. 1 .AND. ISL4 .EQ. 0) GO TO 90

```



COCHRAN - COLL COREY FILE EREP\*ERLP

```

57. IF(1SL2.EQ. 1 .AND. 1SL3.EQ. 0 .AND. 1SL4.EQ. 1) GO TO 150
58. IF(1SL2.EQ. 1 .AND. 1SL3.EQ. 0 .AND. 1SL4.EQ. 0) GO TO 170
59. IF(1SL3.EQ. 1 .AND. 1SL4.EQ. 1) GO TO 140
60. IF(1SL3.EQ. 1 .AND. 1SL4.EQ. 0) GO TO 190
61. IF(1SL4.EQ. 1) GO TO 210
62. GO TO 230
63. K = 4
64. GO TO 240
65. K = 2
66. GO TO 240
67. K = 6
68. GO TO 240
69. K = 3
70. GO TO 240
71. K = 1
72. GO TO 240
73. K = 5
74. GO TO 240
75. K = 7
76. GO TO 240
77. IN = IN + 1
78. NONE(IN) = J
79. GO TO 10
80. 240 IMIS(K) = IMIS(K) + 1
81. NUM = IMIS(K)
82. MISSION(K,NUM) = J
83. GO TO 10
84. 500 DO 550 N = 1,7
85. NUM = IMIS(N)
86. IF(NUM.NE. 0) GO TO 505
87. WRITE(6,4000) IMIS(N)
88. GO TO 550
89. 505 IF(N.EQ. 1) GO TO 510
90. IERPNT(N,1) = IERPNT(N-1,2) + 1
91. IERPNT(N,2) = IERPNT(N,1) + NUM - 1
92. GO TO 520
93. 510 IERPNT(1,1) = 1
94. IERPNT(1,2) = NUM
95. 520 N1 = IERPNT(N,1)
96. N2 = IERPNT(N,2)
97. DO 530 I = N1,N2
98. MPT = MISSION(N1-I+1)
99. DO 530 J = 1,15
100. EREP2(J,1) = EREP(J,MPT)
101. 530 CONTINUE
102. C
103. C
104. C
105. C
106. C
107. C
108. C
109. C
110. C
111. C
112. C
113. C
114. C
115. C
116. C
117. C
118. C
119. C
120. C
121. C
122. C
123. C
124. C
125. C
126. C
127. C
128. C
129. C
130. C
131. C
132. C
133. C
134. C
135. C
136. C
137. C
138. C
139. C
140. C
141. C
142. C
143. C
144. C
145. C
146. C
147. C
148. C
149. C
150. C
151. C
152. C
153. C
154. C
155. C
156. C
157. C
158. C
159. C
160. C
161. C
162. C
163. C
164. C
165. C
166. C
167. C
168. C
169. C
170. C
171. C
172. C
173. C
174. C
175. C
176. C
177. C
178. C
179. C
180. C
181. C
182. C
183. C
184. C
185. C
186. C
187. C
188. C
189. C
190. C
191. C
192. C
193. C
194. C
195. C
196. C
197. C
198. C
199. C
200. C
201. C
202. C
203. C
204. C
205. C
206. C
207. C
208. C
209. C
210. C
211. C
212. C
213. C
214. C
215. C
216. C
217. C
218. C
219. C
220. C
221. C
222. C
223. C
224. C
225. C
226. C
227. C
228. C
229. C
230. C
231. C
232. C
233. C
234. C
235. C
236. C
237. C
238. C
239. C
240. C
241. C
242. C
243. C
244. C
245. C
246. C
247. C
248. C
249. C
250. C
251. C
252. C
253. C
254. C
255. C
256. C
257. C
258. C
259. C
260. C
261. C
262. C
263. C
264. C
265. C
266. C
267. C
268. C
269. C
270. C
271. C
272. C
273. C
274. C
275. C
276. C
277. C
278. C
279. C
280. C
281. C
282. C
283. C
284. C
285. C
286. C
287. C
288. C
289. C
290. C
291. C
292. C
293. C
294. C
295. C
296. C
297. C
298. C
299. C
300. C
301. C
302. C
303. C
304. C
305. C
306. C
307. C
308. C
309. C
310. C
311. C
312. C
313. C
314. C
315. C
316. C
317. C
318. C
319. C
320. C
321. C
322. C
323. C
324. C
325. C
326. C
327. C
328. C
329. C
330. C
331. C
332. C
333. C
334. C
335. C
336. C
337. C
338. C
339. C
340. C
341. C
342. C
343. C
344. C
345. C
346. C
347. C
348. C
349. C
350. C
351. C
352. C
353. C
354. C
355. C
356. C
357. C
358. C
359. C
360. C
361. C
362. C
363. C
364. C
365. C
366. C
367. C
368. C
369. C
370. C
371. C
372. C
373. C
374. C
375. C
376. C
377. C
378. C
379. C
380. C
381. C
382. C
383. C
384. C
385. C
386. C
387. C
388. C
389. C
390. C
391. C
392. C
393. C
394. C
395. C
396. C
397. C
398. C
399. C
400. C
401. C
402. C
403. C
404. C
405. C
406. C
407. C
408. C
409. C
410. C
411. C
412. C
413. C
414. C
415. C
416. C
417. C
418. C
419. C
420. C
421. C
422. C
423. C
424. C
425. C
426. C
427. C
428. C
429. C
430. C
431. C
432. C
433. C
434. C
435. C
436. C
437. C
438. C
439. C
440. C
441. C
442. C
443. C
444. C
445. C
446. C
447. C
448. C
449. C
450. C
451. C
452. C
453. C
454. C
455. C
456. C
457. C
458. C
459. C
460. C
461. C
462. C
463. C
464. C
465. C
466. C
467. C
468. C
469. C
470. C
471. C
472. C
473. C
474. C
475. C
476. C
477. C
478. C
479. C
480. C
481. C
482. C
483. C
484. C
485. C
486. C
487. C
488. C
489. C
490. C
491. C
492. C
493. C
494. C
495. C
496. C
497. C
498. C
499. C
500. C
501. C
502. C
503. C
504. C
505. C
506. C
507. C
508. C
509. C
510. C
511. C
512. C
513. C
514. C
515. C
516. C
517. C
518. C
519. C
520. C
521. C
522. C
523. C
524. C
525. C
526. C
527. C
528. C
529. C
530. C
531. C
532. C
533. C
534. C
535. C
536. C
537. C
538. C
539. C
540. C
541. C
542. C
543. C
544. C
545. C
546. C
547. C
548. C
549. C
550. C
551. C
552. C
553. C
554. C
555. C
556. C
557. C
558. C
559. C
560. C
561. C
562. C
563. C
564. C
565. C
566. C
567. C
568. C
569. C
570. C
571. C
572. C
573. C
574. C
575. C
576. C
577. C
578. C
579. C
580. C
581. C
582. C
583. C
584. C
585. C
586. C
587. C
588. C
589. C
590. C
591. C
592. C
593. C
594. C
595. C
596. C
597. C
598. C
599. C
600. C
601. C
602. C
603. C
604. C
605. C
606. C
607. C
608. C
609. C
610. C
611. C
612. C
613. C
614. C
615. C
616. C
617. C
618. C
619. C
620. C
621. C
622. C
623. C
624. C
625. C
626. C
627. C
628. C
629. C
630. C
631. C
632. C
633. C
634. C
635. C
636. C
637. C
638. C
639. C
640. C
641. C
642. C
643. C
644. C
645. C
646. C
647. C
648. C
649. C
650. C
651. C
652. C
653. C
654. C
655. C
656. C
657. C
658. C
659. C
660. C
661. C
662. C
663. C
664. C
665. C
666. C
667. C
668. C
669. C
670. C
671. C
672. C
673. C
674. C
675. C
676. C
677. C
678. C
679. C
680. C
681. C
682. C
683. C
684. C
685. C
686. C
687. C
688. C
689. C
690. C
691. C
692. C
693. C
694. C
695. C
696. C
697. C
698. C
699. C
700. C
701. C
702. C
703. C
704. C
705. C
706. C
707. C
708. C
709. C
710. C
711. C
712. C
713. C
714. C
715. C
716. C
717. C
718. C
719. C
720. C
721. C
722. C
723. C
724. C
725. C
726. C
727. C
728. C
729. C
730. C
731. C
732. C
733. C
734. C
735. C
736. C
737. C
738. C
739. C
740. C
741. C
742. C
743. C
744. C
745. C
746. C
747. C
748. C
749. C
750. C
751. C
752. C
753. C
754. C
755. C
756. C
757. C
758. C
759. C
760. C
761. C
762. C
763. C
764. C
765. C
766. C
767. C
768. C
769. C
770. C
771. C
772. C
773. C
774. C
775. C
776. C
777. C
778. C
779. C
780. C
781. C
782. C
783. C
784. C
785. C
786. C
787. C
788. C
789. C
790. C
791. C
792. C
793. C
794. C
795. C
796. C
797. C
798. C
799. C
800. C
801. C
802. C
803. C
804. C
805. C
806. C
807. C
808. C
809. C
810. C
811. C
812. C
813. C
814. C
815. C
816. C
817. C
818. C
819. C
820. C
821. C
822. C
823. C
824. C
825. C
826. C
827. C
828. C
829. C
830. C
831. C
832. C
833. C
834. C
835. C
836. C
837. C
838. C
839. C
840. C
841. C
842. C
843. C
844. C
845. C
846. C
847. C
848. C
849. C
850. C
851. C
852. C
853. C
854. C
855. C
856. C
857. C
858. C
859. C
860. C
861. C
862. C
863. C
864. C
865. C
866. C
867. C
868. C
869. C
870. C
871. C
872. C
873. C
874. C
875. C
876. C
877. C
878. C
879. C
880. C
881. C
882. C
883. C
884. C
885. C
886. C
887. C
888. C
889. C
890. C
891. C
892. C
893. C
894. C
895. C
896. C
897. C
898. C
899. C
900. C
901. C
902. C
903. C
904. C
905. C
906. C
907. C
908. C
909. C
910. C
911. C
912. C
913. C
914. C
915. C
916. C
917. C
918. C
919. C
920. C
921. C
922. C
923. C
924. C
925. C
926. C
927. C
928. C
929. C
930. C
931. C
932. C
933. C
934. C
935. C
936. C
937. C
938. C
939. C
940. C
941. C
942. C
943. C
944. C
945. C
946. C
947. C
948. C
949. C
950. C
951. C
952. C
953. C
954. C
955. C
956. C
957. C
958. C
959. C
960. C
961. C
962. C
963. C
964. C
965. C
966. C
967. C
968. C
969. C
970. C
971. C
972. C
973. C
974. C
975. C
976. C
977. C
978. C
979. C
980. C
981. C
982. C
983. C
984. C
985. C
986. C
987. C
988. C
989. C
990. C
991. C
992. C
993. C
994. C
995. C
996. C
997. C
998. C
999. C

```

```

114.      WRITE(10) IERPNT,EREP2
115.C
116.C
117. 1000  FORMAT(IH1,T2,I3,' F.O.S ARE IN CATEGORY ',I3/
      . T2,'THEY COMPRISE EREP ENTRIES ',I4,' TO ',I4/////
119.      . T3,'NUM.',T7,'PROP FO SITE D SENSRS SEASN PASS',
120.      . T4,'NO LIGHT CLOUD SNOW WEIGHT'//
121.      . 500(I2,I4,I1,F5,O,I1,F3,O,I1,F4,O,I1,F2,O,I1,
122.      . F6,O,I1,F5,O,F6,O,I1,F3,O,I1,F4,O,I1,F4,O,I1,
123.      . F3,O,I1,F4,O,I1,F3,O,I1,F4,O,I1,F4,O,I1,F5/I))
124. 2000  FORMAT(IH1//T2,'THE FOLLOWING ARE INCLUDED IN NO CATEGORY: '//
125.      . T10,'PROP',5A,'F.O.')
126. 3000  FORMAT(T10,F5,O,5X,F3,O)
127. 4000  FORMAT(////T2,'.....THERE ARE NO ENTRIES IN MISSION CATEGORY ',
128.      . I3,' .....')
129.      END

```

08 MAR 72 01:51:24 CYCLE: 0

E\*NO\*NDON

```

1.      SUBROUTINE NONDON
2.C
3.C      PURPOSE:  LIST THE F.O.'S WHICH CAN BE DONE ON THE
4.C      CURRENT MISSION FOR WHICH ACCUMULATED
5.C      SCORE = 0 (WORK = 0)
6.C
7.      INCLUDE BLK1,LIST
8.      INCLUDE BLK2,LIST
9.      INCLUDE BLK3,LIST
10.     INCLUDE BLK4,LIST
11.     DIMENSION PRINT(5,2), IPR(5)
12.     EQUIVALENCE (PRINT(1,2),IPR(1))
13.     IPFLG = 0
14.     DO 10 J = 1,5
15.     DO 10 K = 1,2
16.     10 PRINT(J,K) = 0
17.     CALL CINOTE(IPFLG)
18.     WRITE(6,1000)
19.     DO 100 I = 1,4
20.     I1 = IEREP(I,1)
21.     I2 = IEREP(I,2)
22.     DO 100 J = 1,12
23.     IF(TOTSCR(J) .GT. 0) GO TO 100
24.     PRINT(I,1) = PRINT(I,1) + EREP(15,J)
25.     IPR(I) = IPR(I) + 1
26.     IF(IPFLG .NE. 0) GO TO 100
27.     NPRO = EREP(1,J)
28.     NFO = EREP(2,J)
29.     WRITE(6,1050) NPRO,NFO,MCAT(1),EREP(15,J)
30.     100 CONTINUE
31.     EXITL(6,1075)
32.     DO 200 I = 1,4
33.     PRINT(5,1) = PRINT(5,1) + PRINT(I,1)
34.     200 IPR(5) = IPR(5) + IPR(I)
35.     WRITE(6,2000) (MCAT(K), (PRINT(K,L),L = 1,2)),K = 1,4),
36.     * (PRINT(5,L),L = 1,2)
37.     IPFLG = 0
38.     RETURN
39. 1000  FORMAT(/T2,'F.O.'S ON WHICH NO WORK HAS BEEN DONE',
40.     * T2,'PROP',T8,'FO',T13,'MCAT',T20,'POS SCR')
41. 1050  FORMAT(T2,I4,T8,I2,T13,I3,T20,F6.2)
42. 1075  FORMAT(/T2,'SUMMARY',T2,'MISSION CAT',T16,
43.     * 'TOT POS SCR',T30,'NO OF FOS')
44. 2000  FORMAT(4(I3,I3,I18,F7.1,T32,I3/),
45.     * T3,'TOTAL',T18,F7.1,T31,I4)
46.     E*O

```

DATE 041472

08 MAR 72 01:51:27 CYCLE: 0

CUCRRAN - CULL COREY FILE EREP•EREP

E•PASBAD

1. SUBROUTINE PASHAD(TSTART,TEND,THSTRT,THEND,IGOOD)

2.C

3.C A PASS OVER A SITE WILL BE CALLED GOOD IF AT LEAST ONE-HALF THE  
 4.C THETA RANGE OF THE SITE IS COVERED OR ARCMIN DEGREES OF THE  
 5.C ZLV PASS IS SPENT OVER THE SITE.

6.C

7. INCLUDE BLK1,LIST

8.C

9.C

10.C

11. IF(THSTRT .LT. TSTART) GO TO 100

12. IF(THSTRT .GE. TEND) GO TO 180

13.C

PARTIAL COVERAGE AT LEAST

14.C

15. IF(THEND .GE. TEND) THDUM = TEND - THSTRT

16. IF(THEND .LT. TEND) THDUM = THEND - THSTRT

17. IF(THDUM .GE. ARCMIN) GO TO 190

18. IF(THDUM .GE. ((TEND - TSTART)/2)) GO TO 190

19. GO TO 180

20.C

21. 100 IF(THEND .LT. TSTART) GO TO 180

22. IF(THEND .GT. TEND) GO TO 190

23. IF(THEND .GE. (THSTRT + ARCMIN)) GO TO 190

24. IF((THEND - TSTART) .GE. ((TEND - TSTART)/2)) GO TO 190

25.C

26. 180 IG000 = 0

27. RETURN

28. 190 IG000 = 1

29. RETURN

30. END

DATE 041472

08 MAR 72 01:51:31 CYCLE: 0

E,PPASS

1. SUBROUTINE PPASS

2.C

3.C PURPOSE: PRINT A SUMMARY OF THE PASS DATA ACCEPTED SO FAR

4.C

5. INCLUDE BLK3,LIST

6. INCLUDE BLK4,LIST

7. WRITE(A,1000) IPASS

8. DO 100 I = 1,IPASS

9. CALL TINDAT(PASS(3,I),MO,IDAY,HR)

10. HR = HR + PASS(4,I) \* ORBPER/21600.

11. IF (HR .GT. 24) GO TO 20

12. IF (HR .GE. 0) GO TO 30

13. HR = HR + 24.

14. IDAY = IDAY + 1

15. GO TO 30

20. HR = HR - 24

16. IDAY = IDAY + 1

30. IREV = PASS(1,I)

100. WRITE(A,2000) (REV, PASS(2,I), NO, IDAY, HR,

. (PASS(K,I),K = 4,7)

20. RETURN

22. 1010. FORMAT(/T2,I3,T6,PASSES HAVE BEEN ACCEPTED,//

. T19,PASS STARTS,T52,SUN ELEVATION,T2,REV NO,T11, END')

. OMEGA,T19,MO DAY HR,T36,THSTRT THEND START

25. 2000. FORMAT(I3,I4,T11,F6.2,T19,I2,T23,I2,T27,F7.4,T35,

. F6.2,T44,F6.2,T52,F5.2,T60,F5.2)

26. END

27.

08 MAR 72 01:51:35 CYCLE: 0

E.PPGRP

```

1. SUBROUTINE PPROP(IHDG,IFOPRN,IPROP)
2.C
3.C PUNPDSF; PRINT CONDITIONAL, UNCONDITIONAL, AND ACCUMULATED
4.C SCORES FOR A GIVEN PROPOSAL, IPROP. IF IFOPRN = 1,
5.C LIST EACH F.O. SEPARATELY. OTHERWISE LIST ONLY TOTALS.
6.C
7. INCLUDE BLK1,LIST
8. INCLUDE BLK2,LIST
9. INCLUDE BLK3,LIST
10. INCLUDE BLK4,LIST
11. INCLUDE BLK5,LIST
12. DIMENSION PRINT(2,4)
13. IF(IHDG.NE. 1) GO TO 20
14. IHDG = 0
15. WRITE(6,1000) IREV
16. DO 30 J = 1,2
17. DO 30 K = 1,4
18. 30 PRINT(J,K) = 0
19. ISUOU = 0
20. IER = IERPT(7,2)
21. DO 200 I = 1,IER
22. IF(EREP(1,1).NE. IPROP) GO TO 200
23. I-000 = 1
24. DO 150 J = 1,NSCORE
25. IF(SCORE(1,J).NE. 1) GO TO 150
26. PRINT(1,1) = SCORE(5,J)
27. PRINT(1,2) = SCORE(4,J)
28. GO TO 160
29. 150 CONTINUE
30. PRINT(1,1) = 0
31. PRINT(1,2) = 0
32. 160 IF(IFOPRN.NE. 1) GO TO 180
33. IFO = EREP(2,1)
34. WRITE(6,2000) IPROP,IFO,PRINT(1,1),PRINT(1,2),
35. TOTSCR(1), EREP(15,1)
36. 180 PRINT(2,1) = PRINT(2,1) + PRINT(1,1)
37. PRINT(2,2) = PRINT(2,2) + PRINT(1,2)
38. PRINT(2,3) = PRINT(2,3) + TOTSCR(1)
39. PRINT(2,4) = PRINT(2,4) + EREP(15,1)
40. 200 CONTINUE
41. IPRO = IPROP
42. IF(IGOOD.NE. 0) GO TO 250
43. WRITE(6,2050) IPRO
44. RETURN
45. 250 WRITE(6,3000) IPRO, (PRINT(2,K),K = 1,4)
46. RETURN
47. 1000 FORMAT(/T2,'PROPOSAL SUMMARY, REV = ',I4/
48. T2,'PROPN',T10,'FO',T16,'PASS COND',T27,'PASS UNCOND',
49. T41,'ACCOM',T49,'TOT POS')
50. 2000 FORMAT(T2,I4,T10,I2,I7,F5.1,T30,F5.1,T41,F5.1,T50,F5.1)
51. 2050 FORMAT(T2,'PROPOSAL ',I4,' IS NOT LISTED')
52. 3000 FORMAT(T2,I4,T8,'TOTAL',T17,F5.1,T30,F5.1,T41,F5.1,T50,F5.1)
53. END

```

08 MAR 72 01:54:02 CYCLE: 0

E.PSCNMC

SUBROUTINE PSCNMC

PURPOSE: PRINT SUMMARY OF CURRENT SCORE MATRIX BY CATEGORY.

INCLUDE BLK1,LIST

INCLUDE BLK2,LIST

INCLUDE BLK3,LIST

INCLUDE BLK4,LIST

INCLUDE BLK5,LIST

DIMENSION PRINT(5,5), NFO(5)

IF (IDEVIC.EQ. 0) PRINT(1,5)

WRITE(6,2000)

DO 20 J = 1,NSCORE

NPT = SCORE(1,J)

IPROP = EREP(1,NPT)

IFO = EREP(2,NPT)

TWEIGH = EREP(15,NPT)

MATEG = SCORE(6,J)

CUR = TOTSCR(NPT)

20 WRITE(6,2010) IPROP,IFO,(SCORE(K,J),K = 2,3),TWEIGH,

\* (SCORE(K,J), K = 4,5), MATEG, CUR

ZERO OUT PRINT MATRIX

50 DO 65 J = 1,5

DO 60 K = 1,5

PRINT(K,J) = 0

65 NFO(J) = 0

DO 100 I = 1,NSCORE

NPT = SCORE(1,I)

CALL PASBAD(SCORE(2,1),SCORE(3,1),THSTRT,THEND,IGOOD)

IF (IGOOD.EQ. 0) GO TO 100

K = 0

DO 70 J = 1,3

IF (SCORE(6,1).EQ. MATEG(J)) K = J

70 IF (K.EQ. 0) GO TO 80

K = 4

PRINT(K,1) = PRINT(K,1) + SCORE(5,1)

PRINT(K,2) = PRINT(K,2) + SCORE(4,1)

PRINT(K,3) = PRINT(K,3) + TOTSCR(NPT)

PRINT(K,4) = PRINT(K,4) + EREP(15,NPT)

NFO(K) = NFO(K) + 1

DO 300 J = 1,4

DO 200 K = 1,4

PRINT(5,J) = PRINT(5,J) + PRINT(K,J)

NFO(5) = NFO(5) + NFO(J)

300 WRITE(6,3000) IPROP,THSTRT,THEND,((MATEG(K),PRINT(K,J),J = 1,4),

\* NFO(K)),K = 1,4), (PRINT(5,J), J = 1,4), NFO(5)

RETURN

50000 FORMAT(/12,'PROP',T4,F04,T14,'TSTRT',T23,

'TEND',T30,'TWEIGH',T38,'GT U',T46,'WGT C',

T53,'MATEG',T61,'CUR',GT)

2010 FORMAT(T2,T4,T9,T12,T14,F5.1,T22,F5.1,T30,F5.1,

T38,F5.1,T46,F5.1,T54,T60,F5.1)

30000 FORMAT(/12,'SUMMARY OF REV',T13,'TSTRT = ',F5.1,

'TEND = ',F5.1/T2,'MATEG',T10,

57.     . ,GT COMD.,T20.,WGT UNC.,T29.,ACCU.,  
58.     . T36.,TOT WGT POS.,T50.,NO FOS./  
59.     . 4(T3.,I3.,I9.,F7.,I.,T19.,F7.,I.,T27.,F7.,I.,T37.,F7.,I.,  
60.     . T52.,I3./),I2.,TOTAL.,T9.,F7.,I.,T19.,F7.,I.,  
61.     . T27.,F7.,I.,T37.,F7.,I.,T51.,I4)  
62.     END



DATE 041472

08 MAR 72 01:54:07 CYCLE: 0

E.PTSCRM

```

1.      SUBROUTINE PTSCRM
2.C
3.C      PURPOSE:  PRINT OUT NO OF F0'S WORKED ON AND ACCUMULATED
4.C              SCORE FOR CURRENT MISSION--TOTALS AND BY MISSION
5.C              CATEGORY.  IF PRINTER, PRINT EACH F.O.O.
6.C
7.      INCLUDE BLK1,LIST
8.      INCLUDE BLK2,LIST
9.      INCLUDE BLK3,LIST
10.     INCLUDE BLK4,LIST
11.     DIMENSION PRINT(5,2), KNT(5,2)
12.     DO 10 J = 1,5
13.     DO 10 K = 1,2
14.     PRINT(J,K) = 0
15.     KNT(J,K) = 0
16.     DO 50 K = 1,4
17.     ILIM1 = IEREP(K,1)
18.     ILIM2 = IEREP(K,2)
19.     IF(IDEVIC.NE.'PRINTM') GO TO 40
20.     WRITE(6,1000) MCAI(K)
21.     DO 30 I = ILIM1,ILIM2
22.     NPRO = EREP(I,1)
23.     NFO = EREP(2,I)
24.     WRITE(6,1050) NPRO,NFO,TOTSCR(I),EREP(I,5,1)
25.     DO 50 I = ILIM1,ILIM2
26.     PRINT(K,1) = PRINT(K,1) + TOTSCR(I)
27.     PRINT(K,2) = PRINT(K,2) + EREP(I,5,1)
28.     KNT(K,2) = KNT(K,2) + 1
29.     IF(TOTSCR(I).LE.0) GO TO 50
30.     KNT(K,1) = KNT(K,1) + 1
31.     CONTINUE
32.     DO 60 I = 1,4
33.     DO 60 J = 1,2
34.     PRINT(5,J) = PRINT(5,J) + PRINT(I,J)
35.     KNT(5,J) = KNT(5,J) + KNT(I,J)
36.     WRITE(6,2000) MISSION
37.     WRITE(6,2050) (MCAI(I),PRINT(I,J),J = 1,2),
38.     * (KNT(I,J),J = 1,2),I = 1,4),
39.     * (PRINT(5,J),J = 1,2),(KNT(5,J),J = 1,2)
40.     RETURN
41. 1000  FORMAT(/,T2,'SUMMARY OF FOS FOR MISSION CATEGORY ',I3/
42.     * T2,'PROP',T9,'FC NO.',T17,'WGT',T22,'SAT.',
43.     * T28,'TOT FGT POS')
44. 1050  FORMAT(T2,I4,T11,I2,T19,F5.1,T31,F5.1)
45. 2000  FORMAT(/,T2,'SUMMARY OF ACCUMULATED SCORE FOR SL',I1,' MISSION,/
46.     * T2,'MISSION CAT',T16,'WGT SAT',T25,'TOT WGT POS',
47.     * T38,'NON ZERO FOS',T52,'TOT FOS')
48. 2050  FORMAT(4(T4,I3,T15,F7.1,T26,F7.1,T42,I4,T54,I4/),
49.     * T4,'TOTAL',T15,F7.1,T26,F7.1,T42,I4,T54,I4)
50.     END

```

DATE 04/14/72

08 MAR 72 01:54:12 CYCLE: 0

COCHRAN - CULL COREY FILE EREP\*EREP

E.PTSCRP

SUBROUTINE PTSCR

1.  
2.C  
3.C PURPOSE: SUMMARIZE ACCUMULATED FO WEIGHTS ACCOMPLISHED  
4.C

5. INCLUDE BLK1,LIST  
6. INCLUDE BLK2,LIST  
7. INCLUDE BLK3,LIST  
8. INCLUDE BLK4,LIST  
9. DIMENSION KNT(8,3), PRINT(8,2)

10. DO 10 J = 1,8  
11. DO 5 K = 1,2  
12. PRINT(J,K) = 0

13. 5 KNT(J,K) = 0  
14. 10 KNT(J,3) = 0

15. KNT(1,1) = 2  
16. KNT(2,1) = 23

17. KNT(3,1) = 24  
18. KNT(4,1) = 234

19. KNT(5,1) = 3  
20. KNT(6,1) = 34

21. KNT(7,1) = 4  
22. KNT(8,1) = 'TOTAL'

23. DO 100 K = 1,7  
24. I1 = IERPNT(K,1)  
25. I2 = IERPNT(K,2)

26.C  
27. DO 100 I = 1,12

28. PRINT(K,1) = PRINT(K,1) + TOTSCR(I)  
29. PRINT(K,2) = PRINT(K,2) + EREP(15,1)

30. KNT(K,3) = KNT(K,3) + 1  
31. IF(TOTSCR(I) .LE. 0) GO TO 100

32. KNT(K,2) = KNT(K,2) + 1  
33. 100 CONTINUE

34. DO 200 I = 1,7  
35. DO 130 J = 1,2

36. 130 PRINT(I,J) = PRINT(8,J) + PRINT(I,J)  
37. DO 140 J = 2,3

38. 140 KNT(I,J) = KNT(I,J) + KNT(I,J)  
39. 200 CONTINUE

40. WRITE(6,1000)  
41. WRITE(6,2000) ((KNT(I,1),PRINT(I,J),J=1,2),

42. (KNT(I,J),J=2,3)),I=1,8)  
43. RETURN

44.C  
45.C FORMATS

46.C  
47. 1000 FORMAT(//T2,'FO WEIGHT ACCOMPLISHED SO FAR: ',

48. T2,'NCATEG',T11,'ACCUM LGT TOT WT PUS',

49. T35,'NUM ZERU FOS',T50,'TOT FOS',

50. 2000 FORMAT(7(T3,T13,T11,F7.1,T22,F7.1,T39,I4,T51,I4/),

51. T2,A6,I1,F7.1,T22,F7.1,T39,I4,T51,I4)

52. END

DATE 041472

11 FEB 72 08:36:25 CYCLE: 0

CUCHAN - CULL COREY FILE EREP•EREP

E•RECOMP

```

1.  RUN  DPNRE,DPNCOM,EREP,35,100
2.  HDG  RECOMPILE ELEMENTS ON EREP FILE  UN SAKULOSKY
3.  SYN  PRINTS,,TPRS
4.  ASG  AX EREP
5.  FOR,S EREP,ALLDON,,ALLDON,,ALLDON
6.  FOR,S EREP,COMMON,,COMMON,,COMMON
7.  FOR,S EREP,FLYOVR,,FLYOVR,,FLYOVR
8.  FOR,S EREP,FOSCAN,,FOSCAN,,FOSCAN
9.  FOR,S EREP,MAIN,,MAIN,,MAIN
10. FOR,S EREP,MISION,,MISION,,MISION
11. FOR,S EREP,NONDON,,NONDON,,NONDON
12. FOR,S EREP,PASBAD,,PASBAD,,PASBAD
13. FOR,S EREP,PPASS,,PPASS,,PPASS
14. FOR,S EREP,PPROP,,PPROP,,PPROP
15. FOR,S EREP,PSCRMCM,,PSCRMCM,,PSCRMCM
16. FOR,S EREP,PTSCRM,,PTSCRM,,PTSCRM
17. FOR,S EREP,PTSCRCP,,PTSCRCP,,PTSCRCP
18. FOR,S EREP,THEMAX,,THEMAX,,THEMAX
19. FOR,S EREP,NGTADD,,NGTADD,,NGTADD
20.  PACK EREP
21.  PREP EREP
22.  MAP,AL  EREP,ERMAP,,ERMAP
23.  PREP EREP
24.  PRINT EREP

```

22 MAR 72 15:55:21 CYCLE: 0

## E.SITES

```

1.C
2.C
3.C
4.C
5.C
6.C
7.C
8.C
9.
10.
11.
12.
13.
14.
15.
16.C
17.C
18.
19.C
20.C
21.C
22.
23.C
24.C
25.C
26.
27.C
28.
29.
30.
31.
32.
33.
34.
35.
36.C
37.C
38.
39.C
40.
41.
42.
43.
44.
45.
46.
47.
48.
49.
50.C
51.C
52.C
53.
54.
55.
56.

THIS PROGRAM READS A NAMELIST INPUT CONTAINING
EREP, AN ARRAY OF EREP SITE LAT AND LON VALUES WITH SITES
SEPARATED BY 9999.0,9999.0
EREPNO, AN ASSOCIATED ARRAY CONTAINING SITE NUMBERS IN THE ORDER
      IN WHICH THEY OCCUR IN EREP

      DIMENSION THOMEG(4,5500), EREP(2,5500), INDEX(2,600),
      . SITE(2,5000)
      INTEGER EREPNO(600)
      REAL INCL
      EQUIVALENCE (THOMEG(1,1), EREP(1,1))
      NAMELIST/INEREP/EREP,EREPNO
      NAMELIST/OUT2/NENTR,INDEX,THOMEG

      ROTATION RATE OF THE EARTH (DEG/HR)
      OMEGAE = 15.04104953

      ORBITAL ROTATION RATE OF THE SATELLITE (DEG/HR)
      OMEGA0 = 360./1.55471699

      REGRESSION RATE OF THE LINE OF NODES OF THE SATELLITE ORGIT (DEG/HR)
      REGRES = .21174988

      PI = 3.14159
      DTOR = PI/180.0
      RTOD = 180.0/PI
      EARTH = (OMEGAE + REGRES)/OMEGA0
      INCL=50.0
      RINCL=INCL*DTOR
      SINRIN=SIN(RINCL)
      RINCL = COS(RINCL)

      HEAD(5,INEREP)

      MAXNO = EREPNO(1)
      J=J+1
      MAXNO = MAX(MAXNO,EREPNO(J))
      10 K=K+1
      20 IF (ABS(EREP(1,K)).GT. 90) GO TO 30
      L=L+1
      SITE(1,L)=EREP(1,K)
      SITE(2,L)=EREP(2,K)
      NPAIR=NPAIR+1
      GO TO 20

      A COMPLETE SITE DEFINITION HAS BEEN READ

      30 JJ=EREPNO(J)
      I=DEA(1,JJ)=L+1-NPAIR
      L=DEA(2,JJ)=NPAIR
      NPAIR=NPAIR+1

```

```

57. NENTR=INT(NTR+1)
58. I = INDEX(1,1,1)
59. *SITE(4,7777) NENTR, EREPNO(J), ((SITE(H1,N2),N1 =1,2),
60. * K2 = 1,1)
61. 7777 FORMAT(//T2,I4,T10,I4,T20,2(I9,4,2X)/)
62. IF(EREP(1,K) .LT. 0) GO TO 40
63. GO TO 10
64. *SITE(2) NENTR, INDEA,SITE
65. *SITE(6,6666) NENTR, EREP(1,K), EREP(2,K),
66. * SITE(1,1), SITE(2,L)
67. 6666 FORMAT(//T2,NENTR = ,I4,2X,EREP = ,2(I9,4,2X),
68. * /T2,SITE = ,2(I9,4,2X))
69.C
70.C COMPUTE THETA AND OMEGA, ASCENDING AND DESCENDING, FOR
71.C EACH LAT AND LON
72. DO 100 K=1,L
73. *LAT=SITE(1,K)
74. *LON=SITE(2,K)
75. IF(OLON .LT. 0) OLON=360+OLON
76. IF(ABS(LAT) .LT. INCL) GO TO 50
77. IF(OLAT .LT. 0) OLAT=-49.999
78. IF(OLAT .GT. 0) OLAT=49.999
79. 50 THETA=ASIN(SIN(OLAT*DTOR)/SINKIN)
80. THETA=PI-THETA
81. OMEGA=OLON*DTOR-ATAN2(KINCL*SIN(THETA),
82. * COS(THETA))+EARTH*THETAU
83. OMEGA=OLON*DTOR-ATAN2(KINCL*SIN(THETA),
84. * COS(THETA))+EARTH*THETA
85. THETA=THETA*RTOD
86. THETA=THETA*RTOD
87. OMEGA=OMEGA*RTOD
88. IF(OMEGA .GE. 360) OMEGA=AMOD(OMEGA,360.)
89. IF(OMEGA .LT. 0) OMEGA=OMEGA+360
90. OMEGA=OMEGA*RTOD
91. IF(OMEGA .GE. 360) OMEGA=AMOD(OMEGA,360.)
92. IF(OMEGA .LT. 0) OMEGA=OMEGA+360
93. THPLG(1,N)=THETA
94. THMEG(2,K)=OMEGA
95. THMEG(3,K)=THETA
96. THMEG(4,K)=OMEGA
97. *SITE(3,OUT2)
98. DO 200 N=1,MAXNO
99. KK=N
100. ISTART=INDEX(1,K)
101. IEND=ISTART-1+INDEX(2,K)
102. 200 *SITE(4,1000) KK,((SITE(J,M),J=1,2),
103. * (THMEG(J,1),J=1,4)),N=ISTART,IEND)
104.C
105. 1000 FORMAT(//T2,SITE ,I3,T20,PLATITUDE,I33,PLONGITUDE,
106. * T47,THETAU,I50,OMEGA,I74,THETA,I87,OMEGA,//
107. * 2(I21,I7,2,I34,F7,2,I46,F7,2,I59,F7,2,I73,F7,2,I86,F7,2/I)
108. END

```

COCHRAN - CULL CONEY FILE EXP-ERTP

6. SLOPE

```
1.
2.
3.
4.
5.
6.
SUBROUTINE SLOPE(X1,Y1,X2,Y2,OM,THET)
SLOPE = (Y2-Y1)/(X2-X1)
YINTCP = Y1 - SLOPE * X1
THET = SLOPE*OM+YINTCP
RETURN
END
```

DATE 041472

PAGE

47

28 JAN 72 14:15:21 CYCLE: 0

04 FEB 72 08:23:56 CYCLE: 0

## E•STCOMP

```

1.EREP,DPNCOM,DPN,EREP,15,100
2.EREP,COMPILE ELEMENTS ON EREP
3.EREP,PRINTS,TPR5
4.EREP,AA,EREP
5.EREP,PKT,EREP
6.EREP,PDP,FL
7.EREP,COM,COM
8.EREP,SUBROUTINE ALLDON
9.EREP,EREP,ALLDON,ALLDON
10.EREP,SUBROUTINE COMMON
11.EREP,EREP,COMMON,COMMON
12.EREP,SUBROUTINE ELTHET
13.EREP,EREP,ELTHET,ELTHET
14.EREP,SUBROUTINE FLYOVR
15.EREP,EREP,FLYOVR,FLYOVR
16.EREP,SUBROUTINE FOSCAN
17.EREP,EREP,FOSCAN,FOSCAN
18.EREP,SUBROUTINE LATLON
19.EREP,EREP,LATLON,LATLON
20.EREP,SUBROUTINE LUNRAN
21.EREP,EREP,LUNRAN,LUNRAN
22.EREP,MAIN ROUTINE
23.EREP,EREP,MAIN,MAIN
24.EREP,MISSION ROUTINE
25.EREP,EREP,MISSION,MISSION
26.EREP,SUBROUTINE NUNDON
27.EREP,EREP,NUNDON,NUNDON
28.EREP,SUBROUTINE PASBAD
29.EREP,EREP,PASBAD,PASBAD
30.EREP,SUBROUTINE PPASS
31.EREP,EREP,PPASS,PPASS
32.EREP,SUBROUTINE PPROP
33.EREP,EREP,PPROP,PPROP
34.EREP,SUBROUTINE PSCRM
35.EREP,EREP,PSCRM,PSCRM
36.EREP,SUBROUTINE PTSCRM
37.EREP,EREP,PTSCRM,PTSCRM
38.EREP,SUBROUTINE PTSCRP
39.EREP,EREP,PTSCRP,PTSCRP
40.EREP,SUBROUTINE SUNEL
41.EREP,EREP,SUNEL,SUNEL
42.EREP,SUBROUTINE THEMEX
43.EREP,EREP,THEMEX,THEMEX
44.EREP,SUBROUTINE TIMDAT
45.EREP,EREP,TIMDAT,TIMDAT
46.EREP,SUBROUTINE SLOPE
47.EREP,EREP,SLOPE,SLOPE
48.EREP,SUBROUTINE VROTAT
49.EREP,EREP,VROTAT,VROTAT
50.EREP,SUBROUTINE ANTAGD
51.EREP,EREP,ANTAGD,ANTAGD
52.EREP,EREP,ERMAP,ERMAP

```

ON SAKOLOSKY

DATE 041472

26 JAN 72 14:11:36 CYCLE: 0

CUCCHMAN - CULL COREY FILE EREP+ERE

E.SUNEL

SUBROUTINE SUNEL(ETA,TNOON,THETA,EL,NOSUCH)

1.

2.C

3.C

4.C

5.C

6.C

7.C

8.C

9.C

10.C

11.

12.

13.

14.

15.

16.

17.

18.

19.

20.

21.

22.

23.

PURPOSE: GIVEN ETA, AND THE VALUES OF THETA WHEN THE SUN  
ELEVATION ANGLE IS 30 DEG. AND INCREASING AND  
30 DEG. AND DECREASING (OR ANY SIMILAR PAIR),  
SOLVE FOR THE SUN ELEVATION ANGLE AT THE GIVEN  
VALUE OF THETA.

RTOD=180./3.1415927

NOSUCH=0

IF(TNOON.GE.360.)TNOON=TNOON-360.

C=ACOS(COS(TNOON/RTOD)\*COS(ETA/RTOD))

X=ACOS(TAN(TNOON/RTOD)/TAN(C))

Z=COS(THETA/RTOD)\*COS(C)+SIN(THETA/RTOD)\*SIN(C)\*COS(X)

IF(ABS(Z).GT.1.)GO TO 30

EL=90.-(ACOS(Z)\*RTOD)

RETURN

NOSUCH=1

EL=0.

RETURN

END



DATE 041472

08 MAR 72 01:55:29 CYCLE: 0

E.THEMAX

```

1.C
2.
3.C
4.
5.
6.
7.
8.
9.
10.C
11.
12.
13.
14.
15.
16.
17.
18.
19.
20.
21.
22.
23.C
24.C
25.C
26.
27.
28.
29.
30.
31.
32.
33.
34.
35.
36.
37.
38.
39.
40.C
41.C
42.C
43.
44.
45.
46.
47.
48.
49.
50.
51.
52.
53.
54.
55.
56.C

SUBROUTINE THEMAX(ARCPAS,MCATEG,THMAX,THMIN,*GTMXC,*GTMXU)

INCLUDE BLK1,LIST
INCLUDE BLK3,LIST
INCLUDE BLK5,LIST
COMMON/PRNT/IPRINT
DIMENSION TMAX(3,250)
INTEGER THETA(400), OP1, OP2

NTOTAL = 0
NMAX = 0
ISTART = 0
DO 5 I = 1,3
  DO 5 J = 1,250
    5 TMAX(I,J) = 0
  IF(THMIN.NE.0) .OR. THMAX.NE.0) GO TO 10
  BEGIN = THMIN
  XEND = THMAX
  GO TO 15
10 BEGIN = THMIN
  XEND = THMAX
20.
21.
22.
23.C
24.C
25.C
26.
27.
28.
29.
30.
31.
32.
33.
34.
35.
36.
37.
38.
39.
40.C
41.C
42.C
43.
44.
45.
46.
47.
48.
49.
50.
51.
52.
53.
54.
55.
56.C

BUILD THETA VECTOR
15 DO 100 J=1,NSCORE
  MISS = SCORE(6,J)
  IF(MCATEG.EQ.234) GO TO 40
  IF(MCATEG.LE.4) .AND. MISS.NE. MCATEG) GO TO 100
  OP1 = MCATEG/10
  OP2 = MOD(MCATEG,10)
  IF(MISS.EQ. MCATEG) .OR. MISS.EQ. OP1) .OR. MISS.EQ. OP2)
    . GO TO 40
  GO TO 100
20 CALL PASBAD(SCORF(2,J),SCORE(3,J),BEGIN,XEND,IGOOD)
  IF(IGOOD.EQ.0) GO TO 100
  NTOTAL = NTOTAL + 1
  THETA(NTOTAL) = J
30.
31.
32.
33.
34.
35.
36.
37.
38.
39.
40.C
41.C
42.C
43.
44.
45.
46.
47.
48.
49.
50.
51.
52.
53.
54.
55.
56.C

PUT DATA IN ASCENDING ORDER BY THETA
110 IF(NTOTAL.NE.0) GO TO 112
  WRITE(A,1500)
  RETURN
112 ISTART = ISTART + 1
  L = THETA(ISTART)
  DO 115 J=ISTART,NTOTAL
    K=THETA(J)
    IF(SCORF(2,L).LE. SCORF(2,K)) GO TO 115
    THETA(ISTART) = K
    THETA(J) = L
    L = K
115 CONTINUE
  IF(ISTART.NE. NTOTAL) GO TO 112

```

```

57.C EVALUATE THE INTERVAL
58. 120 NMAX = IMAX + 1
59. K = THETA(NMAX)
60. RBEGIN = SCORE(2,K)
61. REND = RBEGIN + AKCPAS
62. IF (REND .GT. XEND) GO TO 220
63. TMAX(3,NMAX) = RBEGIN
64. DO 200 K=NMAX,NTOTAL
65. N = THETA(K)
66. CALL PASRAD(SCORE(2,N),SCORE(3,N),RBEGIN,REND,15000)
67. IF (1000 .EQ. 0) GO TO 200
68. TMAX(1,NMAX) = TMAX(1,NMAX) + SCORE(5,N)
69. TMAX(2,NMAX) = TMAX(2,NMAX) + SCORE(4,N)
70. CONTINUE
71. IF (NMAX .NE. NTOTAL) GO TO 120
72. NMAX = NMAX - 1
73. MSTRXC = TMAX(1,1)
74. MSTMXU = TMAX(2,1)
75. THSTRT = TMAX(3,1)
76. DO 300 K=1,NMAX
77. IF (AMAX1(MSTRXC,TMAX(1,K)) .EQ. MSTRXC) GO TO 300
78. MSTRXC = TMAX(1,K)
79. MSTMXU = TMAX(2,K)
80. THSTRT = TMAX(3,K)
81. CONTINUE
82. THEND = THSTRT + AKCPAS
83. DO 400 K=1,NMAX
84. TMAX(1,K) = 0
85. MSTRXC = MSTRXC
86. MSTMXU = MSTMXU
87. IF (JPRINT .EQ. 1) RETURN
88. WRITE(6,2000) IREV, THSTRT, THEND, NC, RU
89. RETURN
90. 1500 FORMAT(//I2,'THERE ARE NO ENTRIES IN SCORE IN THE ',
91. 'SPECIFIED MISSION CATEGORY.')
92. 2000 FORMAT(I2,'REV ',I4,3X,'THSTRT = ',F8.3,3X,'THEND = ',F8.3/
93. 'T13,MSTRXC = ',F8.3,2X,'MSTMXU = ',F8.3)
94. END

```

26 JAN 72 14:12:52 CYCLE: 0

## E.TIMDAT

SUBROUTINE TIMDAT(TIME,MO,IDAY,HR)

PURPOSE: CONVERTS TIME EXPRESSED AS TIME ELAPSED FROM  
MIDNIGHT ON DEC 31 TO MONTH, DAY, AND HOUR

IF(TIME .GE. 120) GO TO 40  
IF(TIME .GE. 31) GO TO 10

MO = 1

DUM = TIME

GO TO 150

10 IF(TIME .GE. 59) GO TO 20

MO = 2

DUM = TIME - 31

GO TO 150

20 IF(TIME .GE. 90) GO TO 30

MO = 3

DUM = TIME - 59

GO TO 150

MO = 4

DUM = TIME - 90

GO TO 150

40 IF(TIME .GE. 151) GO TO 50

MO = 5

DUM = TIME - 120

GO TO 150

50 IF(TIME .GE. 181) GO TO 60

MO = 6

DUM = TIME - 151

GO TO 150

50 IF(TIME .GE. 212) GO TO 70

MO = 7

DUM = TIME - 181

GO TO 150

70 IF(TIME .GE. 243) GO TO 80

MO = 8

DUM = TIME - 212

GO TO 150

80 IF(TIME .GE. 273) GO TO 90

MO = 9

DUM = TIME - 243

GO TO 150

90 IF(TIME .GE. 304) GO TO 100

MO = 10

DUM = TIME - 273

GO TO 150

100 IF(TIME .GE. 334) GO TO 110

MO = 11

DUM = TIME - 304

GO TO 150

MO = 12

DUM = TIME - 334

IDAY = DUM

HX = (DUM - IDAY) \* 24

IDAY = IDAY + 1

XTIME

END

DATE 041472

20 MAR 72 09:24:33 CYCLE: 0

CUCHEAN - CULL COREY FILE EREP•EREP

E.TOTAL

```

1. DIMENSION EREP(15,1000)
2. NAMELIST/INEREP/EREP
3. READ(5,INEREP)
4. DO 100 K = 1,1000
5. IF(EREP(1,K) .LT. -9000) GO TO 200
6. IF(EREP(1,K) .LT. 0) GO TO 100
7. NUMEX = EREP(1,K)
8. WTEA = EREP(15,K)
9. ISTART = K + 1
10. DO 90 J = ISTART, 1000
11. IF(EREP(1,J) .LT. -9000) GO TO 95
12. IF(EREP(1,J) .LT. 0) GO TO 90
13. NUM = EREP(1,J)
14. IF(NUM .NE. NUMEX) GO TO 90
15. EREP(1,J) = -EREP(1,J)
16. WTEA = WTEA + EREP(15,J)
17. 90 CONTINUE
18. 95 WRITE(6,1000) NUMEX, WTEA
19. 100 CONTINUE
20. 1000 FORMAT(/T2,'EXPERIMENT NUMBER:',15,5A,
21. 'TOTAL *EIGHT = ',F8.4)
22. 200 END

```

26 JAN 72 14:13:43 CYCLE: 0

E.VROTAT

```

1.C
2.      SUBROUTINE VROTAT (V, R, THETA, VNEW)
3.C
4.C      THE PURPOSE OF THIS SUBROUTINE IS TO ROTATE AN ARBITRARY VECTOR V(3)
5.C      ABOUT AN ARBITRARY AXIS R BY AN ANGLE OF THETA, GIVEN IN DEGREES.
6.C      THE ROTATION AXIS IS DESCRIBED BY THE UNIT VECTOR R(3). THE
7.C      RESULTING VECTOR IS GIVEN BY VNEW(3).
8.C
9.C      DIMENSION V(3), R(3), VNEW(3), B(3), C(3), D(3), DSINT(3),
10.C      CCOST(3), DANDC(3)
11.C
12.C      SINHE = SIN (THETA * .17453293E-01)
13.C      COSTHE = COS (THETA * .17453293E-01)
14.C
15.C      C IS THE COMPONENT OF V ALONG R
16.C
17.C      VDOTR=VDOT(V,R)
18.C      B(1)=VSCALR(R,VDOTR)
19.C
20.C      C IS THE COMPONENT OF V PERPENDICULAR TO R
21.C
22.C      C(1)=VSUB(V,B)
23.C
24.C      D COMPLETES A RIGHT HANDED ORTHOGONAL COORDINATE SET
25.C
26.C      D(1)=VCROSS(R,C)
27.C
28.C      VNEW, THE RESULTING ROTATED VECTOR, IS GIVEN BY D * SIN(THETA) + C *
29.C      COS(THETA) + B
30.C
31.C      DSINT(1)=VSCALR(D,SINHE)
32.C      CCOST(1)=VSCALR(C,COSTHE)
33.C      DANDC(1)=VADD(DSINT,CCOST)
34.C      VNEW(1)=VADD(DANDC,B)
35.C
36.C      RETURN
37.C      END

```

08 MAR 72 09:12:19 CYCLE: 0

E•WGTADD

1. SURROUTINE WGTADD

PURPOSE: SIMULATE THE EFFECT OF HAVING PERFORMED AN EREP PASS  
 BY ADDING TO TOTSCR THE TOTAL WEIGHT ACCOMPLISHED,  
 RECORDING THE DATE THE PASS WAS MADE IF REPEATING  
 PASSES ARE REQUIRED, AND RECORDING VALUE OF OMEGA IF  
 COMPLETE COVERAGE IS REQUIRED

```

INCLUDE BLK2,LIST
INCLUDE BLK3,LIST
INCLUDE BLK5,LIST

IADD = 0
DO 100 I = 1,NSCORE
  CALL PASBAD(SCORE(2,I),SCORE(3,I),THSTRT,THEND,IGOOD)
  IF(IGOOD .EQ. 0) GO TO 100
  IADD = 1
  NPT = SCORE(1,I)
  TOTSCR(NPT) = TOTSCR(NPT) + SCORE(5,I)
  TOTSCR(NPT) = AMINI(TOTSCR(NPT),EREP(15,NPT))
  ICOV = EREP(7,NPT)
  ITYPE = MOD(ICOV,10)
  IF(ITYPE .EQ. 0) GO TO 100
  IF(ITYPE .EQ. 1) GO TO 20

```

THERE IS CURRENTLY NO LOGIC TO HANDLE TOTAL COVERAGE REQUIREMENTS.

GO TO 100

REPEATING COVERAGE

```

20 DO 50 K = 1,200
  IF(NPT .EQ. PASDAT(1,K)) GO TO 60
  50 IF(PASDAT(1,K) .EQ. 0) GO TO 55
  WRITE(6,6000)
  RETURN
  55 PASDAT(1,K) = NPT
  60 PASDAT(2,K) = PASDAT(2,K) + 1
  NENTR = PASDAT(2,K)
  NPLACE = NENTR + 2
  PASDAT(NPLACE,K) = TRAJ(4,LINE)
  100 CONTINUE
  IF(IADD .NE. 0) GO TO 120
  S = THSTRT
  E = THEND
  WRITE(6,5000) IREV, S, E
  RETURN
  120 IPASS = IPASS + 1
  PASS(1,IPASS) = IREV
  PASS(2,IPASS) = TRAJ(2,LINE)
  IF(PASS(2,IPASS) .LT. 0) PASS(2,IPASS) = 360. + PASS(2,IPASS)
  PASS(3,IPASS) = TRAJ(4,LINE)
  PASS(4,IPASS) = THSTRT
  PASS(5,IPASS) = THEND
  CALL SUNEL(TRAJ(3,LINE),TRAJ(5,LINE),THSTRT,EL,NOSUCH)
  PASS(6,IPASS) = EL

```

```

57. CALL SUNEL(TRAJ(3,LINE),TRAJ(5,LINE),THEND,EL,NOSUCH)
58. PASS(7,IPASS) = EL
59. RETURN
60.C
61.C
62. 5000 FORMAT(/T2,'SUBROUTINE "GTADD"--FOR IREV = ',I4,' AND'/
63. . T2,'THSTKT = ',F8.3,',', THEND = ',F8.3,', NO ENTRIES IN ',
64. . 'SCORE ARE ACCEPTABLE.')
```

END CULL - DATE: 04/14/72 TIME: 15:33:19 CARD COUNT: 2166 LABEL COUNT: 0 ITEM COUNT: 0

REPORT E.  
FORPUB 0023-04/14-15:33

# ERE\*ERE\* ELEMENT TABLE

D NAME	VERSION	TYPE	DATE	TIME	SEQ #	SIZE-PRE,TEXT	(CYCLE WORD)	PSKMODE	LOCATION
ELTHET		FOR SYMB	26 JAN 72	14:00:19	1	6	5	0	1792
ELTHET		RELOCATABLE	26 JAN 72	14:00:26	2	1	5		1798
SUNEL		FOR SYMB	26 JAN 72	14:11:36	3		5	0	1804
SUNEL		RELOCATABLE	26 JAN 72	14:11:40	4	1	7		1810
TIMUAT		FOR SYMB	26 JAN 72	14:12:52	5		5	0	1818
TIMUAT		RELOCATABLE	26 JAN 72	14:12:58	6	1	11		1829
VRUTAT		FOR SYMB	26 JAN 72	14:13:43	7		5	0	1841
VRUTAT		RELOCATABLE	26 JAN 72	14:13:50	8	2	5		1849
SLOPE		FOR SYMB	28 JAN 72	14:15:21	9		5	0	1856
SLOPE		RELOCATABLE	28 JAN 72	14:15:27	10	1	2		1858
LLIST		SYMBOLIC	28 JAN 72	15:02:42	11		5	0	1861
LUNRAN		SYMBOLIC	01 FEB 72	12:29:09	12		5	0	1863
LATLON		SYMBOLIC	01 FEB 72	11:00:39	13		5	0	1873
LATLON		RELOCATABLE	01 FEB 72	13:45:48	14	1	4		1877
LATLON		RELOCATABLE	01 FEB 72	13:46:12	15	1	6		1892
STCOMP		SYMBOLIC	04 FEB 72	08:23:56	16		5	0	1899
RECOMP		SYMBOLIC	11 FEB 72	08:36:25	17		5	0	1910
FILES		SYMBOLIC	22 FEB 72	12:32:37	18		5	0	1916
CIMOTE		ELI SYMB	29 FEB 72	10:52:23	19		5	0	1919
CUM		FOR PROC	07 MAR 72	22:32:40	20		1	0	1940
ALLDON		SYMBOLIC	08 MAR 72	01:44:32	21		0	0	1946
ALLDOH		RELOCATABLE	08 MAR 72	01:44:34	22	2	12		1959
COMMON		SYMBOLIC	08 MAR 72	01:45:40	23		0	0	1973
COMMON		RELOCATABLE	08 MAR 72	01:45:43	24	2	26		1997
MAIN		SYMBOLIC	08 MAR 72	01:48:39	25		0	0	2025
MAIN		RELOCATABLE	08 MAR 72	01:48:42	26	4	46		2071
MAIN		SYMBOLIC	08 MAR 72	01:51:24	27		11	0	2124
MAIN		RELOCATABLE	08 MAR 72	01:51:26	28	2	10		2135
MAIN		SYMBOLIC	08 MAR 72	01:51:27	29		0	0	2147
MAIN		RELOCATABLE	08 MAR 72	01:51:29	30	1	6		2154
MAIN		SYMBOLIC	08 MAR 72	01:51:31	31		0	0	2161
MAIN		RELOCATABLE	08 MAR 72	01:51:33	32	2	8		2168
MAIN		SYMBOLIC	08 MAR 72	01:51:35	33		13	0	2178
MAIN		RELOCATABLE	08 MAR 72	01:51:37	34	2	11		2191
MAIN		SYMBOLIC	08 MAR 72	01:54:02	35		15	0	2204
MAIN		RELOCATABLE	08 MAR 72	01:54:04	36	2	17		2219

PTSCM	08 MAR 72	01:54:07	37	2	13	U	0	1	2238
PTSCM	RELOCATABLE	08 MAR 72 01:54:09	38		14				2251
PTSCM	SYMBOLIC	08 MAR 72 01:54:12	39		11	U	0	1	2267
PTSCRP	RELOCATABLE	08 MAR 72 01:54:13	40	2	11				2278
THEMAX	SYMBOLIC	08 MAR 72 01:55:29	41		20	U	0	1	2291
THEMAX	RELOCATABLE	08 MAR 72 01:55:31	42	2	19				2311
AGTAUD	SYMBOLIC	08 MAR 72 09:12:19	43		16	5	0	1	2332
AGTAUD	RELOCATABLE	08 MAR 72 09:12:40	44	2	14				2348
FOSCAN	SYMBOLIC	08 MAR 72 09:42:32	45		58	5	U	1	2364
FOSCAN	RELOCATABLE	08 MAR 72 09:42:50	46	3	52				2422
MASCAN	SYMBOLIC	08 MAR 72 15:17:38	47		25	5	0	1	2477
MASCAN	RELOCATABLE	08 MAR 72 15:17:49	48	2	27				2502
MISSION	SYMBOLIC	14 MAR 72 13:07:17	49		14	5	0	1	2531
MISSION	RELOCATABLE	14 MAR 72 13:07:19	50	2	13				2545
FLYOVR	SYMBOLIC	16 MAR 72 14:58:18	51		50	5	0	1	2560
FLYOVR	RELOCATABLE	16 MAR 72 14:56:43	52	2	46				2610
TWGT	SYMBOLIC	20 MAR 72 09:24:33	53		5	5	0	1	2658
TWGT	RELOCATABLE	20 MAR 72 09:24:50	54	1	6				2663
MSORT	SYMBOLIC	20 MAR 72 09:46:40	55		30	5	0	1	2670
MSORT	RELOCATABLE	20 MAR 72 09:46:51	56	2	34				2700
CINOTE	RELOCATABLE	20 MAR 72 13:26:31	57	1	1				2736
ENMAP	SYMBOLIC	20 MAR 72 13:27:11	58		1	5	0	1	2738
ENMAP	ABSOLUTE	20 MAR 72 13:28:07	59		488				2739
SITES	SYMBOLIC	22 MAR 72 15:55:21	60		25	5	0	1	3227
SITES	RELOCATABLE	22 MAR 72 15:55:44	61	2	25				3252
ENMAP	ABSOLUTE	22 MAR 72 15:56:52	62		488				3279
ENMAP	RELOCATABLE	22 MAR 72 15:56:52	62		488				3279

NEXT AVAILABLE LOCATION-

## ASSEMBLER PROCEDURE TABLE EMPTY

CUBOL PROCEDURE TABLE EMPTY

# FORTAN PROCEDURE TABLE

[illegible]

## ENTRY POINT TABLE

D NAME	LINK	D NAME	LINK	D NAME	LINK	D NAME	LINK
ALLOOL	22	CINOTE	57	COMMON	24	ELTHET	2
FURMAINS	61	FURMAINS	56	FURMAINS	54	FURMAINS	50
FOSCAN	46	LATLON	15	LOMRAN	14	NONDON	48
PASBAU	30	PPASS	32	RPROP	34	PSCRCM	36
PTSCRPP	40	SLOPE	10	SUNEL	4	TMDAY	42
VROJAT	H	PGIADU	44				



CUCHAN - CULL COREY FILE EREP.EREP

TIME: 00:00:14.359 IN: 7 OUT: 0

PAGES: 56

INITIATION TIME: 15:30:36-APR 14, 1972

TERMINATION TIME: 15:33:22-APR 14, 1972

COR-BLK-SECS: 1335

IO COUNT: 77

CHARGE: 1.519

DATE 041472

PAGE

58

APPENDIX B

PROGRAM OUTPUT EXAMPLES

EXAMPLES OF "SIGN-ON" PROCEDURE, AND  
OF "evaluate" AND "maxwgt" OPTIONS

```
@asg,ax erep
READY
@add erep.files
READY
READY
READY
READY
READY
READY
READY
READY
READY
READY
READY
READY
@asg,ax ersav
READY
@asg,a ersav2
READY
@use 20,ersav
READY
@use 21,ersav2
READY
@xqt erep.ormap
```

CURRENT VALUES OF CONTROL VARIABLES ARE:

```
APASDE = 60.0000   MCATDE = 234   MISSION = 2
THMIDE = -90.0000   THMADE = 270.0000   PERWT = 1.000
IERSAV = 0         IEROLD = 0         IDEVIC = TERMNL
ARCMIN = 1.000     OK?
yes
```

```
FIRST REV = 16.      LAST REV = 440.
FIRST REV WITH ABS BETA GREATER THAN 50: 0
LAST REV WITH ABS BETA GREATER THAN 50: 0
```

```
OPTION =
evaluate
```

```
REV NO.?      (14 FORMAT)
0061
```

```
OPTION =
maxwgt
```

```
ARCPAS = 60.0000   MCATEG = 234
THMIN = -90.0000   THMAX = 270.0000   O.K.?
yes
```

```
REV 61   THSTRT = 22.483   THEND = 82.483
        WGTMXC = 697.163   WGTMXU = 697.163
```

```
OPTION =
```

EXAMPLE OF PRINT OPTION "prpsum"

print  
PRINT OPTION =  
prpsum

TYPE PROPOSAL NUMBER (14 FORMAT)  
0416

PROPOSAL SUMMARY, REV = 0							
PROP	FO	PASS	COND	PASS	UNCOND	ACCUM	TOT POS
416	1	.0			.0	11.1	33.3
416	2	.0			.0	33.3	33.3
416	3	.0			.0	20.4	33.3
416	TOTAL	.0			.0	64.8	100.0

OPTION =

# EXAMPLE OF PRINT OPTION "prgsum"

prgsum

WEIGHT ACCOMPLISHED SO FAR:				
DATE	ACCUM WGT	TOT WT POS	NON ZERO FOS	TOT FOS
2	767.1	1883.2	40	146
23	383.3	697.4	7	18
24	.0	50.0	0	2
234	3437.4	8352.3	124	335
3	480.1	1698.2	27	152
34	143.3	683.3	4	19
4	.0	2002.6	0	185
TOTAL	5211.3	15367.1	211	857

OPTION =

# EXAMPLE OF PRINT OPTION "pscore"

print  
PRINT OPTION =  
pscore

EREP PTR	THETA S	THETA E	WGT UNC	WGT CON	CATEG
1.00	48.92	55.86	14.29	14.29	2.00
70.00	48.48	53.71	6.67	6.67	2.00
83.00	52.57	52.86	10.00	10.00	2.00
84.00	22.64	29.99	50.00	50.00	2.00
97.00	51.22	52.64	33.33	33.33	2.00
98.00	60.66	79.96	6.67	6.67	2.00
100.00	26.08	39.80	6.67	6.67	2.00
118.00	51.53	52.25	33.33	33.33	2.00
142.00	22.48	40.75	4.35	4.35	2.00
163.00	46.40	52.36	50.00	50.00	23.00
167.00	48.92	55.86	50.00	50.00	234.00
201.00	51.15	52.21	11.11	11.11	234.00
203.00	45.37	61.37	1.85	1.85	234.00
252.00	48.92	55.86	50.00	50.00	234.00
262.00	48.92	55.86	25.00	25.00	234.00
266.00	48.92	55.86	25.00	25.00	234.00
271.00	50.46	52.67	25.00	25.00	234.00
295.00	47.15	47.18	10.00	10.00	234.00
315.00	51.49	52.15	10.00	10.00	234.00
345.00	51.23	52.90	5.88	5.88	234.00
364.00	33.48	34.92	8.33	8.33	234.00
365.00	33.48	34.92	8.33	8.33	234.00
368.00	33.48	39.26	8.33	8.33	234.00
369.00	33.48	39.26	8.33	8.33	234.00
372.00	50.50	57.34	8.33	8.33	234.00
376.00	26.52	40.75	25.00	25.00	234.00
377.00	51.05	89.02	25.00	25.00	234.00
388.00	-8.65	.00	50.00	50.00	234.00
394.00	37.80	50.73	16.67	16.67	234.00
402.00	51.24	52.90	5.00	5.00	234.00
428.00	51.49	52.27	7.69	7.69	234.00
432.00	51.24	52.90	50.00	50.00	234.00
433.00	50.46	52.67	47.00	47.00	234.00
480.00	37.80	50.73	50.00	50.00	234.00

OPTION =

# EXAMPLE OF PRINT OPTION "revsum"

```
print
; PRINT OPTION =
revsum
```

```
THSTRT = 22.4826    THEND = 82.4826    O.K.?
yes
```

SUMMARY OF REV 61, THSTRT = 22.5, THEND = 82.5

MCATEG	WGT COND	WGT UNC	ACCUM	TOT WGT POS	NO FOS
2	165.3	165.3	.0	165.3	9
23	50.0	50.0	.0	50.0	1
24	.0	.0	.0	.0	0
234	481.9	481.9	.0	753.9	24
TOTAL	697.2	697.2	.0	969.2	34

OPTION =

# EXAMPLE OF PRINT OPTION "revsum"

p  
PRINT OPTION =  
revsum

THSTRT = 165.6282      THEEND = 225.6282      O.K.?

SUMMARY OF REV \*\*\*, THSTRT = 165.6, THEEND = 225.6

LOCATED	WGT COND	WGT UNC	ACCUM	TOT WGT POS	NO FOS
23	.0	.0	.0	.0	0
234	262.5	287.5	100.0	412.5	11
3	22.4	73.3	54.3	76.7	10
34	.0	.0	.0	.0	0
TOTAL	284.9	360.8	154.3	489.2	21

OPTION =



EXAMPLE OF PRINT OPTION "missum"

print  
PRINT OPTION =  
missum

SUMMARY OF ACCUMULATED SCORE FOR SL3 MISSION

MISSION CAT	WGT SAT	TOT WGT POS	NON ZERO FOS	TOT FOS
23	383.3	697.4	7	18
234	3437.4	8352.3	124	335
3	480.1	1698.2	27	152
34	143.3	683.3	4	19
TOTAL	4444.1	11431.3	162	524

OPTION =

## EXAMPLE OF PRINT OPTION "eresum"

P  
PRINT OPTION =  
eresum

PROPOSAL SUMMARY, REV = 2194

PROP	FO	PASS COND	PASS UNCOND	ACCUH	TOT POS
352	TOTAL	.0	.0	10.6	100.0
399	TOTAL	.0	.0	100.0	100.0
5472	TOTAL	.0	.0	100.0	100.0
455	TOTAL	.0	.0	100.0	100.0
363	TOTAL	.0	.0	40.0	100.0
446	TOTAL	.0	.0	12.5	100.0
523	TOTAL	.0	.0	.0	100.0
582	TOTAL	.0	.0	33.3	100.0
373	TOTAL	.0	.0	.0	100.0
485	TOTAL	.0	.0	100.0	100.0
402	TOTAL	.0	.0	.0	100.0
2540	TOTAL	.0	.0	100.0	100.0
8540	TOTAL	.0	.0	14.3	100.0
3540	TOTAL	.0	.0	.0	100.0
6540	TOTAL	.0	.0	.0	100.0
416	TOTAL	.0	.0	64.8	100.0
386	TOTAL	.0	.0	25.0	100.0
69	TOTAL	.0	.0	55.6	100.0
5540	TOTAL	.0	.0	.0	100.0
345	TOTAL	.0	.0	100.0	100.0
23	TOTAL	.0	.0	.0	100.0
390	TOTAL	.0	.0	.0	100.0
352	TOTAL	.0	.0	10.6	100.0
531	TOTAL	50.0	50.0	.0	100.0
117	TOTAL	.0	.0	100.0	100.0
3407	TOTAL	.0	.0	.0	100.0
413	TOTAL	.0	.0	5.6	100.0
473	TOTAL	.0	.0	50.0	100.0
528	TOTAL	100.0	100.0	.0	100.0
3568	TOTAL	.0	.0	.0	100.0
4568	TOTAL	.0	.0	.0	100.0
212	TOTAL	.0	.0	100.0	100.0
323	TOTAL	.0	.0	.0	100.0
353	TOTAL	.0	.0	50.0	100.0
385	TOTAL	.0	.0	.0	100.0
443	TOTAL	.0	.0	.0	100.0
459	TOTAL	.0	.0	50.0	100.0
496	TOTAL	12.5	12.5	.0	100.0
497	TOTAL	.0	.0	75.0	100.0
498	TOTAL	.0	.0	75.0	100.0
501	TOTAL	.0	.0	75.0	100.0
558	TOTAL	.0	.0	100.0	100.0
325	TOTAL	.0	.0	.0	100.0
332	TOTAL	.0	.0	.0	100.0
349	TOTAL	.0	.0	.0	100.0
380	TOTAL	.0	.0	.0	100.0
400	TOTAL	.0	.0	100.0	100.0
423	TOTAL	.0	.0	100.0	100.0
434	TOTAL	.0	.0	50.0	50.0
487	TOTAL	.0	.0	50.0	100.0
491	TOTAL	.0	.0	100.0	100.0
495	TOTAL	.0	.0	100.0	100.0
507	TOTAL	.0	.0	60.0	100.0
527	TOTAL	50.0	50.0	.0	100.0
536	TOTAL	.0	.0	.0	100.0
541	TOTAL	.0	.0	70.0	100.0
557	TOTAL	.0	.0	.0	100.0

# EXAMPLE OF PRINT OPTION "aldone"

print  
PRINT OPTION =  
aldone

## COMPLETED F.O.S

PROP	FO	HCAT	POS SCR	ACCUM SCR
1540	1	2	14.29	14.29
486	8	2	7.69	7.69
436	12	2	7.69	7.69
562	1	2	50.00	50.00
410	1	2	33.33	33.33
352	20	2	2.44	2.44
382	1	2	16.67	16.67
452	1	2	5.55	5.55
452	3	2	5.55	5.55
452	5	2	5.55	5.55
473	1	2	50.00	50.00
529	5	2	2.38	2.38
499	6	2	6.67	6.67
499	7	2	6.67	6.67
499	8	2	6.67	6.67
499	9	2	6.67	6.67
499	10	2	6.67	6.67
520	1	2	11.10	11.10
3009	1	2	50.00	50.00
393	1	2	33.33	33.33
401	7	2	10.00	10.00
438	1	2	50.00	50.00
463	1	2	33.33	33.33
474	1	2	50.00	50.00
504	1	2	25.00	25.00
504	3	2	25.00	25.00
471	1	2	33.33	33.33
517	1	2	33.33	33.33
550	1	2	6.67	6.67
550	2	2	6.67	6.67
550	3	2	6.67	6.67
550	5	2	6.67	6.67
3313	1	2	33.33	33.33
398	1	2	25.00	25.00
417	1	2	33.33	33.33
454	12	2	2.17	2.17
454	14	2	2.17	2.17
454	18	2	2.17	2.17
454	36	2	1.45	1.45
535	4	2	6.25	6.25
599	1	2	4.35	4.35
599	13	2	4.35	4.35
399	1	23	100.00	100.00
5472	1	23	100.00	100.00
455	1	23	100.00	100.00
383	2	23	20.00	20.00
440	5	23	12.50	12.50
485	1	23	50.00	50.00
2549	1	234	100.00	100.00

EXAMPLE OF PRINT OPTION "nowork"

print  
PRINT OPTION =  
nowork

F.O.S	ON WHICH NO WORK HAS BEEN DONE	PROB	FO	MCAT	POS	SCR
486	1	2			7.69	
486	4	2			7.69	
486	6	2			7.69	
486	10	2			7.69	
1457	1	2			33.33	
1457	2	2			33.33	
2457	1	2			50.00	
4457	1	2			100.00	
475	13	2			4.76	
475	16	2			4.76	
475	19	2			4.76	
475	10	2			4.76	
475	1	2			4.76	
475	4	2			4.76	
475	7	2			4.76	
52	21	2			3.33	

COMPLETE PRINTOUT WAS INTERRUPTED AT THIS POINT.

SUMMARY	MISSION CAT	TOT POS SCR	NO OF FOS
2		1075.8	97
23		312.5	11
24		50.0	2
234		4493.2	211
TOTAL		5931.5	321

OPTION =  
OPTION IS NOT VALID.

OPTION =

# EXAMPLE OF PRINT OPTION "passum"

print

PRINT OPTION =  
passum

31 PASSES HAVE BEEN ACCEPTED

REV NO	OMEGA	PASS STARTS			THSTRT	THEND	SUN ELEVATION	
		NO	DAY	HR			START	END
61	248.03	5	4	16.2122	22.48	82.48	58.45	48.34
77	228.79	5	5	17.0753	30.84	90.84	58.14	44.93
93	209.56	5	6	17.9382	40.13	100.13	57.98	41.26
103	214.04	5	7	17.2472	37.80	97.80	51.72	46.58
123	218.51	5	8	16.5560	37.80	97.80	46.64	40.81
137	246.69	5	9	14.3108	65.36	125.36	56.86	33.98
153	227.46	5	10	15.1733	31.80	91.80	31.80	56.96
169	208.24	5	11	16.0357	41.40	101.40	33.73	56.31
184	212.72	5	12	15.3441	42.98	102.98	30.00	57.97
199	217.20	5	13	14.6524	45.31	105.31	27.00	59.45
213	126.89	5	14	20.1761	106.93	166.93	61.17	29.80
249	112.16	5	16	20.3460	103.76	163.76	59.62	44.01
264	116.65	5	17	19.6539	99.38	159.38	54.70	53.18
294	125.63	5	19	18.2693	107.86	167.86	56.87	57.07
325	110.91	5	21	18.4383	96.96	156.96	39.84	77.76
340	115.41	5	22	17.7457	199.32	259.32	41.38	*****
1778	238.77	8	23	19.8505	33.84	93.84	67.42	7.76
1793	243.36	8	24	19.1517	29.40	89.40	76.12	16.13
1854	238.02	8	28	17.9094	34.93	94.93	71.13	23.45
1869	242.61	8	29	17.2103	-2.82	57.18	47.67	58.49
1930	237.29	9	2	15.9667	24.69	84.69	43.51	45.20
1945	241.89	9	3	15.2672	31.58	91.58	41.87	43.10
2010	141.82	9	7	20.2355	82.28	142.28	44.38	29.75
2041	127.33	9	9	20.3892	107.81	167.81	47.29	24.69
2056	131.94	9	10	19.6892	109.69	169.69	48.05	29.16
2086	141.15	9	12	18.2891	111.40	171.40	48.27	39.56
2102	122.06	9	13	19.1422	111.27	171.27	47.11	45.79
2117	126.67	9	14	18.4418	140.35	200.35	59.58	30.31
2132	131.28	9	15	17.7414	104.34	164.34	37.48	60.81
2173	121.43	9	18	17.1928	111.78	171.78	35.97	73.65
2193	126.05	9	19	16.4919	108.81	168.81	29.23	79.45

OPTION =



Subject: A Computer Program to Aid Skylab Earth Resources  
Experiment Planning

From: D. A. Corey

Distribution List

NASA Headquarters

R. O. Aller/MLO  
H. Cohen/MLQ  
J. M. DeNoyer/SR  
\*J. H. Disher/MLD  
\*J. P. Field, Jr./MLB  
T. L. Fischetti/SG  
\*W. D. Green, Jr./MLA  
\*T. E. Hanes/MLA  
E. L. Keith/MLO  
\*A. S. Lyman/MAP  
\*C. W. Mathews/E  
A. B. Park/SRR  
\*M. Savage/MLE  
W. C. Schneider/ML

MSC

M. Hagan/FCD  
\*C. R. Hoss/FM  
G. L. Hunt/FM/13  
M. P. Kilbourn/FM2  
\*K. S. Kleinknecht/KA  
E. L. Lemons/FM2  
E. C. Lineberry/FM6  
R. O. Linney/FCD  
F. C. Littleton/KM  
D. W. Owens/FCD  
O. G. Smith/KW  
P. J. Stull/TD  
H. E. Whitacre/KM  
K. L. Young/FM

MSFC

\*L. F. Belew/PM-SL-MGR  
\*C. C. Hagood/S&E-CSE-A  
\*R. C. Lester/S&E-CSE-MP  
\*R. E. Tinius/S&E-CSE-MP

Lockheed Electronics  
Company, Houston Texas

J. G. Baron  
R. H. Sutton

Philco, Houston, Texas

J. Nemec

BTL

D. A. De Graaf  
D. H. Hagner  
\*H. Z. Hardaway  
\*E. E. Sumner  
\*M. P. Wilson

---

\* Abstract and Introduction Only.